

25 YEAR RE-REVIEW

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION

SPEC. NO. HS

CODE IDENT NO. 73030

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WINDSOR LOCKS, CONNECTICUT, U. S. A.

1.0 GENERAL INFORMATION

1.1 SCOPE

> This specification covers methods and procedures for dry calibrating the IFC-Main Engine Control.

1.2 EQUIPMENT REQUIRED

1.2.1 Adjusting tools such as screw drivers and wrenches will be controlled by assembly floor supervision as to correct application in order to prevent part mutilation yet do the job requirement.

1.2.2 569455**T**6 CIP Valve - Dry Calibration

CIP Dial Indicator 2" stroke, .001" accuracy

Speed Servo Dry Calibration Indicator, 2" stroke, .001" accuracy.

Tll Adapter Arm

T20 CIP Servo Position Locating Fixture

T15 Tt2 Dial Indicator 1 stroke, .001 accuracy

Tl7 Gage Block

T32 Adapter, Sequencing Valve Dry Calibration

The TV simulating Fixture with Two Dial Indicators, 1" stroke .001"

accuracy and Last Word Gage

T55 Cam Motion Tool

T82 Min. Ratio Set Block

T95 Dummy Servo Housing

T96 Fixture - Tt2 Simulator

Tll2 Tt2 Servo Lock

Hunter Force Gage (10# load, 0.1# accuracy)

1.2.3 A pneumatic pressure and vacuum source capable of maintaining any pressure from 2 to 200 psia in order to simulate engine burner pressure (Pb) or engine

2.0 INSPECTION REQUIREMENTS

> There shall be no Inspection requirement other than that data shall be subject to Engineering approval. Wherever initial shimming is changed to meet a functional requirement, this information must be recorded and witnessed as such on initial

3.C SERVO HOUSING

3.1 From the match grind data sheet of the Assembly Check List obtain the "5" dimension from the null point of the CTP Pilot Valve. This dimension is from the bottom of the Pilot Valve Gear to the top of the boss around the Pilot Valve bore in the Pilot Valve Housing.

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- Place the assembled Servo Housing on the dry calibration bench with the parting face up. Support it on the Power Arm using #569455Tl4, Adapter. Place a thickness of the shims (which is equivalent to "S" dimension) between the underside of the Pilot. Valve Gear and the top of the boss around the Pilot. Valve bore to set the CIP Pilot Valve in the null position.
- Being cautions on sealing surface, clamp a 1" Dial Indicator to the flange of the Servo Housing so that the tip of the Dial Indicator is against the upper surface of the Pilot Valve Gear. With the CIP Pilot Valve set in the null position set the Dial Indicator to zero. Set the Indicator so that there is CAUTION- Do not damage sealing surfaces.
- Using a stiff rod, bottom the CIP Servo. Using a 2 to 3 inch depth micrometer measure the distance from the Servo Housing to the machined edge of the Servo Piston. Subtract this from the dimension obtained from HS1502, Para. 2, for zero displacement to obtain the displacement of the CIP Servo Piston. Attach Fixture #569455T7 and set the Dial Indicator to the displacement figured
- 3.5 Connect a 0-50 psia pressure source to the CIP Sensor. The pressure should be capable of being regulated to and read to 1 psi.
- 3.6 Bolt Fixture #569455T20 to the Servo Housing in place of the Servo Stop Cover. The thumb screw on this Fixture and on #569455T6 should be used to move the CIP Servo Piston during the dry calibration.
- With the CIP Servo Piston set at zero displacement, increase the pressure to the CIP Sensor to 5 psia. With this pressure held constant, increase the displacement of the CIP Servo until the CIP Pilot Valve returns to the null displacement of the CIP Servo and plot this point on the Pilot Valve Gear. Read the bration Curve, F-3883 (Ref. Page 16). Repeat this procedure with the CIP Sensor Inlet Pressures of 10, 20, 30 and 40 psia. Always rotate the CIP (to eliminate hysteresis) when "nulling" the CIP Pilot Valve, or tap the fixture with a hammer at frequency of 1 to 3 times per second.

Determine from this plotted curve (Para.3.7) the rate and position adjustments make required adjustments.

Rate: -Counterclockwise swings curve counterclockwise, more effect at top. Position: -Clockwise raises position of curve.

- 3.9 Repeat Paragraphs 3.7 and 3.8 until the actual and the desired curves coincide.
- 4.0 TEMPERATURE SERVO
- 4.1 Shim Temperature Cover Levers for proper location and parallelism, ref. HS1502.

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| 4.2 | Build | บท | temperature | | | | | | 21.8 |
|-----|-------|-----|-------------------|------|-------|-----|--|--------|--------|
| | Lever | and | 1 Feedback Lover, | but | it is | not | necessary to lockwire | the | Inma 4 |
| | | | TOO GO A DEVEL ST | this | time. | Do | necessary to lockwire not install Position | A cl 4 | Tube |

- Attach a Dial Indicator to measure Input Lever travel as shown in Fig. I, page 4.3
- Place a .501" block (#569455T17) between the Input Lever and Feedback Lever and 4.4 holding them in a parallel position, set the Dial Indicator to null position,
- 4.5 Remove .501" block.
- 4.5 Full the Input Lever in the direction towards the Indicator and record the travel beyond null. Record.
- 4.7 Release the Input Lever until it contacts upper limit stop. (See Fig. I).
- 4.8 The total travel of the Input Lever-should be .060" minimum.
- 1.9 The values obtained in Para. 4.6 and 4.7 should be within .010" of each other. If not, remove or add shims under Evacuated Temperature Bellows until this condition is obtained.
- 4.10. If the Indicator reading in Para. 4.6 is larger, add shims under the Evacuated
- 4.11 If the Indicator reading is Para. 4.7 is larger, remove shims from under the Evacuated Bellows.

Example: Value obtained in Para. 4.6 - .100" Value obtained in Para. 4.7 - .040"

> Total stroke therefore is $.100^m + .040^m - .140^m$ or $\pm .070^m$ Therefore, .100" -.070" = .030" extra stroke in pull direction. Since lever ratio results in .006" stroke change for every .001" shim change. A total of $.005^{\text{H}}$ shims must be added $(.03/6 = .005^{\text{H}})$

- 4.12 Lockwire the Input Lever and Feedback Lever and assemble Position Adjustment
- 4.13 Install the Temp. Sensor Simulator #569455T96 (screws to be torqued to 60 lbs.) Simulator must have with it a curve of pressure vs. temperature servo position which is determined by calibrating the simulator against the actual temperature bulb. Curve F-5408, page 15, is for reference only.
- Mount Temp. Sensor Cover vertically as shown in Fig. I, Page 112. 4.14
- 4.15 Set Pressure in Temp. Sensor Simulator to value which corresponds to .953 servo position, then attach 43.6# weight to pin which connects Position Adjustment

5.4

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| | WINDSOR LOCKS, CONNECTICUT, U. S. A. PAGE 5 OF |
|--------|---|
| 4.16 | Adjust Position Adjustment Spring to return Indicator on Input Lever to mill |
| 4.17 | Remove 43.6# weight and install standard Temperature Sensor. Torque screwe |
| 4.18 | Place Sensor in 60°F bath and add weights until Input Lever returns to nell position established in Para. 4.4. |
| 4.19 | Put sensor bulb in 30° or lower bath. Dial Indicator must show a minimum of .020" motion from zero position (null). |
| 4.19.1 | Put sensor bulb in 200° or higher bath. Dial Indicator must show a minimum of .020" motion from zero (nell) position. |
| 4.20 | If stroke in Para. 4.19 is less than specified, remove Sensor and lengthen Connecting Rod. If stroke in Para. 4.19.1 is less, shorten Connecting Red. Amount that the Rod must be changed can be calculated from the following formal |
| e. | Required change in length = (.020 - actual strokd) .174 + .001. |
| 4.21 | Recheck stroke after changing Connecting Rod per Para. 4.18, 4.19, 4.19.1. The standard Sensing Bulb should be used. |
| 4.22 | Reinstall Simulator and reset Position Adjustment for Para. 4.13 thru 4.16 |
| 4.23 | Install Temperature Servo Cover on Linkage Housing. Set mull Indicator on Tt2 O'F end of Ng cam detent. Use tool #569455755 to rotate 3-D Cam. |
| 4.24 | Set Pressure in Temperature Sensor Simulator to pressure value used in Para. 1. 1 |
| 4.25 | Adjust Rate Adjustment until Servo Position is .953" with PV at null position |
| 4.26 | Vary pressure and obtain Servo Position. |
| 4.27 | Data should fall on simulator calibration curve. |
| 5.0 | MIN MAX. RATIO LINES |
| 5.1 | Assembly Spider Housing, Acceleration Limit Lever, Multiplying Linkage and P3 System. (Speed Servo not to be installed at this time). |
| 5•2 | Attach Dial Indicator to Housing to measure roller travel and a Last Word Indicator to measure P.V. null point. Last Word Indicator must be located on the P.V. itself. |
| 5.3 | Set rollers to .057" from pivot with min. ratio adjust by referring to scribed dimension on Multiplying Lever, and Fixture #560,55782 |

Position Droop Cam so that follower is on minimum radius of cam.

(This is max. ratio).

Reference to Curve F-4649, page

dimension on Multiplying Lever, and Fixture #569455T82. Ref. HS1502, page 14.

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- PAGE 6 OF 5.6 Divide value obtained in Para. 5.5 by 52. Record. 5.7 Return Droop Cam to min. ratio position (Ref. Para. 5.1). 5.8 Attach Simulator #569455T46, T.V. Spring to Housing and engage spring with Multiplying Linkage. NOTE: Use TV Spring from unit being tested. Ref. Fig. 2 5.8.1 Install spring so that open end of hook is facing away from WI/P3 Roller 5.9 Install rigging pin in T.V. Pilot Valve and "zero out" Last Word Indicator at null position as logged on calibration sheet. 5.10 Attach a line with an air pressure source to P3 sensor so that pressure can be 5.11 Set air pressure to 40 psi and adjust T.V. until T.V. Pilot Valve is mulled out and record reading on T.V. Simulator. Record. 5.12 Repeat above with air pressure at 80 psi. Record. Subtract value obtained in Para. 5.11 from value obtained in Para. 5.12. Record. 5.13 5.14 Rotate Droop Cam for max. ratio condition. Ref. Para. 5.4. Set air Pressure to 40 and adjust TV until TWPV is mulled out and record 5.15 reading of Dial Indicator on TV Simulator. Record. 5.16 Repeat above with air pressure at 80 psi. Record. 5.17 Subtract value obtained in Para. 5.17 by value obtained in Para. 5.13. 5.18 Divide value obtained in Para. 5.17 by value obtained in Para. 5.13. Value, Para. 5.13
- 5.19 Compare value obtained in Para. 5.7 to value obtained in Para. 5.18, If values are equal min. ratio has been properly act.
- If value obtained in Para. 5.18 is greater than value obtained in Para. 5.7. adjust min, ratio to move rollers away from pivot. NOTE: when adjusting rollers for min. ratio, attach dial indicator to measure roller position and note change on Dial Indicator when making adjustment. This is important since adjustment may slip and reference point is lost.
- 5.21 If necessary to readjust min. ratio in Para. 5.20, repeat Para. 5.8 thru 5.18 until value obtained in Para. 5.18 is within \$.05 of that obtained in Para. 5.7 NOTE: Sensitivity - .05/.001 approx.
- 5.0 SEQUENCING VALVE OPERATION

5.20

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| PAGE | I | O | 2 | |

- 6.ì Install sequencing valve, spring and retainer in Linkage housing. 6.2
- Install rubber gasket to replace "HiCeal" in servo pressure transfer tube line.
- 6.3 Install test fixture #569455T95.
- 6.4 Install Power Lever and index to 15° as per procedure for setting Idle and Military Droop. Ref. Para. 7.5.1.
- 6.5 Attach Pressure Gages to S.O.V. and recirculation Valve signal lines.
- 6.8 Connect a .032" orifice between the gage lines.
- Supply 200 psi to servo line in Linkage Housing. Ref. Fixture #569455732. 6.7
- Adjust button on P.L. until S.O.V. signal = 150 psi with PLA between 1. **6.8** and circulate valve signal = 150 psi with PLA between 6° - 12°.
- 6.9 Rotate PLA from 0° to 15° in 1° staps and record pressure in S.O.V. and recirculation valve signal line in the day of the compact of the line.
- .0 IDLE AND MILITARY TRIMMERS
- .1 Preliminary setup.
- Shim under Trimmer Housing per HS 1502 so that holes in Trimmer Housing line up .1.1 with holes in Servo Housing.
- .1.2 Locate Trimmer Honeing with respect to P.L. Shaft per HS 1502.
- 7.3.3 Build up Trimmer Housing and install nominal shims as experience dictates. (Suggested shims, .020 in Idle and .120 in Military.)
- 7.1.4 Adjust acceleration limit cam follower all the way into lever.
- 7.1.5 Install 3-D cam and position and lock cam at 59° position as follows:
- Locate detent in 3-D Cam and rotate 3-D Cam counterclockwise looking into open 7.1.5.1 end of Ng bore until 0° end of detent is located.
- Using depth micrometer, measure from surface of Tt2 Piston to parting face of 7.1.5.2
- Subtract .075" from reading in Para. 7.1.5.2. Record. 7.1.5.3
- .1.5.4 Reset Tt2 Serve Piston to dimension obtained in Para. 7.1.5.3 and Tock with Tt2
- With follower still in detent, attach #569455T8 to measure Speed Servo Position. .1.6 Set Dial Indicator to read 1.357.

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PAGE _ 8 7.1.7 Install #569455T46 and attach Mal Indicator to measure roller travel. With droop cam position at min. ratio position, set Dial Indicator to read .037, 7.1.8 Attach Dummy Servo Housing #569455T95, Power Lever, and Trim Blades. 7.2 Droop Slope 7.2.1 Set P.L. at 75° - 120°, idla trim 2 turns from full counterclockwise. 7.2.2 Adjust Military Trimmer so that a roller position of .2124" ± .001" is obtained with Ng position of .500". 7.2.3 Vary Ng Serve from .600" to .300" in .100" increments. Slope of line must match slope of line shown in Curve F-5407, page 18. 7.2.3.1 If slope is steeper than desired, remove shims from under Tt2 reset lever 7.2.3.2 If slope is shallower than desired, add shims under Tt2 reset bracket. 7.2.3.3 This requires removal of Tt2 reset bracket. Shim until shope is obtained 7.3 Trimmer Housing Location for Minimum Interaction 7.3.1 After obtaining desired slope, reset the 75° - 120° lever angle and .500" Ng servo position. Adjust Idle Trimmer from full counterclockwise to full clockwise and note if roller position changes. If a change in roller position occurs, Trimmer Housing must be repositioned with respect to its location from Power Lever centerline as follows: 7.3.1.1 If a clockwise adjustment of the Idle Trim Blade causes rollers to move toward min. ratio, Trimmer Housing should be moved toward the P.L. 7.3.1.2 If a clockwise adjustment of the Idle Trim Blade causes roller to move toward max. ratio, Trimmer Housing should be moved away from the P.L. 7.3.1.3 When no apparent motion occurs, a final check should be made as follows: 7.3.1.3,1 Set Idle Trim Blade two turns from full counterclockwise, PL at 75° - 120°. 7.3.1.3.2 Bring Ng servo from .800" position to .495" position and note roller position 7.3.1.3.3 Set Idle Trim Blade at one full turn from full clockeise, PL at 75° - 120°. 7.3.1.3.4 Bring Ng servo from .800" position to .495" position and note roller position. 7.3.1.3.5 Value obtained in Para. 7.3.1.3.4 should equal value obtained in Para.

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7.5.3.3

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WINDSOR LOCKS, CONNECTICUT, U. S. A. PAGE 9 OF 7.3.1.3.6 It value obtained in Para, 7.3.1.3.4 is greater than the value octained in Para. 7.3.1.3.2 by more than .006", then Trimmer Housing must be relocated 1030 0307 If value obtained in Fara. 7.3.1.3.4 is less than value obtained in Para. 7.3.1.3.2 by more than .006", then Trimmer Housing must be relocated away from Fower Lever. 7.30.04 Scribe Housing in final location. 7.4 Military Trim Set Point 7.4.1 Adjust Idle Trimmer se that it is two full turns from full counterclockwise position and Military 5 clicks from full counterclockwise. 7.4.2 Set PL at 75° - 120° and bring Ng servo from .800° position to .195° position 1040 Adjust Military Trim Biade to be within 5 clicks from full clockwise position i olioi Bring Ng serve from .800" position to .495" position and record Wf/P3 7.4.5 The mid position between Para. 7.4.2 and 7.4.4 should be .213" roller position. 10405.1 If the roller position, as determined in Para. 7.4.5 is less than .213", remove shims from Military Trim Push Rod. 7.11.5.2 .001" worth of shims is equivalent to about .003" roller travel. NOTE: 1.k.5 After completion of simming, set Military Trimmer to .213" roller position. This should occur at mid range of adjustment. 7.5 IDLE TRIM SET POINT 7.5.1 With Ng servo .495" retate Power Lever from 40° position until no change in roller position occurs and set P.L. protractor to read 59° at this point. Stop plate should then be set so that the min. power lever stop is at 0° on 7.5.2 Set P.L. at idle (15° Index Pin location). 7.5.3 Move Ng servo from 1.300" position until Wf/P3 rollers move to .115" position 7.5.3.1 Desired Ng servo position is 1.236%. 7.5.3.2 If Ng Servo indicator reads high, remove shims from Idle Trim Rod.

If Ng servo indicator reads low, add shims to Idle Trim Rod.

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PAGE __ OF 7.50 1 Sain and Na serve is 1.236" with roller at .115" position. 7.5.1 Recheck salitary set coint. This should not have changed. If change did occur reshord Irimer Housing location per Para. 7.3. Anta Lange .5.1 With Military Irlumer set at nominal position (Ref. Para. 7.4.6) and P.L. at 3, receis title trim within 5 clicks of full clockwise position. 1.6. Move Ng serve from 1.30% position until .115% roller position is obtained. Ng del vo lodicator of old read 1.15" or less. Record actual value. Bestern Icle Trimmer to point 2 varms from full counterclockwise. o la 7 . 6 . 1. Set P. L. at 75° and rotage military trimmer within 5 clicks of full countersockal Mays Ng serve from . 100" position to .195". Roller position should be .199 or least Rotat Military Trimer within 5 clicks of full clockwise and move Ng servo from . 300" to .435" position. Bollers should be at .241 or more position. into: Breet Military Trimmer to nominal position. ... ACCELARATION LIMITING ADJUSTMENT 8.1 To be done after Idla and Military set points are shimmed and min. and max. 8.2 Remove serve Housing and set Dial Indicator on Wf/P3 rollers to read .057 when droop can is no disard to min. ratio position. 0.3 Move De serve to 1.1154 position and adjust cam follower on acceleration limiting lever to locate Wiles rellers at .246 position. 3.4 Em calibration so indicated below: Spaed Serve Dial Indicator Actual Position 30t 1.45# Spec. Position 1.40 1.35 .124 ± .002 1.30 .136 ± .002 1.20 .165 ± .002 1.154

8.5

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

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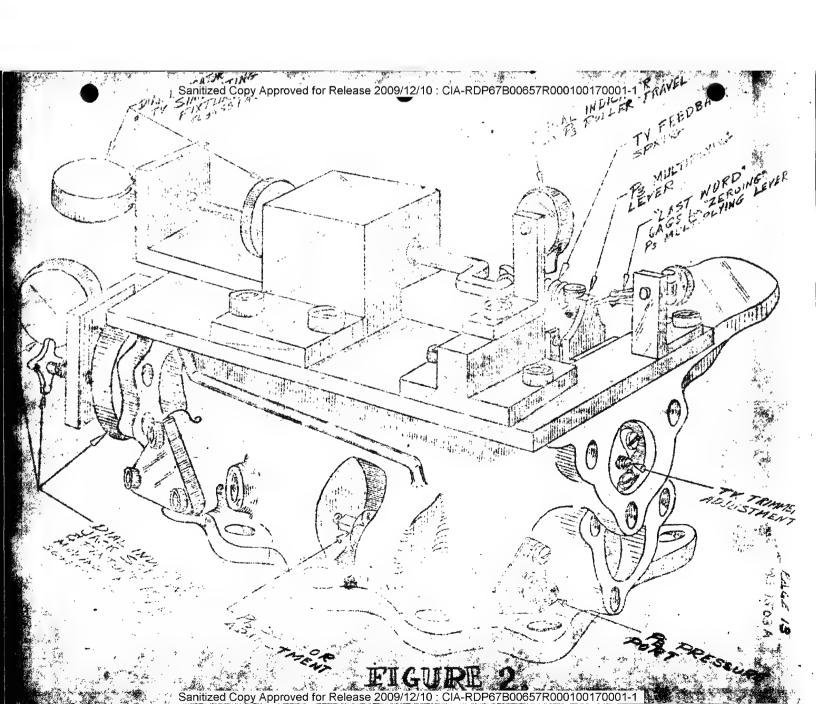
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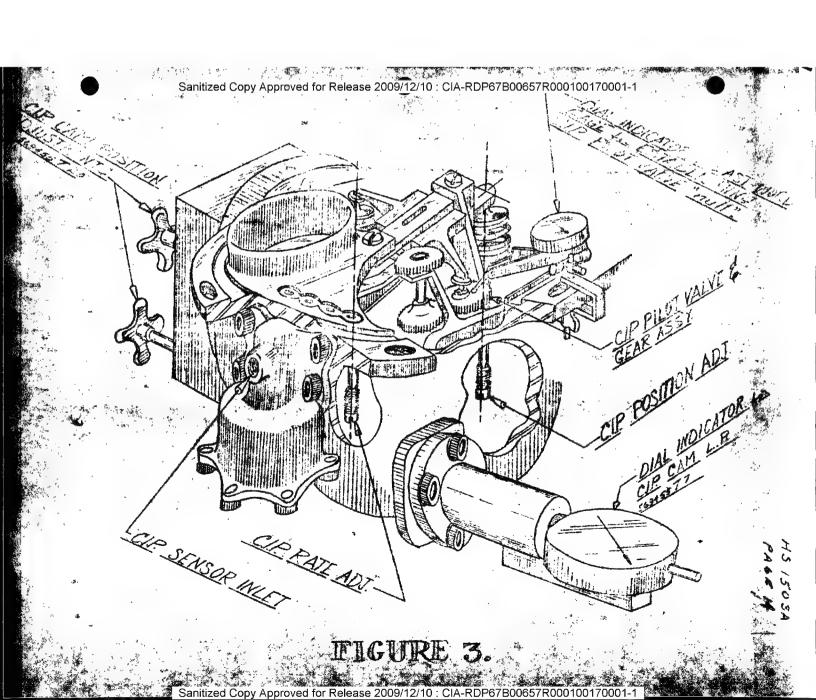
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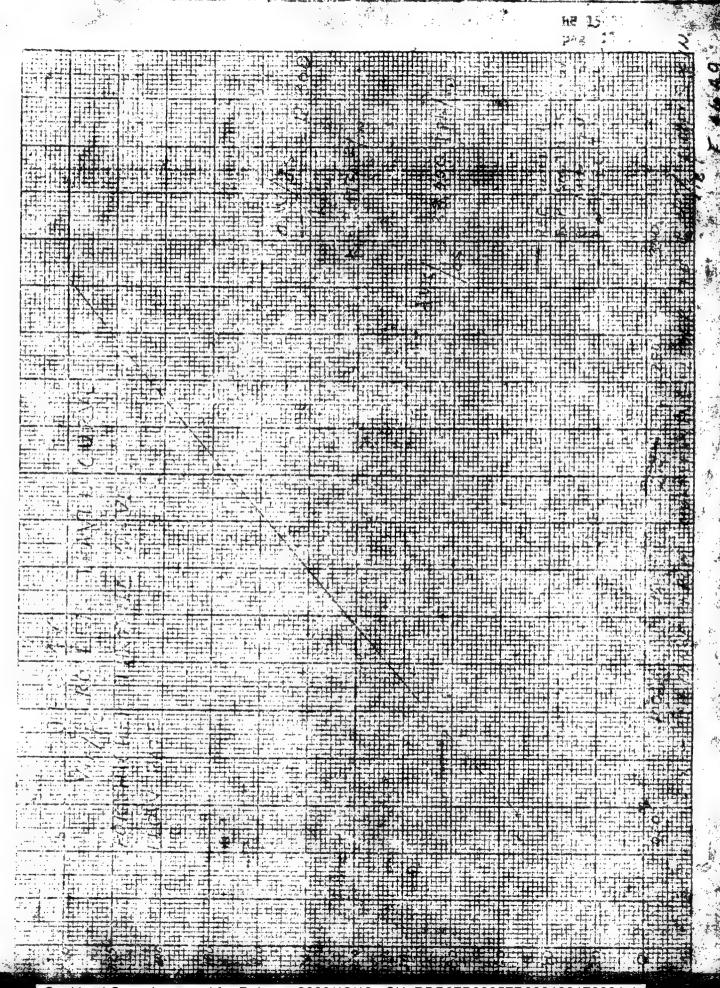
elevation limit each or check points may be reduced to points indicated.

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT

H.S. 1593 A

Amend. /

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E.C. 73182

Date: 8-10-62

H.S. 1503 A "Dry Calibration JFC 47 Main Control"

Amendment /

1. Add Paragraph 1.1.1 as follows:

For purposes of this specification, controls are grouped as follows:

Group I X Controls (Including Xcrm)
Group II Y Controls (Production Units #12,13)
Group III Y Controls (#14 P.U. & Subsequent)

2. Change Paragraph 7.4.2 from:

7.4.2 Set PL at 75° - 120° and bring Ng servo from .800" position to .495" position and record Wf/P3 roller position.

To read:

- 7.4.2 Set PL at 75° 120° and bring Ng servo from .800" position to .495" position and record Wf/P3 roller position. Call this Valve A.
- 3. Change Paragraph 7.4.4 from:

7.4.4 Bring Ng servo from .800" position to .495" position and record Wf/P3 roller position.

To read:

- 7.4.4 Bring Ng servo from .800" position to .495" position and record Wf/P3 roller position. Call this Valve B.
- 4. Change Paragraph 7.4.5 from:
 - 7.4.5 The mid position between Para. 7.4.2 and 7.4.4 should be .213" roller position. is. 7.4.2 + 7.4.4 = .213"
 - To read:
 - 7.4.5 Military trimmer must be shimmed such that 3A + B 213 For Group II Control 227 For Group III Control 227 For Group III Control 227 For Group III Control

DIVISION OF UNITED AIRCRAFT CORPORATION VIDESON LOCKS, CONNECTION

A.S. 5034 Amond, 2 Page 1 of 1 B.C. A273323 Date: P-22-62

H.S. 1503A "DRY CALIBRATION JFC47 MAIN CONTROL"

Amendment &

- 1. In paragraph 7.4.2 change last sentence which reads "Call this Valve A" to read "Call this Valve B."
- 2. In paragraph 7.4.4 change last sentence which reads "Call this Valve B" to read "Call this Valve A."

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H.S. 1503 Amond. Fage 14.6 E.C. Am7hh-06 Date: 11/2/62

H.S. 1503 A Dry Calibration JFCh7 Main Centrol

Amendment 3

Amend as follows:

- 1. Delete paragraph 2.0 in its entirety.
- 2. Change Paragraph 6.0, Sequencing Operation, to Paragraph 4.0 Change Paragraph 4.0, Temperature Servo, to Paragraph 5.0 Change Paragraph 5.0, Min Max Ratio Lines, to paragraph 6.0
- 3. Paragraph 6.7. Change #200 psi* to read #250 psi*.

Change Paragraph 6.9 to read:
Rotate PLA from 0° to 15° in tabular form. Results shall fell within limits of Curve P55hh.

L. Change Paragraph 7.4.2 to read: Set PL at 75 - 120° and roller position. Call this value B.

Change Paragraph 7.4.4 to read: Bring Ng serve from roller position. Call this value A.

Add the following:

- 9.0 Compressor Bleed Actuator (CRA)
- 9.1 Mount assembled CBA unit on fixture plate using rubber "O" seals. Use 75 1250 terque on screws.

9.2 Mount protractor fixture to CBA susput shaft.

- 9.3 Attach supply line to component test rig and with P1 & P2 closed, set supply pressure to 50 psig.
- 9.4 Set valves as follows in the sequence shown:
 - 1. Open P body
 - Open P2 drain
 Close P1 drain
 - 4. Open PI
- 9.5 Index protractor fixture to 0° position.
- 9.6 Set valves as follows in the sequence shown:
 - 1. Close Pi
 - 2. Open Pl drain
 - 3. Close P2 drein
 - 4. Open P2

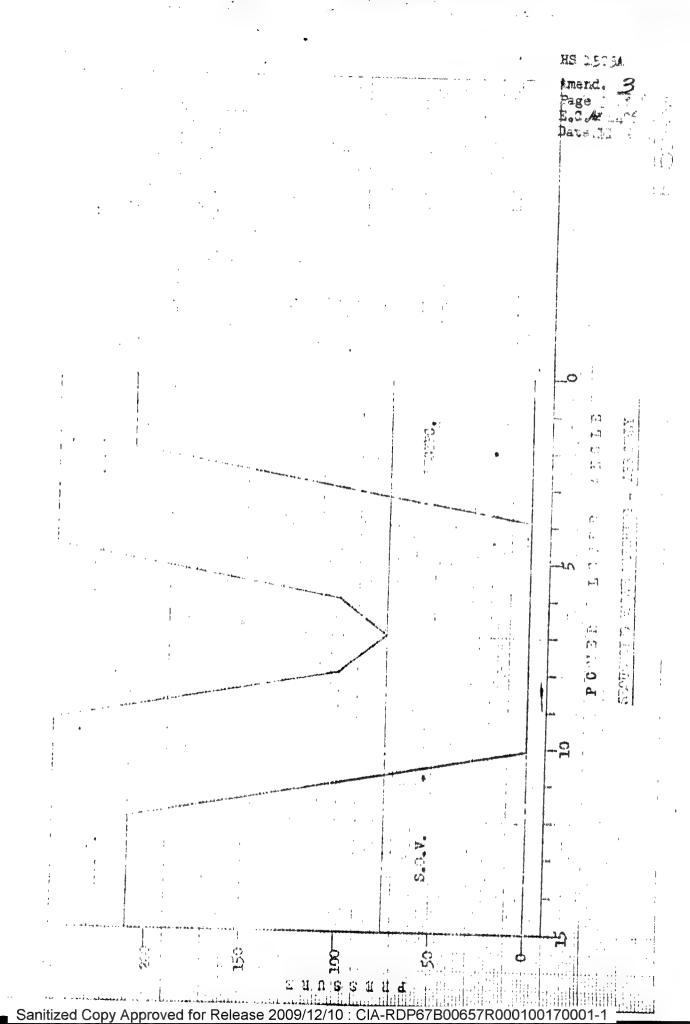
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WINDSON LOCKS, CONNECTION

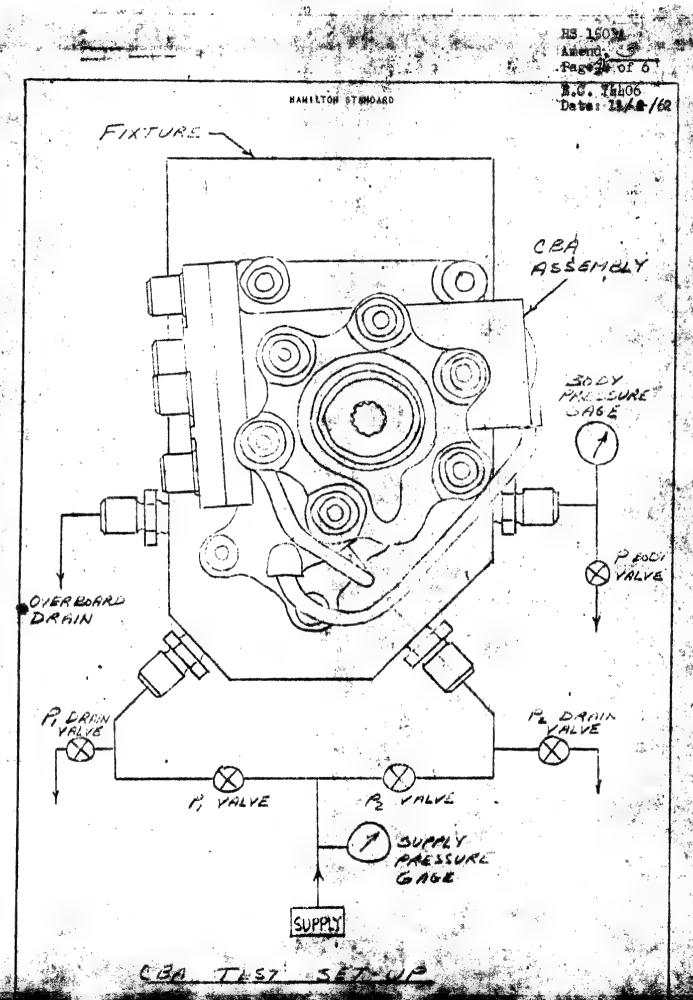
H.S. 1503 April 9 Page 2 01. 6 E.C. AZ74406 Date: 11/2/62

H.S. 1503 A Dry Calibration JFC47 Main Control-

Amendment 3

- 9.7 Record protractor reading. Angular movement (hydraulically) shall be 30° to 35°: If not within limits check to see whether CBA piston length piston bore depth, and piston cover stop length are within blueprint tolerances.
- 9.8 Close Pl drain valve and increase supply pressure to 200 psi. Slowly close P body valve until body pressure reaches 150 psi.
- 9.9 Measure overboard drain leakage. Leakage shall be no greater than 5 drops per minute. If leakage exceeds this amount, check face of carbon seal and lap if necessary.
- 10.0 Shut off supply pressure and open body pressure and drain valves. Remove whit from fixture.





HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTIONS:

H.S. 1503 A

Amend. /

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E.C. 73182

Date: 8-10-62

H.S. 1503 A "Dry Calibration JFC 47 Main Control"

| Amendment | |
|-----------|--|
| | |

5. Change Paragraph 7.4.5.1 from:

7.4.5.1 If the roller position, as determined in Para. 7.4.5 is less than .213", remove shims from Military Trim Push Rod.

To read:

7.4.5.1 If the roller position, as determined in Para. 7.4.5 is less than vlave in table, remove shims from Military Trim Push Rod.

6. Change Para. 7.6.4.1 froms

7.6.4.1 Move Ng servo from .800" position to .495" Roller position should be .193" or less.

To read:

7.6.4.1 Move Ng servo from .800" position to .495". Roller position should

7. Change Para. 7.6.5 from:

7.6.5 Rotate Military Trimmer within 5 clicks of full clockwise and move Ng servo from .800" to .495" position. Rollers should be at .244 or more

To read:

7.6.5 Rotate Military Trimmer within 5 clicks of full clockwise and move Ng servo from .800" to .195" position. Rollers should be at F or more

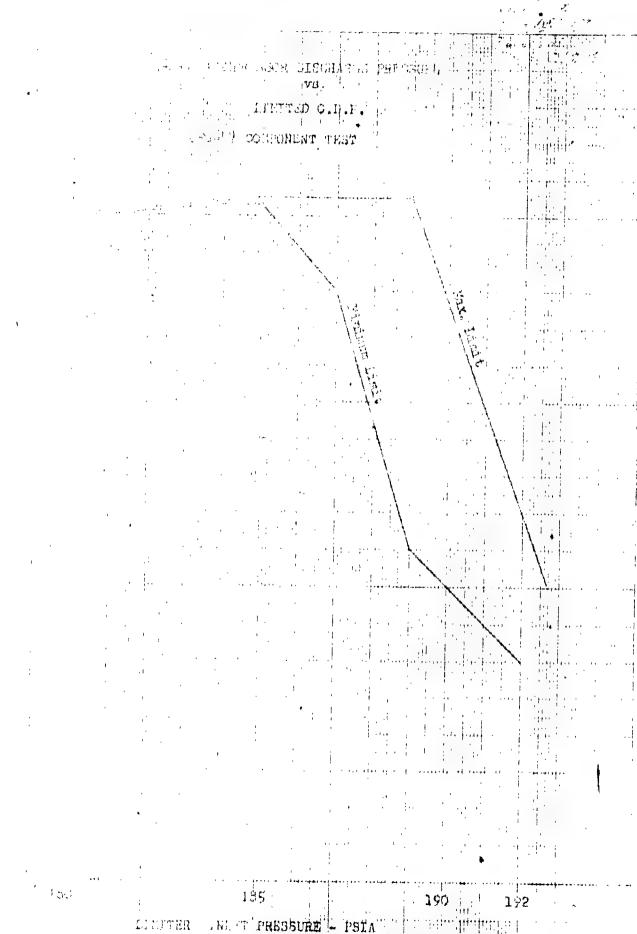
For Group I Controls .176 .224 Group II Controls .184 .232 Group III Controls .190 .240 DIVISION OF UNITED AIRCRAFT CORPORATION WINDSON LOCKE, CONNECTION

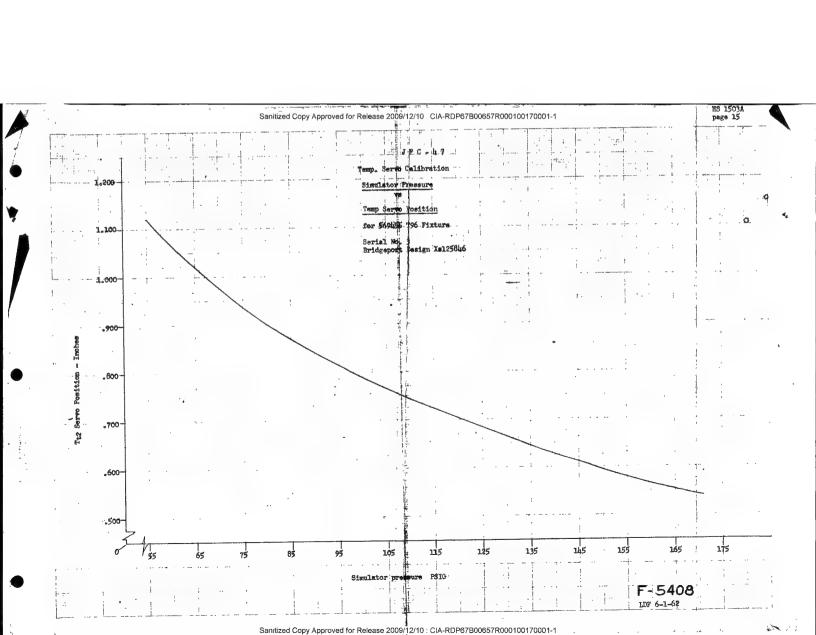
H.8. 1503A Amond. 3 Page 5 of 6 E.C. AZ 74406 Date: 11/25/62

H.S. 1503A Dry Calibration JFCh? Main Control

Amendment.

- 10.0 ODP LIMITER
- 10.1 Install CDP Limiter sub assembly into CDP Limiter dry calibration fixture 569455-T-139, using 4 serems.
- 10.2 Install inlet and outlet lines to limiter with a 200 psi Heise gage in each
- 10.3 Supply air at 180 psia to inlet, and read corresponding outlet pressure outlet gage.
- 10.4 Continue to increase inlet pressure in 1.0 psi increments recording the corresponding outlet pressure at each point.
- 10.5 Outlet pressures shall fall within the limits of spec. curve.
- 10.6 If points are below limit, turn adjusting nut clockwise; if above, turn counterclockwise. Sontinue adjusting antil limits of paragraph 5 are met.
- 10.7 No leakage before the cracking pressure is acceptable. (one indication of leakage is when the outlet pressure does not equal the inlet pressure at values of 18k pais and below).
- 10.8 After final calibration insert locking key into shaft and looksire key to hexagon mut.







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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS CONNECTICUT. U. S. A.

SPEC NO. HS 1520A

CODE 730 30

1.0 SCOPE

The purpose of this specification is to provide a leak test procedure for steel housings used on the JFC-47 exhaust nozzle control and regulator in order to insure detection of minute leaks. Under no circumstance does this specification apply to any other JFC-47 housing.

- 2.0 EQUIPMENT REQUIRED
- 2.1 A pressure test rig capable of maintaining 5.750±50 psi for 15 minutes under seepage conditions.
- 2.2 Pressure gages.

High pressure 0-6000 psig - 1% Accuracy. Low pressure 0-500 psig - 1% Accuracy.

- 2.3 Suitable fixtures to apply pressure to the high pressure area, as specified on the applicable blueprint, and to bleed air from the area being tested.
- 2.4 Suitable fixtures to apply pressure to the low pressure area, as specified on the applicable blueprint, and to bleed air from the area being tested.
- 3.0 TEST FLUID

Test fluid shall be MIL F-7024A, type II corrosion inhibted demineralized water (potassium dichromate, .1% - .2% by weight.)

4.0 DEFINITION OF EXTERNAL LEAKAGE

With the required pressure applied to the designated portion of the housing, remove all traces of fluid from the exterior surfaces. The term "no leakage" shall be defined as no appearance of fluid on the external surface of a housing, including no seepage or wetting on the surface, regardless of the fact that fluid does not run off the surface of the housing or forms droplets.

- 5.0 METHOD OF TEST
- 5.1.0 Leak test of high pressure area.
- 5.1.1 Install suitable fixtures to pressurize the high pressure area of the housing as designated by the blueprint. Apply pressure and bleed air from this area. Increase pressure to 5750 ± 50 psig and hold for 5 minutes.
- 5.1.2 Cycle pressure from 1000psig 4000 psig 50 times. Time required to increase pressure from zero to 4000 psig should be 35% to 12% seconds for each cycle.
- 5.1.3 Apply 5750 ± 50 psig in high pressure area and hold for 15 minutes. There shall be no external leakage during this time.

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

SPEC NO. HS 1520A

CODE 733 30

| 5.2.0 | 92 1 . | | • | . . | | |
|-------|--------|------------------------------|----|------------|--|-------|
| 20200 | ьеак | test | OI | TOM | pressure | area. |
| | | ACCRECATE AND REAL PROPERTY. | | - | Constitution of the last of th | |

- 5.2.1 Install suitable fixtures to pressurize the low pressure area of the housing as designated by the blueprint. Apply pressure and bleed air from this area. Increase pressure to 320 ± 20 psi and hold for 5 minutes.
- 5.2.2 Cycle pressure from 1000 to 250 psig 50 times. Time required to increase pressure from zero to 250 psig should be 55 12 seconds for each cycle.
- 5.2.3 Apply 320 ± 20 psig in low pressure area and hold for 15 minutes. There shall be no external leakage during this time.



HSF-755.1A 5/61

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

| SPEC. | NO. | H\$ | 1579 | A | |
|-------|------|-------|-------|---|--|
| CODE | IDEN | T NO. | 73030 | | |
| PAGE | 2 | OF | | | |

1.0 SCOPE

This specification describes the phases of assembly which are not included on the appropriate assembly drawing of the Bypass and Shut-Off Valve, 576497.

2.0 DESCRIPTION

This unit consists essentially of a dump valve and sleeve, a shut-off valve and sleeve, a recirculating valve and sleeve and a housing. Each valve is spring loaded and shim adjusted. It is located downstream of the main fuel control. The operation of the unit is dependent on the position of the sequencing valve which, in turn, is positioned in accordance with power lever angle.

3.0 EQUIPMENT REQUIRED

No special tools are required.

- 4.0 ASSEMBLY PROCEDURE
- 4.1 Cleanliness of Parts

All parts must be kept free of dirt, dust, grit and other foreign matter.

- 4.2 Assembly of Check and Dump Valve
- 4.2.1 Assemble one (1) shim 520128, washer 571398, and spring 583422 into the housing and piston set 583487.
- 4.2.2. Install shims 569669 under the packing 560006 to reduce the clearance between the packing and the cover and tubes assembly 580825 to .000-.002. The thickness of shims can be found by the following procedure:

A - Lay the packing 560006 in the housing.

- B Measure the distance from the flange parting surface to the packing.
- C Measure the distance from the flange surface of the cover and tubes assembly 580825 to the packing sealing surface and add .001.
- D Subtract dimension C from dimension B and the remainder should be the thickness of shims required (ref. Fig. 1).

Remove the packing 560006 and install the required thickness of shims. Install the packing 560006 on the shims and gasket 69397A30 at the parting surfaces of the housing and piston.

- 4.2.3 Assemble washer 571397 to cover and tubes assembly 580825. Install cover and tubes assembly 580825. Install cover and tubes assembly to the housing using two (2) bolts 69408B25-15 and six (6) bolts 69408B25-7 per the print 576497.
- 4.2.4 Torque each bolt to 125-135 in. lb. and secure with lockwire MS20995N32.
- 4.3 Assembly of Recirculating Valve

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

| | SPEC. | NO, H | 5 | 15 19 | A |
|---|-------|-------|----------|-------|---|
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| | DAGE | 3 | 06 | | |

- This valve is located between the dump valve and the minimum pressure and shut-off valve as shown on the assembly drawing 576497. To determine the thickness of shims 569669 required to obtain a .000-.002 clearance between the seal retainer 558905 and the parting surface of housing, measure and record the dimension between the packing 583488 and the parting surface of the housing (ref. Fig. 2). Subtrace from this dimension the distance from the cover flange surface to the seal retainer mating surface on the cover 558904.
- Assemble two (2) rotating rings 575577, from (4) gaskets 69588B58, and (2) packings 69587A-58 in the housing, piston and valves set 583487 (Ref Fig 3). Measure and record the total height of the sleeve, from the packing 583488, sealing stack, and the length of the seal retainer 558905. Subtract the sum of the above dimensions plus .001 from the dimension obtained in paragraph 4.3.1. The remainder will be the thickness of shims 569669 required.
- 4.3.3 Assemble the sealing stack with the thickness of shims 569669 as determined above in the housing, piston and valves set 583487. Install seal retainer 558905 into the housing and then carefully install piston to avoid damage to the chevron seal.
- 4.3.4 With the piston 583488 in a fully downward position, install .060" of shim 520128, spring retainer 571403, and helical springs 579202 and 579203 into the bore of housing, piston and valves set 583487.
- 4.3.5 Assemble spring retainer 571403 on top of the helical springs and assemble gasket 69397A32 to the housing. Install cover 558904 to the housing using five (5) attaching bolts 69408B25-9.
- 4.3.6 Torque each bolt to 125-135 in. 1bs and safety with lockwire MS20995N32.
- 4.3.7 Install name plate 69444B4 using two (2) screws 69415-0-4.
- 4.4 Assembly of Minimum Pressure and Shut-off Valve
- 4.4.1 Assemble the minimum pressure and shut-off valve by repeating paragraphs 4.3.1 through 4.3.6.
- 5.0 Preservation and Storage
- After completion of testing, the bypass and shutoff valve shall be drained of fuel and prepared for storage in accordance with HS Spec. 1613. Suitable covers shall be used to prevent damage or contamination of the assembly.
- Preparation For Shipping
 The unit shall be completely free of internal and external foreign material at the time of packaging and during shipping. All ports shall be capped with suitable plastic caps or their equivalent.

H.S. 1579A Page 4 of

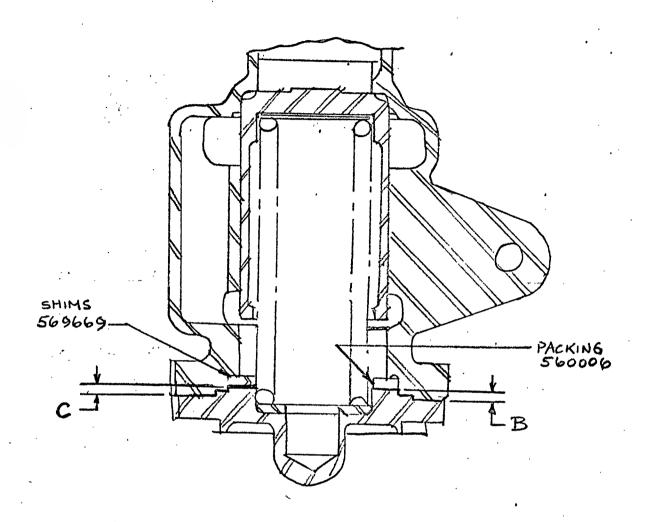
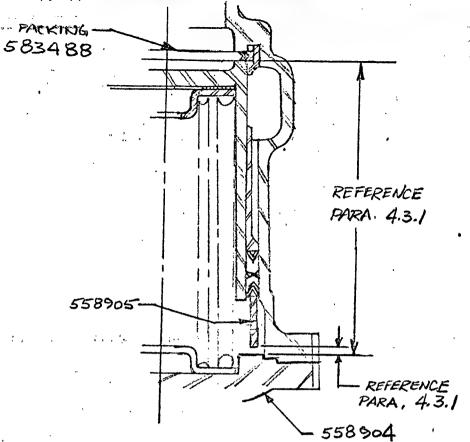


Figure 1



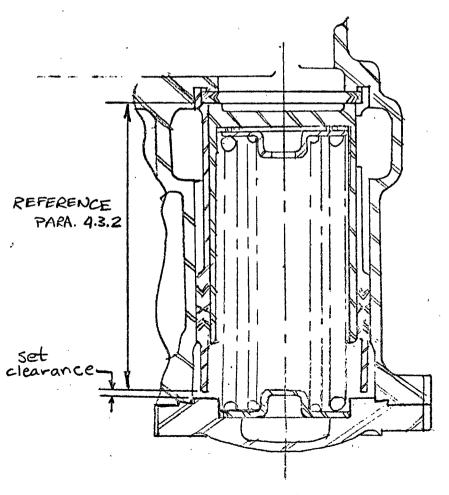


Figure 2

Figure 3

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

HS SPEC NO_1368_

1.0 SCOPE

This specification defines the qualify, performance and design requirements for a device for weads full mains of a fivel, adjustment on an airborne fuel control used on a gas turbiles sugles.

2.0 APPLICABLE SPECIFICATIONS AND PUBLICATIONS

The following specifications and publications shall form a part of this specification. In case of conflict between the referenced documents and this specification, the provisions of this specification shall apply:

MIL-D5028 - Drawings and Data Lists, Preparation of

MIL-I-5007B - Engines, Afmoraft, Turkedet, General Spec for

MIL-S-77hS - Serew Threads, Standard Assonautical

MIL-E-50720 - Environmental Testing, Astronautical and Associated equipment, Ceneral Spec for

MIL-W-Soll - Welding - Munion, of Steels and Corrosion and Heat Resisting Alloys, Processes for

MII-F-Ild - Preservation, Methods of

MIL-I-9500 - Exterobangeability and Replaceability of Component Parts for Aircraft.

MII-E-5009B - Engines, Turbojet, Qualification Testing of

HS178 - Specification for Corrosion Resistant Steel Parts

HS191 - Fusion Welding

HS782 - Brazing - Hydrogen, Nickel Base Brazing Alloys

USAF Bulletin No. 23 - Materials and Process Spec.

ANA Bulletin No. 143 - Specification and Standards, Use of

ANA Bulletin No. 147 - Non-Government Agency Specifications

ANC-5 - Strength of Aircraft Elements

HSD Drawing #567085 - Housing and Inserve, Serve

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HAMILTON STANDARD

DIVISION OF UNITED AIRCRAFT CORPORATION

ON

SPEC NO. HS

1368

CODE 73930

WINDSOR LOCKS, CONNECTICUT, U. S. A.

3.0 DESCRIPTION

The remote trimmer shall consist of a package with externally available Idle Trimmer screw in ground adjustment, and elso brically operated Military trimmer screw for in-whight adjustment, with mounting pad. In addition, the Mil speed shall be capable of ground adjustment with test stand tardware without removal of the unit from the control and while the control is pressurised.

4.0 DESIGN

It shall be the goal. If the design to the sapeble of successfully completting a nomponent qualiffication test as lettined in the following paragraphs of the respective military posification:

MIL-E-5009B

4.3.3.3.3 - Explication Proof Test

1.3.2.5.6 - Imart Test

MIL-E-5' 10

Paration Brook Brook Book

Later - Hamilton Dead

L.S.I. - Sally Simay Test

4.7.12 - Vibration Test. In addition, the unit shall be vibrated for I hour in each plane at a frequency of 125 cps and an acceleration of 10 g/s.

h.8 - Fungus Test - if fungi nutrient materials have been used in construction.

4.11.3 - Sand and Dust Test

- 4.1.1 Component qualification test mentioned above is to be performed by the vendor where requested by separate purchase order.
- h.1.2 Component wedesign and retest shall, be required at vendor's expense in the event of any failure of the unit undergoing the above mentioned tests.

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

SPEC NO. HS 1368

CODE 73030

- 4.2 The dealgn shall meet the embel was requiremente of the applicable Hamilton Standard drawn a which will make he applicable version approval.
- 4.3 Standards, Manerials and Wirkmanacky
- 4.3.1 Standars Farte

AN and MS standard parts and corresponding AN and MS part numbers, shall be total whenever suitable for the purpose. Commercial quality, or year designal standard parts, such as serews, ruts, bolts, washers, etc. may be used provided they are replaceable by AN and MS standard parts without alteration.

4.3.2 Materials

The materials used in this unit shall be of high quality, shall be compatible with P&WA 523, JP-5, JR150 or RJ-1 fuel under conditions specified in paragraphs 5.5.4, 5.5.7 and 5.5.9 and shall not be adversely affected when placed in a radiation field. The use of AMS 5600, 5601, 5630, 5631, and 5632 stainless steels and metals containing silver, copper, or cadmium which shall be in contact with fuel are problemated without written authority from Hamilton Standard.

4.3.3 Workman Filiti

The workcarchip and finish on all parts shall be in accordance with high grade manufacturing practices covering this type of aircraft equipment.

4.4 Data Flats

A data plate shall be attached to the unit and shall include the following information:

- (a) Manufacturer's name and trademark
- (b) Manufacturer's serial number
- (c) Manufacturer's patent and/or patent pending information
- (d) Maintfacturer's part number
- 4.5 Mockup
- h.5.1 A morkup of the remote trimmer shall be supplied to Hamilton Standard and submitted for approval of the engine manufacture.

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

HS SPEC NO_1368_

4.5.2 The models shall be he described with approved changes affecting the external configuration of the unit of the limit about a hallation.

4.6 Drawings

The remote trimmer manufacturer through the following drawings to Hamilton Stanfari prior to fabrication on accounters of the assembly:

- (a) Installation crawings
- (b) Sib-warins (if applicable)
- (c) Detail drawings (full sat)

4.7 Interchangeability

- 4.7.1 All parts of equipment components having the same park numbers shall be interchangeable or replaceable in accordance with, and to the extent required by MH-D-5128, and shall be manufactured in conformity with the provision of such specification.
- 4.7.2 Changes in design, dimensioning, or material that might affect interchangeability or performance of the unit shall be approved by Hamilton Standard prior to incorporation by the manufacturer.

4.8 Construction

The units shall be designed and constructed so that no parts work loose in service, and shall be built to withstand the strains, jars, vibration, and other conditions incident to shipping, storage, installation and service. Unless otherwise specified, all screw threads shall be in accordance with MIL-S-7742.

4.9 Processes

4.9.1 Welding

Weld construction is permissible. All welding shall be in accordance with MIL-W-8611 per HS 191.

4.9.2 Brazing

Brazed construction is permissible and shall comply with HS 782.

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS CONNECTICUT, U. S. A.

HS SPEC NO 1368

4.10 Handbooks and Instructions

When requested under separate purchase order, the vendor shall furnish to the purchaser:

- (a) Technical orders
- (b) Overhaul and maintenance handbook
- (c) Illustrated parts breakdown.

4.11 Development

The vendor shall agree to replace, without cost to Hamilton Standard, all defective parts of the first two delivered units, or parts whose failure could be attributed to a defect in any operational function of the same units. In addition, he shall agree to give Hamilton Standard the right to return, at no expense, all units which do not meet this specification in its entirety.

4.12 Engineering Changes

4.12.1 Definitions

4.12.1.1 Class I Changes

A Class I change shall be defined as a change in which any one of the following apply:

- (a) Model specification, control specification, control price, weight or delivery is affected.
- (b) Performance or durability is affected to such an extent that superseded parts, sub-assemblies, complete articles or complete assemblies must be reworked, replaced or discarded in service at or before the next overhaul.
- (c) The design change is recommended for any retrofit to delivered articles.
- (d) Complete interchangeability of installation or performance of the end item, or the complete component on the end item is affected.
- (e) Parts, sub-assemblies, or complete assemblies replaceable by the user are affected to such an extent that the superseded and superseding parts, sub-assemblies or complete assemblies are not directly and completely interchangeable with respect to installation or performance.

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

HS SPEC NO 1368

h.12.1.2 Class II Changer

A class II change shall be defined as miror changes not affecting the article ar debail parts in the sames muldined in hell-lel above.

4.12.2 · Identification

4.12.2.1 Detail Fart Numbers

The part number identification shall be changed on all parts and sub-assemblies afterted by a Class I change.

4.12.2.2 Assembly Numbers

The assembly number of the vendor's end item shall only be changed on a Chaes I change that affects the end item interchangeability with respect to installation and/or performance, and on any change requiring retrefit at or before next scheduled overband.

4.12.2.3 Data Flate

The data place will indicate the incorporation of all Class I change by a prefix or suffix revision to the vendor's end item part number or parts list number.

4.12.3 Approvals

4.12.3.1 Class I Changes

Official authorization wast be obtained from Hamilton Standard Purchasing Department prior to shipment of units incorporating a class I change. Authorization by Hamilton Standard will be based on Pratt and Whitney Aircraft approval of the change proposal. Requests for such approval should be submitted to the Hamilton Standard Purchasing Department in the form of Engineering Change Proposals. Firm copies of each proposal must be submitted. The proposal is to include the following:

- (a) Drawings and parts Lists to define the change
- (b) An add and cancel list
- (c) An indication of saleable and repairable items.
- (d) The estimated incorporation date
- (e) The applicable cost and change in price

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

HS SPEC NO 1368

4.12.3.2 Class II Changes

Hamilton Standard approval is not required for the incorporation of Class II changes, except that all specification changes must be coordinated with Hamilton Standard Engineering. Copies of all Class II changes with applicable revised drawings must be sent to Hamilton Standard for review after publication.

- 5.0 Detail Requirements
- 5.1 Weight

The weight of the complete remote trummer package shall not exceed 7.0 pounds. This weight shall be considered a maximum limit and not a target value. The final weight of the package must be justified.

5.2 Assembly Limitations

The unit shall be so assembled that it requires no brazing at overhaul.

5.3 Envelope and Installation Details

The configuration, dimension, and mounting shall comply with applicable H.S.D. drawings.

- 5.4 Life
- 5.4.1 The unit shall be designed for an engine operating life of 1000 hours under normal service conditions prior to overhaul.
- 5.4.2 The casting life expectancy shall be 10,000 hours.
- 5.5 Functional Requirements
- 5.5.1 Adjustments
- 5.5.1.1 The military adjustment shall be so designed that it will have a total adjustment range of not less than 10 turns, of which only 5 turns shall be available to the pilot for trim in-flight.
- 5.5.1.2 The Mil and Idle adjustment must be accessible for trim on the ground with the remote trim device in place.
- 5.5.1.3 The rate of operation of the military adjustment of the remote trimmer shall be 1 ± 1/4 RPM at 400 cps power supply.

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HAMILTON STANDARD OF UNITED AIRCRAFT CORPORATION

SPEC. NO. HS _36

CODE 73030

WINDSOR LOCKS CONNECTICUT, U. S. A.

| 5.5.1.4 | Tha | normal | operating | Èns | starting | <i>ใ</i> เดียนูนิธ | งนับ _{ที่} นั้น | عسدها المتاه | be 5 inch- | pounds |
|---------|-----|---------|------------|-----|------------|--------------------|--------------------------|--------------|------------|--------|
| | and | the dev | fice shall | nav | iladic a e | i Myus | | i i in | pounda. | |

- 5.5.1.5 The remote trim shall have adjustable Highly stops capable of setting the operating struke to within Appendiculation of the desired valve. The unit shall also have the caushility of locating the remote trim adjustment (commally 5 resultations) arowhere within the 10 revolutions of total adjustment with a position accuracy of Appendicular.
- 5.5.1.6 The point shall be provided with a rayoutile or mu hardwall brake to prevent rotation of the notice areafore foring proved-off periods. The brake shall be capable of braking an imput of 5 in-lbs.
- 5.5.2 Electrical Requiremends
- 5.5.2.1 Prwor Supply

The actuator shall be capable of satisfactory operation when supplied with 208 45% VAC, hoofso cps, 3 phase power.

5.5.2.2 Electrical Interference

The unif shall comply with electrical interference requirements per paragraph 3.11.2 of MILE-5007B. The compliance of the unit to this spec, shall be demonstrated by the ventur.

5.5.2.3 Durby Syr 19

The duty cycle of the unit shall be In minutes off.

5.5.3 Fuel.

The fuel supplied for cooling the unit shall be P&WA 523, however, fluids such as JP-6, JP-150, RJ-1, etc. should also be considered as possible fuels.

- 5.5.4 Fuel and Ambient Temperature Range
- 5.5.4.1 The unit shall be designed to operate satisfactorily with ambient air temperatures between -65°F and +1082°F, and fuel inlet temperatures between -65°F and +5.00°F.
- 5.5.4.2 The switch control assembly shall be designed to operate satisfactorily in an inhabitable environment. It shall be required to meet the temperature requirements of MIL-E-5272C procedure II for high temperature and procedure I for low temperature, the altitude requirements of MIL-E-5272C, and the vibration requirements of MIL-E-5272C procedure XII at room temperature.
- 5.5.5 Ambient Pressure

The ambient (macalla) prassure range is from .34 to 20 psia..

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5.5.6 Coolant Flow Rate

Fuel as specified in 5.5.3 may be supplied to the unit for purposes of imbrigating and cooling the internal gearing. Coelant flow at the temperatures specified in 5.5.4 shall not expeed 100 PPH at 150 psi supply.

5.5.7 Fuel Prenours

Fuel shall be supplied to the unit at a pressure of 150-1000 paig and returned at a pressure of 5-165 psig. The differential between the supply and return pressures shall never co less than 11.5 psi. The unit shall be designed for and shall demonstrate capability of withstanding, without structural failure or external leakage, the following pressures acting singly or in conjunction in any combination:

| | | Fuel Temp. |
|-----------------|----------------|------------|
| Inlet Pressure | 0 - 2200 psig | 700cr |
| Inlet Precioure | 0 - 1500 'prig | 500°F |
| Return Pressure | 0 - 500 psig | 500°F |

The high pressure some of the unit shall be capable of withstanding a proof pressure test of 1500 psig without permanent deformation or external leakage. The low pressure zone shall similarly be capable of withstanding a 300 psig proof test. The trimmer package shall provide the necessary orifice to limit the coolant flow rate as specified in 5.5.6.

5.5.8 Leakage

The external adjustments shall have a double seal with a vent to overboard drain in between. The maximum leakage to overboard drain shall not exceed 1 cc/min. There shall be no external leakage of the unit; that is, it must be drop-tight and exhibit no external wetting of the surface.

5.5.9 Fuel Contamination

The unit shall operate satisfactorily after fuel containing the following contaminant is passed through a ho micron filter provided in the fuel control.

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HAMILTON STANDARD

DIVISION OF UNITED AIRCRAFT CORPORATION

ION

SPEC. NO. HS

1368

GODE 73030

WINDSOR LOCKS, CONNECTICUT, U. S. A.

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First Combaninasi

(before 40 micron filtration)

| Combanic asity | Partition Sine | Grant Lay |
|--|---|---|
| Iron Orida | 0 - 5 mioroca | 15.77 gm/1000 gal. |
| Iron Caide | 5 = 10 ಚರ್ವಜಾಗಕ | .83 gm/1000 gal. |
| Sharp Silica Sand | 40-50 west. | .03 grv/1000 gal. |
| Shamp Sillima Sand | 50-200 mest. | .03 gn/1000 gal. |
| Prepared dirt conforming to A.C. Spark Plug Co. Part No. 1543637 (Coarse Arizona Road Dust) | C-5 microns 5-10 microns 10-20 microns 20-40 microns 40-80 microns 80-200 microns | .53 gm/l000 gal. .53 gm/l000 gal. .62 gm/l000 gal. l.02 gm/l000 gal. .02 gm/l000 gal. |
| U.S. Standard Staple No. 7 prime worthom limbers | As ground in a No. h Wiley Mill and someoned through a h um source. | .55 gm/l000 gal. |
| | | and her word |

Crude Naphthenic Acid

.33% by vol.

Salt Water in accordance with salt spray solution per MIL-E-5272

.01% entrained

NOTE: The 40 micron filter will filter out approximately 45% of the particles under 40 microns in size and 97% of the particles over 40 microns. If a filter finer than 40 microns is required, it shall be a part of the remote trimmer package.

6.0 QUALITY CONTROL

The inspection tests listed below define the extent of responsibility of the Hamilton Standard Quality Control Department under this specification. Control shall be established to insure compliance with the following paragraphs of this specification:

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HAMILTON STANDARD

DIVISION OF UNITED AIRCRAFT CORPORATION

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CODE 73030

WINDSOR LOCKS, CONNECTICUT, U. S. A.

6.0 (mondatabad) QUALITY CONTROL

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Promissos

Weight

5,3 Envelope and Installation Details

5.5.7.3 Rate

5.5.7.1 Trans

5.5.6 Coolar's Flow Rate

Fuel Pressure

5.5.8 Leakage

8.1 Cleaning

Packaging

7.0 ACCEPTANCE TESTS

An acceptance test of the remote trimmer shall be mutually agreed upon by the vendor and Hamilton Standard, and shall be subjected to final approval by the military service utilizing the engine. Ref. H.S. Spec 1350.

8.0 PACKAGING AND IDENTIFICATION FOR SHIPMENT

8.1 Cleaning

Each unit shall be thoroughly cleaned of dirt, sand, metal chips and all other foreign material during final assembly.

8.2 Packaging

The package shall be treated to insure protection against corrosion during shipment and storage in accordance with MII-P-116.

8.3 All parts shall be covered to exclude dirt and threads protected to prevent them from damage.

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WINDSOR LOCKS, CONNECTICUT, U. S. A.

8.4 All mounting faces shall be provided with temporary cover plates for protection from dire and damage.

- 8.5 Each individual package shall be durable and legibly marked with the following information in such a manner that the markings will not become damaged when the package is opened:
 - (a) Name of apparatus
 - (b) Model designation
 - (c) Manufacturer's name or trademark
 - (d) Purchaser's order number
 - (e) Manufacturer's drawing number
 - (f) Manufacturer's serial number

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT

H.S. 1368
Amend. /
Page 1 of 2
ECNN65111
Date: 7-7-6/

HS1368 "Procurement Specification for the JFC47 Fuel Adjustment Remote Trimmer"

Amendment___/

1. In paragraph 2.0 change:

| MIL - DJ 028 | ົບວ | MIL-D-70327 | |
|--------------|-----|---------------|-----------|
| MIL-S-7712 | tr, | 1.IL-S-771.2A | (12-2-59) |
| MIL-W-8611 | to | MIL-V-LOITA | (== = >), |
| MIL-P-116 | to | MIL-P-116C | |
| MIL-I-8500 | to | MIL-I-85COA | |

- 2. Change paragraph 4.1.2 to read:
 - "4.1.2 Component redesign and retest shall be required at vendor's expense in the event of any failure of the unit undergoing the above mentioned tests after completion of proof tests."
- 3. In paragraph 4.7.1 change "MH-D-5028" to "MIL-D-70327".
- 4. In paragraph 4.9.1 ohther lass centence to read "All welding shall be in accordance with MIL-W-8611 or HS 191."
- 5. Change paragraph 4.11 to resd:

"4.11 Development

The vendor shall agree to replace, without cost to Hamilton Standard, defective parts whose failure could be attributed to a defect in any operational function of the unit. In addition, he shall agree to give Hamilton Standard the right to return, at no expense, units which do not meet phase requirements in its entirety. (See detail B/P). "

- 6. In paragraph 5.1 change "7.0 pounds" to "7.2 pounds."
- 7. Change paragraph 5.5.1.2 to read:
 - "5.5.1.2 The manual Mil and Idle adjustment must be accessible for trim on the ground with the remote trim device in place and require not more than 45 in-lbs of torque for actuation."

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H.S. 1368
Amend. /
Page 2 of 2
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Date: 7-7-6/

HS1368 "Procurement Specification for the JFC47 Fuel Adjustment Remote Trimmer"

| Amendment | 1 |
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| | |

- 8. In paragraph 5.5.2.2 change "MIL-E-5007B" to "MIL-I-6181D".
- 9. In paragraph 5.5.2.3 change "10 minutes on, 10 minutes off" to read "5 minutes on, 5 minutes off."
- 10. In paragraph 5.5.4.2 change the first sentence to read: "A switch control assembly, if provided, shall be designed to operate satisfactorily in an inhabitable environment."
- 11. Change paragraph 5.5.6 to read:
 - "5.5.6 Coolant Flow Rate

Fuel as specified in 5.5.3 may be supplied to the unit for purposes of lubricating and cooling the internal gearing. Coolant flow at 85 ± 10°F shall not exceed 100 PPH at 150 psi across the trimmer."

12. To paragraph 5.5.8 add the following sentence.

"Retorquing of seals at no less than 100 hours of operation is acceptable."

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H.S. 1368
Ammend. 2
Page 1 of 1
E.C. 65620
Date: 9-23-61

H. S. 1368 "Procurement Specification for the JFC47 Fuel Adjustment Remote Trimmer"

Amendment 2

- 1. In the 1st sentence of paragraph 4.3.2 delete: "and shall not be adversely affected when placed in a radiation field."
- 2. In paragraph 5.5.221 change "400 ± 80 cps." to read: "380-520 cps."
- 3. In the first sentence of paragraph 5.5.2.2 delete: "paragraph 3.11.2 of."
- 4. In the 2nd sentence of paragraph 4.11 change: "phase" to read "the integrity test."
- 5. Change the last sentence of paragraph 5.5.8 from: "Retorquing of seals at no less than 100 hours of operation is acceptable."

to read:

"Retorcking of seals at no less than 100 hours of pressurized operation is

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H.S. 1368
Amend. 3
Page 1 of I
E. C. AZ68898
Date: 3-20-6>

| H.S. | 1368 "Procurement | Specification | for | the | JFC47 | Fuel | Adjustment | Remote | Trimmer* |
|------|-------------------|---------------|-----|-----|-------|------|------------|--------|----------|
| | 1 | Imendment 3 | | | | | | | |

- 1. In paragraph 2.0 add the following:
 - "HS236 Specification for Treatment of Metal and Metal Parts"
- 2. Add paragraph 4.9.3 to read:
 - 4.9.3 PAINTING

All exterior unmachined surfaces and tube O.D.'s except screws, nuts, receptacle, and surface under the protective plate on the mounting flange will be painted per HS236, Code 276.



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| PAGE | 2 | OF | |

1.0 GENERAL INFORMATION

1.1 SCOPE

> This specification covers methods and procedures of shimming the JFC-47 main engine control.

1.1.1 For purposes of this specification controls are grouped as follows:

Prod. S/N's 26628 thru 266h2 Group I Group II Prod. S/N's above 26642

1.2 EQUIPMENT REQUIRED

1.2.1 Assembly tools such as screw drivers and wrenches will be controlled by ass mbly floor supervision as to correct application in order to prevent parts mutilation yet do the job requirement.

1.2.2 Micrometer (1 inch) Depth Micrometer (dial indicator type) Feeler Gage

Hunter Force Gage (0 to 10 lb.)

557450T-22 Flyball Alignment Tool 569455T-16 Rigging Pin T-19 CIP weight Fixture

T-25 Tt2 Serve Motion Fixture

T-26 Ng Servo Transfer Fixture

T-27 CIP Dummy Cam

T-31 Integrating Piston Stop

T-35 CIP Shaft Locating Fixture

T-53 CDP Sensor Adj. Wrench

T-66 Ng Servo Dummy Cam

T-70 Roller Gage Block

T-71 Transducer & Integrating P.V. Fixture

T-75 Dummy Lever Paralleling Tool

T-76 Push Rod Adj. & Shim Fixture

T-77TTt2 Transfer Tool

1 20

T-78 Ng Pilot Valve Set Block

T-79 CIP Feedback Bracket Align.Tool

T-80 Speed Set Cam Follower Fix

T-147

Air Pressure & Gage (0 to 50 psig)

569455T-81 Tt2 Pivot Base Tool

T-82 Min Ratio Fixture

T-83 Tt2 Cam Follower Link Set Block

T-85 Dummy CDP Lever & CDP Paralleling Fixture

T-86 CDP Transfer Tool

T-87 CDP Bellows Setting Eixture

T-88 Trimmer Screw Locating Fix.

T-89 Mil. Push Rod Centering Tool T-90 CIP Stop Transfer Fixture

T-91 Dummy CIP Sensor Lever

T-92 Pull Rod Shimming Fixture

T-93 Worm Wheel Align. Tool

T-94 Proportional Gain Lever Fix.

T-95 Power Lever Fixture

T-96 Tt2 Servo Piston Lock

T-97 Ball Check Valve Fixture

T-5 Adapter, CBA & Integrating Piston

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|------------|-------------|
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2.0 INSPECTION REQUIREMENTS

The items marked with an asterisk (*) in this specification are inspection items and as such must be verified by inspection. Whenever initial shimming is changed to meet a functional requirement, the information must be recorded and witnessed as such on initial recording sheets.

3.0 RECORDING SHEETS

Assembly check list sheets as attached to this specification must be filled out where applicable. A copy of these sheets should accompany all units shipped.

NOTE: See appendix for an index of the various systems as listed in this specification.

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TITLE:

Use of Solid Shims

OBJECTIVE:

To determine the proper amount of solid shims to be used.

REFERENCE: None

1. When selecting shims for any application, always select dash numbers that will provide the minimum number of shims.

Ex. For an application where .060% shims are required and shim thicknesses of .020% and .030% are available, use two .030% shims as opposed to three .020% shims to make up the .060% shims needed.

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SPEC. NO. HS 1502 B CODE IDENT NO. 73030

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TITLE:

Run-in of PRV Sensor Assembly

OBJECTIVE:

To break-in the PRV Pilot Valve, Housing, Damper and Drive Gear.

REFERENCE:

Tool 569455-T-105

- 1. Assemble PRV Sensor in its Hydraulic Housing and install Tool 569455-T-105.
- 2. Adjust Pilot Valve for Position 1 defined as full Pilot Valve travel toward T.V. ΔP adjust end. With the PRV Sensor immersed in spindle oil (room temperature to 200°F, 10 micron filtration) drive the sensor at a speed of 3.500 to 4.500 rpm for four hours.
- 3. Upon completion of above four-hour run, adjust Pilot Valve for Position 2, defined as approximate zero position. At this position, repeat the hours and speed called out for Position 1.
- 4. Upon completion of Position 2 running, adjust Pilot Valve for Position 3, defined as .010-.020 less than full Pilot Valve travel toward the damper end. Repeat the hours and speed called out for Position 1.
- *5. Disassemble the PRV sensor and examine parts for distressed area following the 12 hours running. Abnormal wear or scuffing of bearing surfaces shall be cause for rejection, replacement of parts, and rerun of the 12-hour break-in.

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| | 1. | Install section Mount (of Ng appropriate to thumb sectioning | l spaces n of lin Cam Fol: pull roo riate th l nylon the ny: screw th the ass housing Tool a | c, cam finkage ho lower Po l sleeve weaded lon clam alon clam aread ex sembled ag until | ollower using. sitionin) so the hole(pro nd jamb p (norma tends be thumb so the thu nylon cl | lever and Tool at the covided for the later and the crew into the control of the crew into the control of the crew into the crew | on to the down on to the New thread to the New thread to the New thread | ne Ng sport of the pull root of the pull root of the pull root of the pull root of the bor | eation eed se tool sleev cool; a coximat servo the Can | rvo bore locates el de lock ely 1/2 bore of m Follow locks the | the jamb inch of | | |
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HB F-295.4

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

H.S. 1502 B No. c- Page 7

TEP ASSEM DESCRIPTION INS HUMB SCREW NYLON CLAMP CAM FOLLOWER POSITIONING rool SET UP FOR REMOVING PIN END PLAY" RECORD IN FLUSH WITH PIS FOR POSITIONING RECORD SHIMS CAM FOLLOWER SPACER LEVER GCAM FOLLOWER POSITIONING CINSTALL SHIMS HERE AFTER MEASURING ABOVE rook NOTE - POSITIONING TOOL SET UP FOR POSITIONING CIM FOLLOWER LEVER 15 REMOVED BEFORE VIEW A-A MEASURING FOR END-PLAY. SPEED SET CAN FOLLOWER FORTIUNING TOOL 569455.T-80

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| TITLE : | Shimming | of | Speed | Set | Cam | Follower | Lever | (Grown | TT | Controls | ١ |
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OBJECTIVE: To shim the cam follower lever to a specified neight above the centerline of the Ng speed servo bore; and to remove excessive pin "end-play" by the use of shims.

REFERENCE: Figure titled speed set cam follower positioning tool 569455T-80.

- 1. Mount the Cam Follower Positioning Tool on to the Ng speed servo bore (in lieu of Ng pull rod sleeve) so that the dowel pin of the tool locates in the appropriate threaded hole (provided for the pull rod sleeve).
- 2. Install nylon clamp and jamb nut onto Thumb Screw Tool; and lock the jamb nut to the nylon clamp (normal torque) so that approximately 1/2 inch of thumb screw thread extends beyond nylon clamp.
- 3. Insert the assembled thumb screw into the Ng speed servo bore of the linkage housing until the thumb screw threads into the Cam Follower Positioning Tool and the nylon clamp bottoms in the bore and locks the Positioning Tool in place with normal hand torque applied to the thumb screw.
- 4. Assemble pivot pin and cam follower into linkage housing.
- 5. Holding cam follower ball against the <u>Cam Follower Positioning Tool determine</u> the correct amount of shims needed between the <u>cam follower and linkage</u> housing with a feeler gage. Select a shim within **2.001**. Record ______.
- 6. Remove pivot pin and cam follower from linkage housing and install shim determined in para. 5, cam follower, spacer and pivot pin into linkage housing.
- 7. With the use of a feeler gage determine shims required between cam follower and spacer. Select shim within ±.001 Record._____
- 8. Remove pivot pin and install shim determined in para. 7 between spaces and cam follower.
- 9. Install pivot pin, pin retainer and screw.
- 10. Check cam follower end-play to insure that no error has been made in shimming.

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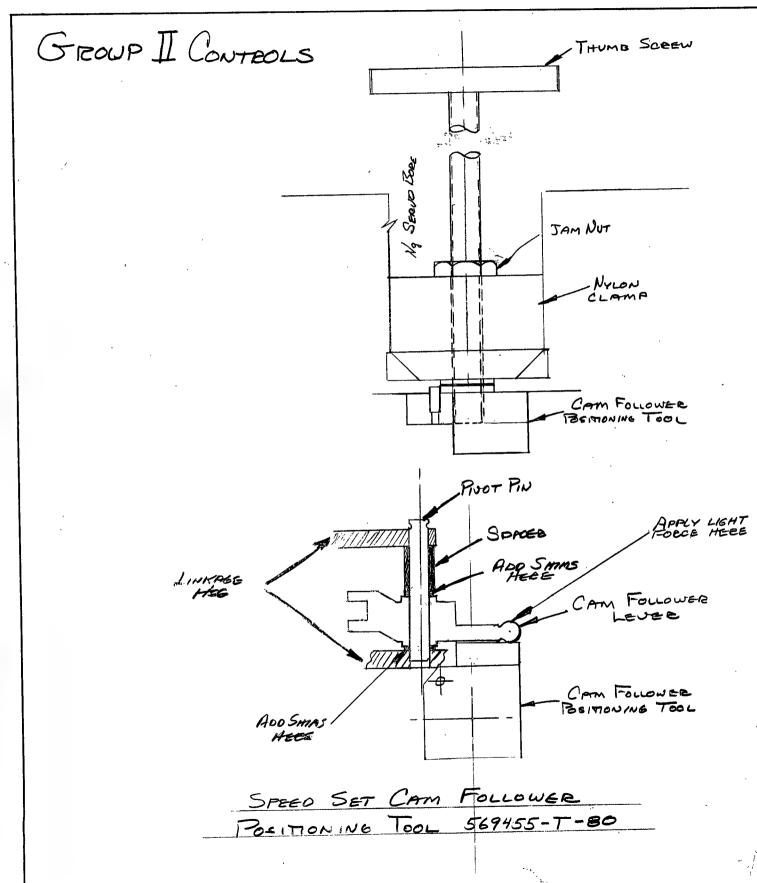
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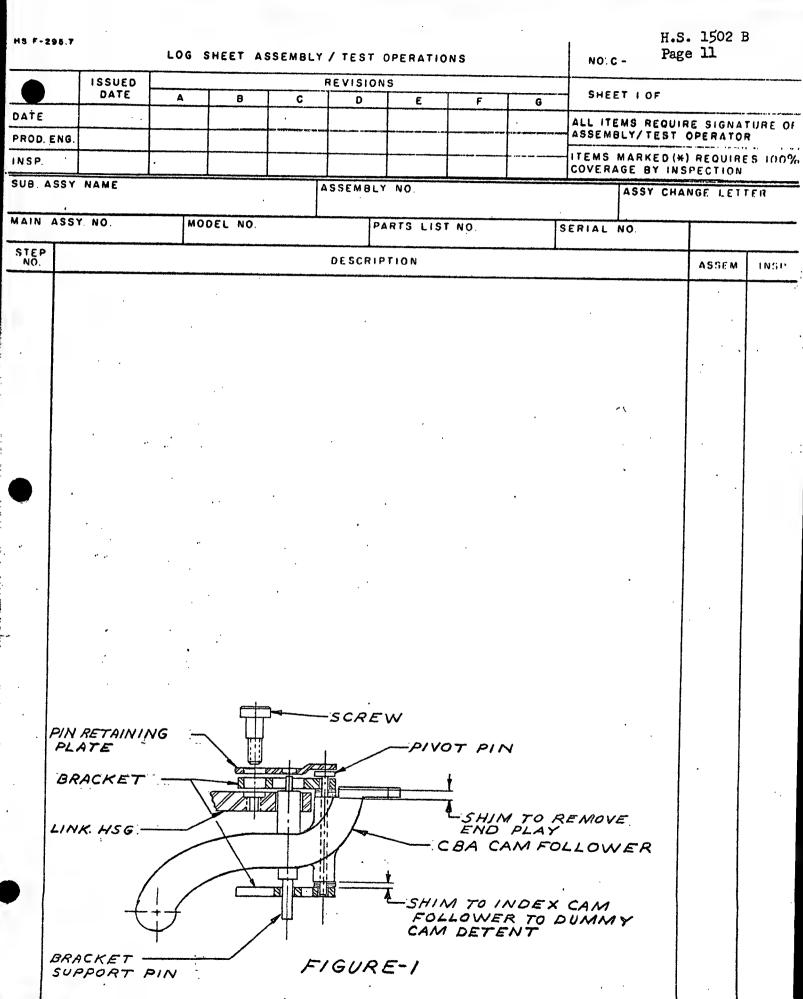
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| TITL | E 8 | Shimming of C.B.A. Cam Follower Lever. |
|------|---------------------------|--|
| OBJE | ECTIVE 8 | To shim the C.B.A. follower for position and to minimize "end play" |
| REFE | ERENCE : | Figure #1 |
| 1. | Install pivot pi | CBA camfollower lever into linkage housing and insert shouldered into provided bore and through the CBA follower. |
| 2. | housing dummy caunder the | Ng Servo Dummy Cam 569455T66 into speed servo bore of linkage so that Tt2 reset follower engages the appropriate detent of the m; and while maintaining this position, install sufficient shims e CBA cam follower lever to align the ball follower with the CBA in the dummy cam. Record shims (if any) required |
| 3. | shims) | hims beneath cam follower; and using depth micrometer (or actual measure from top of pin shoulder to surface of pad that retains n in linkage housing. |
| 4. | The resuplate an | lting dimension is the amount of shims required beneath retainer d pad surface to minimize pivot pin end play. Record shim stack |
| 5. | It is pe | rmissible to have .002 max. end play. The lever must move fter shimming is completed. |



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TITLE:

Run-in of Speed Servo

OBJECTIVE:

To break-in the speed servo in its linkage housing.

REFERENCE:

Figure Tilted: Speed Servo Cycling Schematic Tool:

- 1. Assemble speed servo into its linkage housing and install tool Preload the CBA and accel. limiting cam followers against the speed servo cam. The linkage housing half of the Tt2 servo is utilized for this break-in
- 2. Cycle the speed servo for 5000 cycles full translation and rotation and return with spindle oil (10 micro filtration) at room temperature to 200°F. Maximum hydraulic pressure to servos to be 500 psig.
- * 3. Following completion of 5000 cycles disassemble and examine the sensors, piston rings, and bores for distressed areas. Abnormal wear or scuffing of bearing surfaces shall be cause for rejection, replacement of parts, and rerun of the 5,000 cycle break-in.

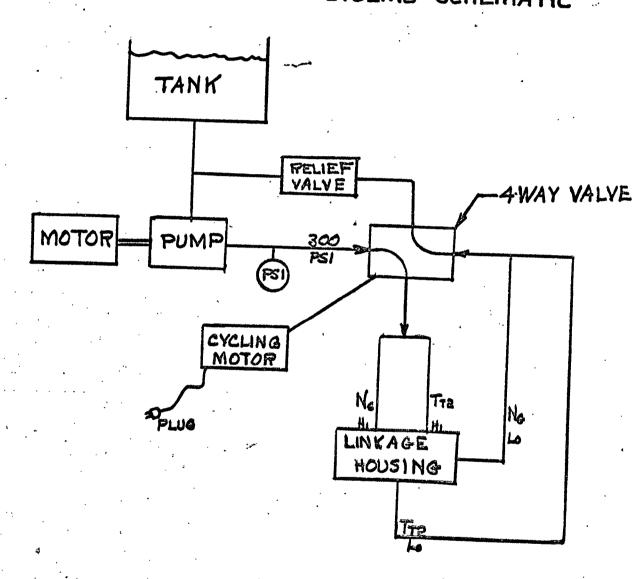
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PAGE_

SPEED SERVO CYCLING SCHEMATIC



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| TITLE 8 | Shimmina | ~ 6 | m | | ٠, | | |
|---------|------------|-----|-------------|-------|----------|-----|----------|
| | outimentag | OI | Temperature | Reset | Follower | and | Bracket. |

OBJECTIVE: To align the temperature reset cam follower with linkage housing slot; and to index the temperature reset linkage.

REFERENCE: Figure Titled: Tt2 Reset Bracket Indexing Fixture 569455-T-84.

- 1. Place edge of reset cam follower hub against the wall of the linkage housing slot that is farthest away from the centerline of the Tt2 servo bore.
- 2. While holding the reset follower lever in this position, use a feeler gage to determine the gaps between the reset follower lever and the adjacent casting walls. The shim stock should not move the follower more than .001 laterally away from slot and the total lateral motion at the follower should be less than .002.
 - a. Record shim stack in lever gap farthest from CL Tt2 servo boreb. Record shim stack in lever gap nearest the CL Tt2 servo bore-
- 3. Install the reset lever belicrank bracket onto linkage housing with a nominal shim stack-up under the bracket of .120. Final shimming of the bracket will be accomplished during the dry calibration per HS 1503.

45150ZE PAGE 15

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

NO. C-SHEET OF

DESCRIPTION ASSEM VIEW XX CAM FOLLOWER HUB ABAINST SIDE OF TLOT & NO SERVO BORE FOLLOWER RECORD RECORD & TEZ SERVE .TTZ RESET BRACKET INDEXING FIXTURE

567455-7-84

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| TITLE 8 | Shimming of Acceleration Lever Cam Follower. | | | | | |
|--------------------|--|--|--|--|--|--|
| OBJECTIVE 8 | To locate and shim the acceleration lever cam follower 180° from to Tt2 reset lever cam follower. | | | | | |
| REFERENCE: | Figure #1. | | | | | |
| l. Install housing | Ng Servo Dummy Cam 569455T66 into speed servo bore of linkage so that Tt2 reset follows falls into corresponding detent of | | | | | |

- dummy cam. The Tt2 reset lever is the master for angular location of other followers.
- 2. With no shims installed, position the limiting lever so that the cam follower will fall into the dummy cam detent 180° opposite from the reset follower detent.
- 3. Determine the gaps between sides of lever and the linkage housing lugs with a feeler gage (or the actual shims) and record gap B.
- 4. Install shims to obtain a total side play of .002" max. Lever must move freely after shimming is completed.

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TITLE 8 Adjustment and Shimming of C.D.P. System.

OBJECTIVE: 1) To set the C.D.P. system at null position by adjusting the C.D.P. bellows assembly and by shimming the C.D.P. multiplying lever and throttle valve pillot valve.

2) To compensate for an "body pressure sensitivity" as it affects the C.D.P. input lever.

REFERENCE: Figures Titled: Dummy CDP Lever & CDP Paralleling Fixture 569455T-85.

CDP Transfer Tool 569455T-86.

CDP Bellows Setting Fixture 569455T-87.

- 1. Install Dummy CDP Lever into the CDP Paralleling Fixture and tighten two screws "B" (use normal torque). Loosen set screws "C" & "D"; and insert the Set Block into the Paralleling Fixture and retain in position by tightening four screws "E" (use normal torque).
- 2. Turn in on Micrometer Head until it contacts pin "A" and pushes the Dummy CDP Lever to "bottom" against the Set Block. This will automatically place the Dummy Lever in a parallel position. Read the Micrometer. Record
- 3. From the actual CDP lever to be used in the control read the offset dimension (scribed on the side of the lever) which is required to compensate for any body pressure sensitivity. Reference: HS 1558. (See Note A, Sheet 20.
- 4. Remove Set Block from the Paralleling Fixture and adjust the Micrometer Head (from the parallel position) in the proper direction to set up the offset dimensions obtained in part 3.
- 5. Tighten set set screws "C" & "D" (use normal torque) while pin "A" is contacting the indexed Micrometer to preset the Dummy CDP Lever in the offset position.
- 6. Remove preset <u>Dummy CDP Lever</u> from <u>CDP Paralleling Fixture</u> and install it into the <u>CDP lever</u> bore in the linkage housing. Retain the <u>Dummy Lever</u> in place with two screws RB" (use normal torque).
- 7. Loosen set screws "G" & "H" on CDP Transfer Tool and install the CDP Transfer Tool into the CDF bellows bors of the linkage housing. Position the Cover and lock it in place with three screws "F" (use normal torque).
- 8. Allow the External Piston of the Transfer Tool to contact the bottom of the CDP beliews bore and lock it in this position with set screw "G" (use normal torque). Lower the Internal Piston of the Transfer Tool until it contacts pin "A" of the preset Dummy CDP Lever and lock it in this position with set screw "d" (use normal torque).

 CAUTION: Always preset the External Piston when the Internal Piston set screw "H" is loose. This will prevent interference with pin "A".
- 9. Remove preset CDP Transfer Tool from linkage housing (loosen and remove 3 screws "F") and install it into the CDP Bellows Setting Fixture so that the Internal Piston will contact ONLY pin "A" of the Adjustable Bracket. Retain the CDP Transfer Tool in this position with two screws "I" (use normal torque).
- 10. Move the two Adjustable Brackets on the CDP Bellows Setting Fixture until they contact the Internal and External Piston of the CDP Transfer Tool.

 Lock the Brackets in position with screws "J" & "K" (use normal torque).

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II. Group I Controls

Remove the CDP transfer tool from the CDP Bellows Setting Fixture and install the CDP bellows assembly in its place. Insert the asbestos gasket under the bellows flange that will be used to curtail air leakage during the assembly of the control.

Group II Controls

Remove the CDP Transfer Tool from the CDP Bellows Setting Fixture and install the CDP bellows assembly in its place. Insert the washer under the bellows flange.

Procedure for adjusting the CDP bellows when NO CDP lever offset is required to compensate for body pressure sensitivity:

Place 2.3# weight on motor bellows (pilot the weight between the three dowels) and adjust the threaded bolt until the pin firmly contacts the slot in the Adjustable Bracket. Suspend 6.5# weight from the evacuated bellows stem (by means of threaded Block); and while maintaining the 2.3# weight on the motor bellows, adjust the evacuated bellows on the threaded bolt until the bellows contacts the lower Adjustable Bracket.

13. For CDP systems requiring CDP lever offset to compensate for body pressure sensitivity, the following procedure should be used:

Obtain the load that is required to move the CDP lever to the offset position (per HS 1558 the required offset weight is scribed on the actual CDP lever along with the offset position). If the CDP lever requires an offset towards the evacuated bellows, subtract the scribed weight from 2.3# and use a Hunter Force Gage to set up the remaining load on the motor bellows. While maintaining the Hunter Force Gage load, adjust the threaded bolt until the pin seats in the slot of the Adjustable Bracket. Remove Hunter Force Gage and use Clamp and two screws "L" (use normal torque) to retain pin in Bracket slot. Suspend 6.5# weight from evacuated bellows stem and adjust the bellows on the threaded bolt until the bellows courtects the lower Adjustable Bracket.

If the CDP lever requires an offset towards the motor bellows, place 2.3# weight on the motor bellows and adjust threaded bolt until pin seats in slot. Subtract the scribed weight from 6.5#3 and while maintaining the 2.3# weight on the motor bellows, use a Runter Force Gage to set up the remaining load on the evernated bellows (by means of threaded Block) and adjust the evacuated bellows until it contacts the lower Adjustable Bracket. (See Note A. Sheet 20).

CAUTION: Load adjustment should always reduce the preload (2.3# and 6.5# respectively) on either bellows to insure adequate bellows life.

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li. Return to linkage housing (with the preset <u>Dummy CDP Lever</u> still in position) and use a small "C" clamp to retain Roller <u>Gage Block 569455-T-70</u> and CDP multiplying lever assembly on to the <u>Dummy Lever</u>. Use actual shims to determine the gap between the CDP multiplying lever bracket and the linkage housing mounting pad. Make sure that the multiplying bracket pivots are making contact during the shimming.

Record shim stack

- 15. Install assembled spider housing into the linkage housing (3 screws normal torque). Set up null position on the throttle valve pilot valve by inserting Rigging Pin 569455-T-16 into appropriate slot and compressing the pilot valve against the spring (out direction) until the pilot valve land contacts the Rigging Pin. With the pilot valve held against the Rigging Pin, insert the Feeler Gage between the pilot valve tip and the mating pad of the multiplying lever (which is still clamped in position on the Dummy CDP Lever) and determine the gap. Refer to the matched valve data sheets and obtain the true null position of the throttle valve pilot valve as referenced from the Rigging Pin. If the null position is listed as "OUT" from Rig Pin position, ADD the specified amount to the Feeler Gage dimension. If the null position is designated as "IN" from Rig Pin position, SUBTRACT the recorded amount from the Feeler Gage dimension.

 Record the computed shim stack
- 16. Remove the Dummy CDP Lever from the linkage housing and install properly shimmed CDP multiplying lever assembly (see part 14); adjusted CDP bellows assembly (per parts 12 or 13); and the CDP input lever. Insert the proper amount of shims (part 15) under the T.V. pilot valve tip.
- 17. With the CDP bellows and cover installed in the linkage housing, pressure test the assembly by putting 100 psi air into the motor bellows and checking for leaks around the cover and out the overboard drain line. No leakage is permitted.

NOTE A

Make the following correction to the $^{n}x^{n}$ or offset dimension and the $^{n}w^{n}$ or load req'd for offset to the $^{n}x^{n}$ and $^{n}w^{n}$ scribed on the side of the lever. This correction, which follows, must be used in paragraph 3 and 13:

- (a) Add +.Oll" to "x" dimension of lever.
- (b) Add +.8 lb. to Www of lever.

NOTE B

On some levers accepted on MRO the offset dimension will be more than .010. (Reason for being put on MRO) when such a lever is used, do not add the full .011 as specified in Note A (see Amendment 1), but add only enough to bring the offset dimension to .021.

From the actual plot of the lever find the "w" or weight required to establish the offset dimension of .021 and use this as the "w" or offset weight when adjusting the CDP bellows in the fixture.

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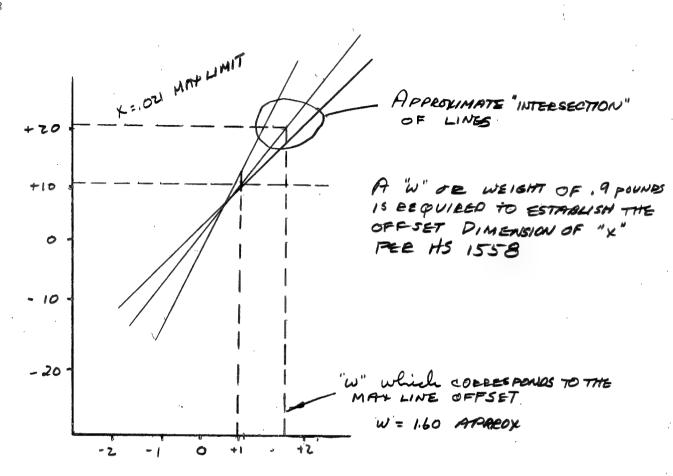
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WINDSOR LOCKS, CONNECTICUT, U. S. A.

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EXAMPLE:



When deviating from HS 1502A and following the note concerning MRO levers the weight should be obtained from the curve as illustrated; that is, $1.60 - .90 \pm 0.70$ pounds.

NO. C-1502B LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION) SHEET OF EP 10. ASSEM DESCRIPTION INSP -MICROMETER < RECORD READING SET -DUMMY CDP LEVER BLOCK IF REQ'D DUMMY COPLEVER & ZD, COP PARALLELING FIXTURE (OPPOSITE 569455-7-85 LINKAGE HSG DUMMY COP LEVER COP TRANSFER TOOL 569455-7-86

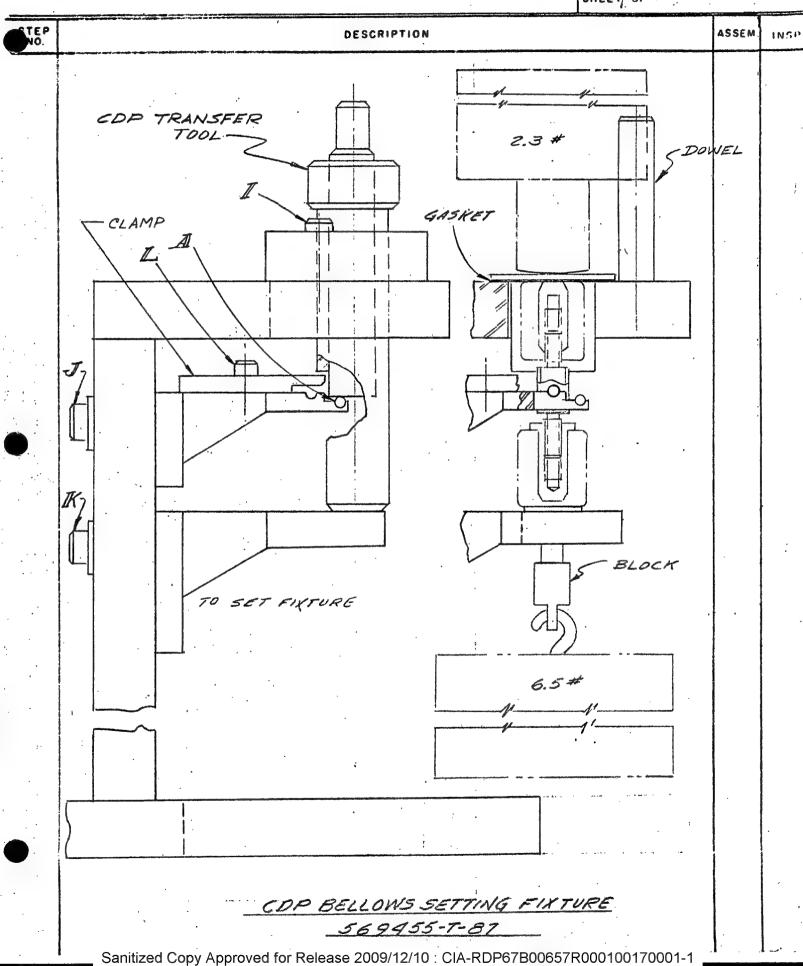
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LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

H.S. 1502 B

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DIVISION OF UNITED AIRCRAFT CORPORATION WHICHOE LOCKS, COMMETICAT, N. S. A.

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Page 24

TITLE:

Adjustment of Min. Ratio

OBJECTIVE:

To establish a .057° distance between the centerline of the Wf/PJ rollers and the centerline of the T.V. multiplying lever pivot.

REFERENCE:

Figure Titled: Min. Ratio Fixture 569455-T-82.

- 1. From the side of the T.V. multiplying lever read scribed dimension G.
- 2. Using formula (shims = Dim. G 2.478) determine shims required and insert them into Min. Ratio Fixture and tighten screw A (use normal torque).
- 3. Install Min. Ratio Fixture on to T. V. multiplying lever pad in linkage housing and secure in position with the two screws provided (use normal torque).

 NOTE: Use shims (recorded on CDP system shimming part lk page 11) under the Min Ratio Fixture to compensate for P3 lever offset.
- 4. Install spider housing into linkage housing (3 screws normal torque) and set the droop lever at min ratio position.
- 5. Set the min ratio adjustment in its midposition and torque the locking screw to 30-32 in-lbs. using the head end (internal hex).
- 6. Shim under the roller head assembly to make the rollers fall into the cradle of the fixture.
- 7. Final adjustment of the min ratio (using eccentric screw) will be made later per the min-max line dry calibration procedure per specification HS 1503.

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SPEC. NO. HS ______1502B

DIVISION OF UNITED AIRCRAFT CORPORATION

CODE IDENT NO. 73030

WINDSOR LOCKS, CONNECTICUT, U. S. A.

TITLE:

Shimming of throttle valve position adjustment (Group II controls only).

OBJECTIVE:

To ensure correct temperature compensation by incorporating required

number of shims.

REFERENCE: None.

- 1. Assemble the adjusting retainer, push rod, spacer and headless pin with .075 amount of shims.
- 2. Ensure that the spring retainer is positioned against the top of the counterbore in the housing.
- 3. Screw in the assembly referred to in 1. until the push rod just butts on the spring retainer.
- 4. Measure the distance from the top of the adjusting retainer to the housing face (Dimension A).
- 5. Measure the corresponding distance from the face on the cover to the bottom of the counterbore in the cover (Dimension B).
- 6. Amount of shims to be removed
 - = A B + .006
- 7. Remove shims specified in 6. and complete assembly.

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| | 5. Disassemble cam and pull rod assembly and install shims. Complete assembly of cam and pull rod. Again, the nut retaining the thrust bearing should be torqued only finger tight. 6. CIP Cam and Pull Rod Assembly: Install the CIP cam assembly into the Pull Rod Shimming Fixture. Place the cam detent and the pull rod trunnion pin slot over the appropriate pins of the Fixture and tighten thumb screw "A" over the cam (use hand torque). Next, move the adjustable block of the Fixture until the thrust bearing is against the lower cam retainer and lock thumb screw "B" in place (use hand torque) while maintaining cam assembly in this rigid position. 7. Using a depth micrometer, measure from the edge of the two protruding dowel pins (of the Fixture) to the surface marked "CIP Cam" on the adjust | | | | | | | nion pin crew "A" of the r and cam | | | | | |
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Sanitized Copy Approved for Release 2009/12/10 : CIA-RDP67B00657R000100170001-1 H.S. 1502B Page 28 NO. C-LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION) SHEET! OF TEP DESCRIPTION ASSEM able block. This dimension is the number of shims required between the sleeve spacer and the shoulder of the pull rod. Record CIP shim stack 8. Disassemble cam and pull rod assembly and install shims. Complete assembly of CIP cam and pull rod. Again, the nut retaining the thrust bearing should be torqued only finger tight. RELORD SHIM RED. GAGE SURFACE SLEEVE SPACER C.A.P. OR NA CAM ASSEMBLY PULL RUD DOWEL -PINS ADJUSTABLE BLOCK PULL ROD SKIMMING FIXTURE GAGE SURFACE NA GAM INSTALL SHIMS HERE RECORD SHIM REO. AFTER DETERMINING AMOUNT REO'D. PULL ROD SHIMMING FIXTURE 5694557-92 Sanitized Copy Approved for Release 2009/12/10: CIA-RDP67B00657R000100170001-1

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HAMILTON STANDARD

SPEC. NO. HS _____1502B

DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

CODE IDENT NO. 73030

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TITLE:

Shimming of droop lever assembly

(Group II controls only)

OBJECTIVE: To control axial clearance of droop lever assembly in spider and guide hous-

ings.

REFÉRENCE: None.

1. Assemble guide housing to spider housing.

- 2. Assemble ball bearings into droop lever making sure that bearings are completely bottomed in their droop lever bores. Also install shims adjacent to the droop lever bearing so that shims protrude slightly beyond end face of droop lever.
- 3. Install droop lever (with bearings), pivot shaft, spacer, and shims into spider housing and guide assembly. The spacer is located between the spider housing and the droop lever bearing.
- 4. Move droop lever axially to remove end play between it and spider housing.
 Using feeler gage, measure clearance between shims and guide housing. Determine shimming required to produce .000-.003 axial clearance.
- 5. Re-assemble with shims determined in 4. Check end play with feeler gage to assure it is .000 to .003 and make certain that droop lever operates freely.

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SPEC. NO. HS ____1502B

DIVISION OF UNITED AIRCRAFT CORPORATION

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WINDSOR LOCKS, CONNECTICUT, U. S. A.

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TITLE: Shimming of Ng Speed Governor and Pilot Valve

OBJECTIVE: To shim the Ng pilot valve for null position.

REFERENCE: Figure Titled: Set Block 569455-T-78

- 1. Install Flyball Alignment Tool 557450-T-22 into assembled governor head and Ng pilot valve set and insert Ng pilot valve into the spider housing.
- 2. Install the sleeve bushing into the bearing housing and install the bearing housing on to the spider housing (3 screws normal torque) so that the sleeve bushing pilots on the stem of the governor weight head.
- 3. Place Set Block over the stem of the Ng pilot valve: and while disc Z is contacting spider housing surface, adjust screw X in set block until pilot valve, governor head, and sleeve bushing "bottom" in bearing housing. Tighten lock nut Y (by hand), and remove Set Block from spider housing.
- 4. Using a depth micrometer, determine depth from surface of <u>Set Block</u> to where adjustable screw X contacted the shoulder of the Ng pilot valve. Call this depth DIM A and record
- 5. From the match data sheets for the spider housing, obtain the "S" dimension for the Ng pilot valve at null position.
- 6. Use formula (Shims = S A) and compute shims required to set the Ng pilot valve at null position. Record shim stack
- 7. Remove Flyball Alignment Tool from governor head and install required shims in the area between the sleeve bushing and the bearing housing.

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HAMILTON STANDARD

SPEC. NO. HS 1502 B

CODE IDENT NO. 73030

DIVISION OF UNITED AIRCRAFT CORPORATION

WINDSOR LOCKS, CONNECTICUT, U. S. A.

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TITLE:

Run-in of Spider Housing Pilot Valves.

OBJECTIVE:

To break-in the Ng, Tt2 and T.V. Pilot Valves over their full

operating ranges.

REFERENCE:

Tool

- 1. Install the Spider Housing Assembly containing the Ng, Tt2, and T.V. Pilot Valves and Gear Train in tool
- 2. Adjust Ng, Tt2, and T.V. Pilot Valves for Position 1 such that the gears run against their respective spider housing faces. With the spider housing immersed in spindle oil (at room temperature to 200°F; 10 micron filtration) drive the gear train at a speed of 3,500 to 4,500 rpm for four hours.
- 3. Upon completion of above four-hour run, adjust Ng, Tt2 and T.V. for Position2, defined as the approximate null position for each valve. At this position, repeat the hours and speed called out for Position 1.
- 4. Upon completion of Position 2 running, adjust Ng, Tt2 and T.V. Pilot Valves for Position 3, defined as the point at which the full diameter of the pilot valve is flush to .020" below the cast surface of the spider housing at the valve tip end. Repeat the hours and speed called out for Position 1.
- Following the 12 hours total running, disassemble pilot valves and gear train and examine parts for distressed areas. Abnormal wear or souffing of bearing surfaces shall be cause for rejection, replacement of parts and rerun of the 12-hour break-in.

Sanitized Copy Approved for Release 2009/12/10: CIA-RDP67B00657R000100170001-1 HS F-298.7 LOG SHEET ASSEMBLY / TEST OPERATIONS H.S. 1502 B NO.C -Page 33 ISSUED REVISIONS DATE A D SHEETILOF Ε G ALL ITEMS REQUIRE SIGNATURE PROD ENG ASSEMBLY/TEST OPERATOR INSP ITEMS MARKED (*) REQUIRES 100% COVERAGE BY INSPECTION SUB ASSY NAME ASSEMBLY NO ASSY. CHANGE LETTER MAIN ASSY. NO. MODEL NO. PARTS LIST NO. SERIAL NO. STEP DESCRIPTION ASSEM INSP TITLE: Shimming of CIP Worm Wheel and Shaft. To properly align worm wheel with worm gear shaft; and to eliminate CIP shaft and spur gear end play. REFERENCE: Figure Titled: Worm Wheel Aligning Tool 569455-T-93. Place spider housing casting on to gage surface of Worm Wheel Aligning Tool so the three mounting lugs contact the gage surface. "Line-up" the spider housing mounting lug (that contains the CIP worm wheel shaft bore) with the slot in the Aligning Tool; and insert the CIP worm wheel and shaft assembly (with shim cup in place) through the aligned bore and slot. 3. Slide the spider housing along slot of Aligning Tool until worm wheel is directly under the gage plate of the Aligning Tool. 4. Use feeler gage to determine gap between spider housing lug and bottom of the shim cup when the worm wheel is making full contact with the gage plate. Do not insert pin through shim cup and worm wheel boss (during measuring operation) because tolerances may prevent shim cup from contacting the bottom of the worm wheel boss. An alternate method would be to keep adding shims into shim cup until worm wheel contacts the gage plate. Record shim stack Insert the shims (determined in part 4) into worm wheel shim cup and install pin and lockwire. Build up the spider housing and the bearing housing. Mount the worm gear shaft into the spider housing to match with the CIp worm wheel and install bearing housing on to the spider housing (3 screws - normal torque). Install spider housing into linkage housing so that worm wheel shaft passes through mounting lug in the linkage housing. Retain spider housing to linkage housing (3 screws - normal torque) and install CIP spur gear, pin and shim cup on other end of worm wheel shaft. 8. Add sufficient shims into the spur gear shim cup to control worm shaft end play (between spur gear shim cup and linkage housing lug) to .002" Record spur gear shim stack

Sanitized Copy Approved for Release 2009/12/10: CIA-RDP67B00657R000100170001-1 H.S. 1502 B SHEET/OF Page 30 LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION) ASSEM. DESCRIPTION INST WORM WHEEL WITH SHIMS IN PLACE POR STEP 1. LINKAGE HOUSING MOUNTING LUGS SPUR GEAR SHIM STEP 2 SHIMMING OF SPUR GEAR SPIDER HOUSING-WORM WHEEL SHIM CUP-SPIDER HOUSING SHIM GAP

> WORM WHEEL ALIGNING TOOL 569455-T-93 Sanitized Copy Approved for Release 2009/12/10 : CIA-RDP67B00657R000100170001-1

STEP 1

SHIMMING OF WORM WHEEL

MOUNTING LUG

SHAFT

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HAMILTON STANDARD

1502B SPEC. NO. HS

DIVISION OF UNITED AIRCRAFT CORPORATION

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6. Add the gaps (measured in part 4 and 5 above) together and divide by two. This will give the average amount of shims required to properly position the trimmer housing with respect to the trimmer blades.

Record shim stack

- 7. Remove the trimmer housing from the Linkage Housing Plate and install the military push rod (stripped down) into the trimmer housing bore.
- 8. Install Military Push Rod Centering Tool through the pin hole in the military push rod (by means of pin B); and tighten nut A against the military push rod (use hand torque). Remove pin B from Centering Tool after nut A is tight against military push rod.
- 9. Position trimmer housing (with shims determined in part 6) into appropriate bore in the linkage housing and install speed trim lever and speed set cam follower (engage cam follower with trim lever).
- 10. Install Power Lever Fixture on to the linkage (3 screws normal torque) housing, and insert the power lever and cover assembly on the dowel pins of the fixture (at one end) 3 screws - normal torque, and the bearing in the linkage housing (at the other extremity).
- 11. Set the power lever to military position (65°) and move trimmer housing and shims laterally on the linkage housing parting surface until the Centering Tool contacts the roller of the speed trim lever. This will align the centerline of the military push rod with the speed trim lever roller at military power lever position.
- 12. Place scribe marks on the trimmer housing mounting flange and the linkage housing to define the alignment established in part 11.
- 13. Remove Centering Tool from trimmer housing and power lever fixture from linkage housing.

HS F-295.8

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

H.S. 1502 B. No. c- Page 37 SHEET OF Page 37

DESCRIPTION ASSEM INSP GROUP I CONTROLS DOWEL. RIMMER SCREW LOCATING FIXTURE SCRIBED & RETAINING SCREW INKAGE HSG PLATE TRIMMER SCREW LOCATING FIXT. LINKAGE HSG PLATE RECORD 569455-T-88 RECORD

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HS F-295.6

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

NO. C-SHEET OF

H.S. 1502 B Page 38

EP 0. DESCRIPTION ASSEM. INSP GROUP I CONTROLS OWER LEVER ASS'Y POWER LEVER FIXTURE LINKAGE HSG P/S MILITARY PUSH ROD PUSH ROD. CENTERING TOOL \mathcal{B} POWER LEVER BRG MILITARY PUSH ROD CENTERING TOOL 569455.T-89 POWER LEYER FINTURE 569455-T- 95

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TITLE: Alignment and shimming of trimmer housing (Group II Controls)

OBJECTIVE: To provide alignment of adjusting screw retainer with servo housing, trim lever roller with idle plunger pin, and trim plungers with droop linkage.

REFERENCE: Sketches and Power Lever Fixture 569455-T-95.

- 1. To determine required amount of shim between flange on trim plunger housing and servo housing measure following dimensions: Ref. sketch 1.
 - a. Distance from servo housing parting face to centerline of trimmer housing clearance diameter. Record as dimension "A".
 - b. Distance from centerline of trim adjusting screw retainer (at click-lock end) to linkage housing side of plunger housing flange. Record as dimension "B".
- 2. Determine required amount of shims by using equation "A"-"A" = "C" when "C" equals required amount of shims. Record dimension "C".

The purpose is to center the adjusting screw retainer in the clearance hole in the servo housing.

- 3. After determination of amount of shims required in step 2 install mil trimmer housing assembly and shims onto linkage housing. The trimmer housing must be positioned on its mounting surface to accomplish the following: Ref. sketch #2.
 - a. Set the trimmer lever roller center line .001 to .005 below (toward plunger) the centerline of the trim platform pin in the IDLE trim plunger with power lever at idle position. (Note: Tolerance is ±.005 on center distance on plunger and on location of roller in lever).
 - b. Place the O.D. of the trim screw retainer in the center of the clearance hole in the servo housing. (Note: Shims locate housing in one direction only.)
 - c. Position the centerline of the mil and idle trim plungers 1.170 ± 005 from of T2 servo datum plane to provide proper relationship between trimmers and droop linkage. This may be done by using a .013 spacer between the machined flat on the trim plunger housing and the support lug on the linkage housing.
- 4. After trimmer housing assembly has been properly positioned on the linkage housing, tighten four mounting bolts (normal torque) and lockwire.

Title: Shimming of military and idle pushrod assemblies.

Objective: To shim the military and idle pushrods to obtain correct Wf/P3 valves.

Reference: Figures titled: Throttle valve fixture 569455T-46

Cam Motion Tool 569455T-55 Power Lever Fixture 569455T-95 HSF-755.1A 5/61

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- 5. Insert a nominal amount of shims in each of the trim plunger assemblies (Idle = 040 ref.-Mil = .080 ref.)
- 6. Using the nominal thickness mil and idle trim cams (dash 1) adjust idle trim screw full CCW and set mil trim screw in midposition (approximately 6 turns from full CCW position). The shimming of the mil and idle plungers must be conducted with the trim cams set at these positions.
- 7. Install T.V. Fixture 569455T-46 on parting surface of linkage housing (hydraulic side) (2 screws normal torque) and index dial indicator on Wf/P3 rollers (put droop lever at min. ratio position and set dial indicator to read .057").
- 8. Install stop-washer and "Spiral-Lock" ring (to prevent Ng cam overtravel) into Ng servo bore, and engage the Cam Motion Tool on to Ng speed servo cam (one screw normal torque). Rotate cam until indexed dial indicator 569455T-25 on Tt2 piston reads 0.953 (59°F). Set dial indicator stop for this 59°F position.

Note: Tt2 dial indicator should be indexed by placing the Tt2 reset follower at 0°F end of Ng cam slot and setting 1.025" on dial indicator; or by placing reset follower at 850°F end of Ng cam slot and setting 0.490" on dial indicator.

An alternate method of setting the 59°F temperature position is to measure the temperature piston-to-linkage housing parting-surface height with the Tt2 reset follower at the 0°F end of Ng cam slot; and then, moving the piston .072" (1.025 - 0.950") towards the linkage housing. Use Tt2 Servo Lock Tool 569455T-96 and one screw (normal torque) to retain the Tt2 servo piston position during shimming operation.

- 9. Shimming of Idle Trimmer (Due to the interaction problem, the idle trim plunger must be shimmed before the military plunger.)
 - a. Set the power lever to idle position (13° to 15° power lever angle). Rotate power lever slightly each way and check to make sure that power lever cam follower is on constant radius idle flat.
 - b. Set Ng cam to idle speed position by inserting correct idle speed gage block into Cam Motion Tool.

This is accomplished by installing the mounting plate of Cam Motion Tool 569455T=55 on to the mounting flange of the linkage housing (2 screws - normal torque); and while keeping the Tt2 reset follower in the slot detent of the Ng cam, positioning knurled plate flush against mounting plate and tightening lock nut (normal torque). With lock nut set, the knurled plate is pulled back to allow insertion of idle speed gage block and then pushed tight against mounting plate to establish idle speed position. The desired settings for speed and W_F/P₃ for the particular control being assembled must be checked to ensure that correct values are being used. This may be done by checking appropriate paragraphs of wet calibration specification (HS1234 reference).

c. With temperature servo position, power lever angle and speed servo position maintained per Paragraph 8, 9a, and 9b, check W_F/P₃ roller position on dial indica tor indexed per Paragraph 7. Sanitized Copy Approved for Release 2009/12/10: CIA-RDP67B00657R000100170001-1 HSF-755.1A 5/61

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9. (Continued)

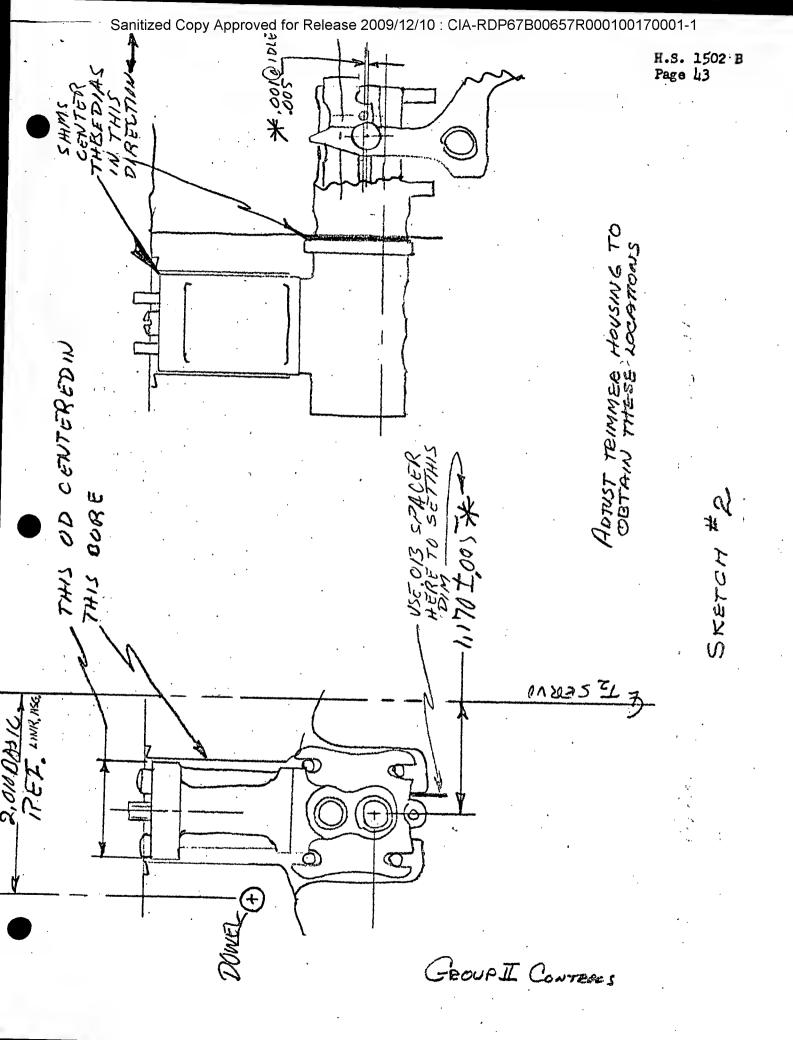
- A correctly shimmed idle plunger should give a rotter position which to desired ratio value as determined from a curve of $W_{\rm F}/P_{\rm 3}$ versus roller travel (F4649 reference). For example, wet calibration specification requires 130 $W_{\rm F}/P_{\rm 3}$ d. A correctly shimmed idle plunger should give a roller position which corresponds that roller position at 130 WF/P, is . Correctly shimmed plunger will place rollers at this position.
- e. If the roller travel is incorrect, add or subtract shim in the idle plunger to obtain correct value. Adding shims decreases roller travel. Sensitivity is about .002 roller travel per .001 shim.
- 10. Shimming of Military Trimmer
 - a. Set the power lever to military position (60° to 75°).

Note: Do not go to maximum power lever angle because power lever cam has contour change at about 1100 angle.

- b. Set Ng cam to military speed position by inserting correct military speed position by inserting correct military speed gage block into Cam Motion Tool per procedure outlined in Paragraph 9b.
- c. With temperature servo position speed servo position and power lever angle maintained per Paragraphs 8, 10a, and 10b, check $W_{\rm p}/P_3$ roller position on dial indica-
- d. A correctly shimmed military plunger should give the desired roller position as defined in Paragraph 9d.
- e. If roller travel is incorrect, add or subtract shims in the military plunger to obtain correct value. Adding shims will decrease roller travel. Sensitivity is about .003 roller travel per .001 shim.

GROUP II CONTROLS

SKETCH # 1



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| | 6. | Rod Ad | ijus tir | ng a | and Shim | ming I | ixturend pl | e ar | nd secur | re in pl fixture | lace wi until | th 2 sci the pir | the Push Yews "T" o in the "F" by | | · |
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| 7. | Install pin "G" into push rod assembly bore and retain in position by inse | rting |
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| | Gage Pin "H" through pin holes in push rod and pin "G". Place block over | dowels |
| | of end plate and retain in position with two screws "I" (normal torque). | Install |
| | Nut "J" on to Pin "G" and screw (by hand) until push rod is held in rigid | position. |

- 8. Using an Indicating Depth Micrometer, determine gap between Gage Pin "H" and surface of the end plate "Y" record. Add depth micrometer reading to .130" dimension scribed on end of Gage Pin and call the sum DIM. A = record.
- 9. Group I Controls Obtain scribed dimension from evacuated diaphragm assembly.

 (Call it DIM. B = record.) Subtract DIM. A from DIM. B. The result is the amount of shims required between the evacuated diaphragm assembly and the temperature cover housing. Record the shim stack

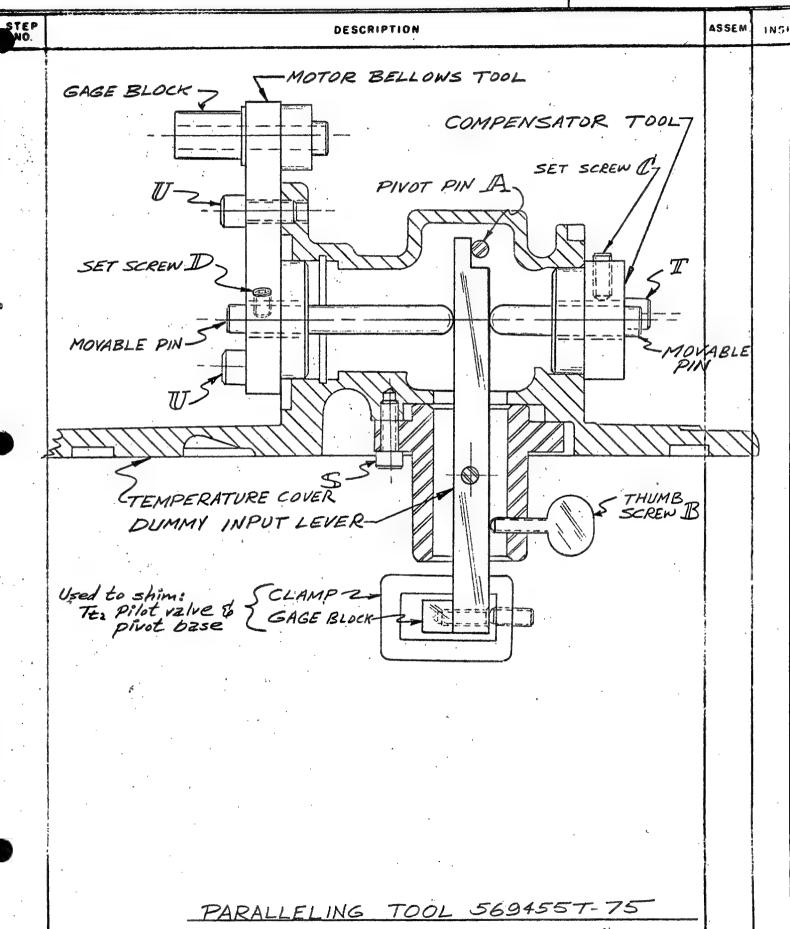
Group II Controls - Obtain the scribed dimension from evacuated diaphragm assembly. (Call it DIM. B. = record.) Subtract DIM. A from DIM. B. The result is the amount of spacers required between the evacuated diaphragm assembly and the temperature cover housing. Do not use more than four spacers to achieve this result.

- 10. Place the preset Motor Bellows Tool into the other end plate of the Push Rod Adjusting and Shimming Fixture and secure in place with three screws "U" (use normal torque). Move the end plate "X" of the fixture until the pin in the Motor Bellows Tool contacts the base plate, and the adjustable sleeve (mounted on the push rod assembly) is centered on the gage block of the Motor Bellows Tool. Tighten thumb screw "K" by hand after making sure that adjustable sleeve does not interfere with positioning of Motor Bellows Tool (i.e., adjust screw "L" in adjustable sleeve CCW).
- Il. Reach through the access hole of the gage block and adjust the screw "L" in the adjustable sleeve until no gap exists between the adjustable sleeve and the gage block surface. Tighten set screw "M" in adjustable sleeve and lockwire. Clockwise adjustment of screw "L" will decrease any gap between sleeve and gage block surface.
- 12. Remove adjustable sleeve and push rod configuration from Push Rod Adjusting and Shimming Fixture and install into temperature cover housing in conjunction with the shims (determined in Part 9) under the evacuated diaphragm assembly.
- 13. Refer to HS1503 specification, Paragraphs 4.2 4.2.2 to check shim value and dimension found in Steps 9 and 11 above.

HB F-206.0

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

NO. C-SHEET OF H.S. 1502 B Page 46



NS F-298.0 Page 1502 B NO. C-Page 47 LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION) SHEET OF ASSEM DESCRIPTION INSI NO. CLAMP. PUSH ROD ASS'Y. ADJUSTABLE SLEEVE & SCREW ASS'Y. SLOT GAGE PIN IT Adjust screw I to eliminate this gap -BLOCK - NUT . MI PIN GAGE BLOCK-ACCESS HOLE. 130" DIA. DEPTH"MIKE" - READING -MOTOR BELLOWS TOO COMPENSATOR TOOL END PLATE BASE-PLATE Y \mathcal{X} PUSH ROD ADJUSTING & SHIMMING FIXTURE 569455T-76

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

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| PAGE | 48 | he . | |

TITLE: Shimming of Tt2 Pilot Valve

OBJECTIVE: To shim the Tt2 pilot valve for null position.

REFERENCE: Figure #1 with Dummy Lever paralleling tool 569455T-75

- 1. Install Roller Gage Block (use normal torque on 2 screws X) on to Dummy Input Lever of Paralleling Tool 569455T-75 (still locked in position on temperature cover); and, using Clamp Tool, fasten Tt2 multiplying lever and pivot base assembly onto Roller Gage Block, and Dummy Input Lever to simulate the null condition for Tt2 input linkage. Tighten set screw Y (use normal torque).
- 2. Next, install the temperature cover into the assembled linkage housing and retain cover on linkage housing by the use of four screws (normal torque).
- 3. Set up null position (see "S" dimension on match data sheets) on Tt2 pilot valve by inserting correct size drill rod between pilot valve gear and spider housing; and using a feeler gage, determine the gap between pilot valve tip and Tt2 multiplying lever pad. The gap dimension is the amount of shims required beneath the pilot valve tip to establish null position.

| Record the shim stack | • |
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A. Remove temperature cover and conduct shimming operation.

CAUTION: The Tt2 pilot valve tip has left-hand threads.

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LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

NO. C-SHEET OF

TEP NO. DESCRIPTION ASSEM INS SHIMMING OF THE PILOT VALUE SHIM HERE THIS GAP TO MULTIPLYING LEVER & R TYLEVER. 569455-T-75 ROLLER GAGE BLOCK FIGURE I.

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TITLE:

Shimming of Tt2 Piston Servo with Adjustable Rack. (Group II Controls Only)

OBJECTIVE:

To shim the Tt2 Piston Serve to Provide a Means of Meeting Backlash Requirements

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REFERENCE :

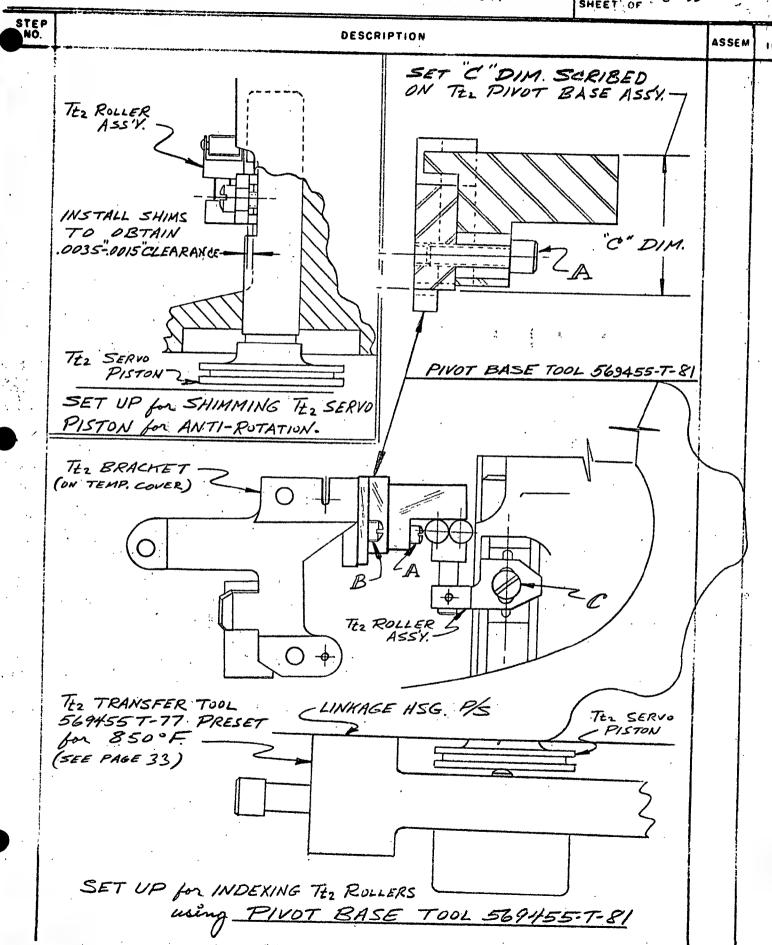
- 1. Assemble piston, servo, and rack plate and solid shim using initial shim .020 shim only.
- 2. Install above assembly in linkage housing.
- 3. With Ng cam locked to prevent rotation, measure backlash on piston with indicator gage.
- 4. Remove rack and piston assembly, disassemble and choose correct shim (shim to limit backlash between Ng servo and rack plate to .001 maximum).
- 5. Reassemble, install, and recheck backlash. IMPORTANT! Backlash must be checked at both ends of rack and with 3-D cam at both extremes of travel with backlash set as per requirements of .00l maximum. Cam followers must be disengaged from the 3-D cam during the check of 3-D cam and rack for freedom of movement. No tight spots or binding must exist. This motion of the 3-D cam and rack is a combination of horizontal and vertical movements to their extremes.
- 6. Remove rack and piston assembly and securely fasten, using rivets. Formed HD of rivets must meet envelope requirements (0.D. of rack plate and 0.D. of rivet HD and formed HD must not protrude into envelope created by minimum 0.D. of piston shank.)

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| | OB. REF 1. 2. | With T neath Tt2 ro housin Instal torque temperand ret Instal a "snu that the piston the rois not on the Instal torque Tool 50 the rois the rois the Remove bly (at torque Repeat | Figure Figure t2 piste the Tt2 ller ass g. Reco l Pivot , to "C' ature co ain in: l Tt2 ro g" condi ne rolle as it i llers to accessi linkage l temper lers co Tt2 Tra tempera indexi on scre | on corre roller sembly a ord shim Base To dimens over (in place of the Tt ble for a housin eature c sing pro- ture cor ng posit w C. above to | ctly i assembly to the place (2 scr sembly Positi to the 2 pist adjusting over one pive to the pive pool control to the pive pive pool control to the pive pive pool control to the pive pive pive pive pive pive pive piv | roller ot Bas ndexed ly to adjac 455-7- ribed of Tt ews B on to on the e 8500 on is r tment a tt Tran rt2 pis ot base ntacts om link to Tt2 | to Ng cobtain. ent mach 81, (precon pivot 2 multiple normal the Tt2 rollers pivot base F position to the stool are the link age hous piston to | am, insomo on the set by set base), lying letter on the set tool ion. The because of temper ousing (of 56945 rds the set by apply apply | tall sufforce of the control of the | - use no the line - use no the line - use no the brack de pivot be eque scre in such de down dedure of eller ass ever is crews - and Cam | shims be- petween the lakage rmal let of the lase assy. w C to a manner the Tt2 indexing embly scre installed normal Motion n so that position rface. r assem- late | | |

HS-F-296.0

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

H.S. 1502 B No. C Page 53 SHEET OF



Sanitized Copy Approved for Release 2009/12/10: CIA-RDP67B00657R000100170001-1

| H3 F-: | 298.7 | | LOG | SHEET AS | SSEMAL | Y / TEST | 2000. | | | 1 . | | H.S. | 1 ና ስ : | מנ |
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| | | NAME | | | | | | | | COVER | MARKED AGE BY | (*) REC | JUIR | ES IOO |
| | | NAME. | | | | ASSEMBL | YNO | | | | | HANGE | | |
| MAIN | ASSY | . NO. | MO | DEL NO. | | P | ARTS LIS | T NO. | | SERIAL | 1 . | | | |
| STEP NO. | T | | | | | | | | | SERIAL | NO. | | | |
| 14(). | - | TOTE. | | 4 | | DESCRI | | | | <u></u> | | AS: | SEM | INSP |
| | - | ITIE: | | xing of P | roport | ional Ga | in Lever | Rollers | 3 | | | _ | | |
| | 0 | BJECTIVE: | : To pr | roperly l | ocate : | rollers | to overr | ide leve | · er nivot | + | | | . | |
| | RJ | EFERENCE: | Figur | re Titled | • Pro | pontions | 7 O-2- 7 | | * prior | . | | | | r , |
| | | . Assemb | 4 | | • 110 | hor erolla | l Galn 1 | ever Fix | ture 56 | 59455T - 9 | 4 | | | |
| 1 | 4.0 | hollow | ie prop pin an | ortional do wire (X | gain) | lever in | to mount | ing brac | ket and | retain | with | | | • |
| | 2. | | _ | | -/- | | | | | | | | | |
| | ٠. | the pr | l the a oportio | lissassemb nal gain in and wi | oled Ar | m of the | Propor | tional G | ain Lev | er Fixt | ure int | 0 | | |
| | | with he | ollow p | in and wi | ire(Y) | e / Tii TTO! | l or rue | roller | oush ro | d) and | retain | | | |
| | 3. | Lower t | the mou | nting bra | ecket a | บไศ์แลวล | ~~~ 1 24~4 | · · · = A | | | | | | |
| | | and sec | ure in | place wi | th two | screws | (nse nor | mal torc | into tr | le serv | o housi | n | | |
| . | 4. | Ińsert | Gage Pi | in into + | ho area | | | | | | | | | |
| | | housing | bracke | et and in | stall | the inte | grating | P.V. hou | integra sing ir | iting pi ito the | lot va | 1 ve | | |
| ٠. ا | | | | | | | or que). | | | | | | | |
| | 5. | Install position | Dummy so th | CIP Cam | <u>569455</u> 7 | <u>r-27</u> int | o CIP bo | re of th | e servo | housin | g and | | | |
| | | 3220 RP | M cam p | nat the proposition. | roporti Dummy c | ional ga: cam should | in cam f | ollower | will co | ntact t | he . | 4 220 | . | 2 |
| | 6. | Tnsert | Ad ingto | hio Come | 75 Th = 10 | | · mount c | ve contra | Hamera | CO GISTAN | ea IN ha | ni 4 pag | e m | 3. |
| 1 | , | lightly | · | ble Gage | R TOCK | onto the | Fixture | Arm and | d tight | en scre | M II DII | | | |
| . | 7. | | • | | | | | | | • | | | | |
| | . • | Position secure i | in place | e with sc | on to | parting | surface | of the | servo | housing | and | | | |
| | .8. | | | | | | | | | | | | | |
| 1 | • | Engage F | torque) | Arm with while the cam posi | the M | ounting | Plate an | d tighte | n screv | a iiCii (i | 15 [.] 6 | | | |
| | • . | 322 0 RPM | l dummy | cam posi | tion. | | . бали са | W TOTTOM | er cont | tacts th | 10 | • | | n |
| | 9. | Lower th | ю Adjus | table Ga | ge Blo | ck in th | e Fixtur | e Arm un | .+47 44 | At 9 | | | | • . |
| 1 | | tacts th | e Gage | Pin in the | he over | rride le | ver slot | of the | integra | firmly ting P. | con- | • | 1. | • • |
| , | ۱۸ | • | _ | , | 0110 | 2CI GM 1 | n: ru br | ace using | g norma | ıl targu | e. | | • | • |
| | -∪• | Remove thassembly | he two | screws th | hat ret | cain the | proport | ional ga | in leve | r brack | et | , | | |
| 1 | , | Plate and | d lift | the prese | et Prop | , nemove ortional | ∍ screws L Gain L | "A" & "F ever Fix: | 3" from | the Mo | unting | | | |
| | ′ | housing. | | | 1000000000000000000000000000000000000 | *** | | The same of the sa | JUL O TT | Our otte | servo | : | | |
| | • | • | | | | | | | | | . | | Ι, | 1 |
| 1 | • 7 | | | | | | | | | | | | 1/ | |

HS, F-295.6

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

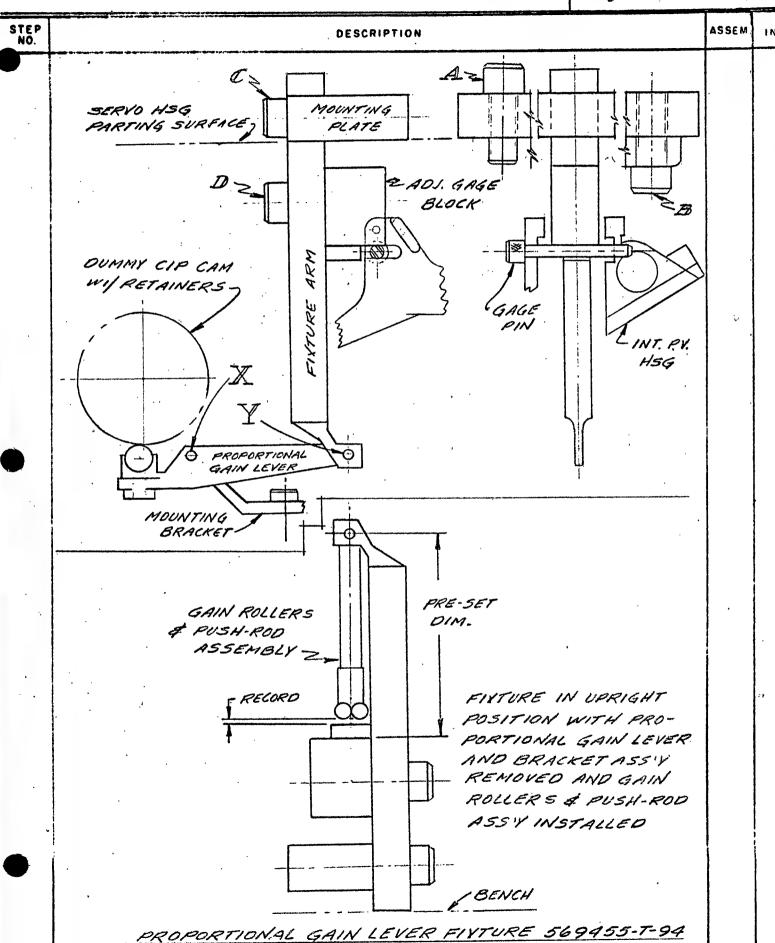
NO. C- H.S. 1502 B

| STEP NO. | · | DESCRIPTION | ASSE |
|-------------|-----|---|------|
| 4.49 | 11. | Place Fixture in an upright position; remove proportional gain lever and bracket assembly; and suspend gain rollers and push rod assembly from preset Fixture Arm by means of hollow pin(Y). | |
| | 12. | Add shims to the roller assembly until the rollers firmly contact the present Adjustable Gage Block. Record shim stack = | |
| | | A. For proportional gain lever which will incorporate the shims beneath the cam follower, use Feeler Gage to determine the gap between the rollers and the Adjustable Gage Block. Record Divide the gap measurement by 3 and add this amount of shims beneath | |
| | | the proportional gain lever cam follower. Record shim stack | |

HS F-296.0

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

NO. C- H.S. 1502 B SHEET OF Page 56



| HS F-21 | | LOG S | | • | | 9/12/10 : | | P67B0065 | NO: C | 5 н. | 1 S. 1502 age 57 | В |
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| DATE | | A | <u> </u> | <u> </u> | | - | - | - 6 | | MS REQUI | | TURE C |
| PROD. E | NG. | | | | | _ | | | | LY/TEST | | |
| INSP. | | | | : | · · | | | - | | MARKED (* Ge by ins | | 5 100 |
| | SSY NAME | | | 1 | SSEMBLY | NO. | | | | ASSY. CHA | | FFR |
| MAIN | ASSY. NO. | MO | DEL NO. | | P. | ARTS LIS | T NO. | | SERIAL | NO. | | |
| STEP NO. | | | | | DESCRIF | TION | · · · · · · · · · · · · · · · · · · · | | | | ASSEM | INST |
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| | TITLE: | | • | ntegrati | _ | | 24 h.* 16 m | ahimmin | - | • | · | |
| | OBJECTIVE | : To con | trol th | e braron | ring g | roove wi | acn by | SITIMITI | š.• | | | |
| | REFERENCE | : Rigure | s #1 & | #2 | | | | | | | | |
| | | | | • | | | | | • | • | | · |
| | | | | | | | | | | | | |
| • | 1. Shim | piston ri | ng groo | ve to ob | tain a | width of | .105 | = .002 a | 5 10110 | 15 | | |
| | b) | Using der flange wh record Measure t record | here pis | ton ring | s will Reco | seal. Ord other | Call the end - | is dimen | DIM. B | and -• and | | |
| | c) | Shims = (| (A - B - | .105) | E | | | Record | outboard | i side. | | |
| | | | | | • | | | Record | inboard | side. | | |
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Sanitized Copy Approved for Release 2009/12/10 : CIA-RDP67B00657R000100170001-1 H.S. 1502 B Page 58 LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION) SHRE Wer To ASSEM DESCRIPTION INST 105" PISTON RING GROOVE RED'D. SHIMS - A-IB-.105 575661 REF. FIGURE 1 TOOLING BLOCK CBA HOUSING : MOUNTING SURFACE 1171 whi. Same FIGURE 2

HSF-755. Sanitized Copy Approved for Release 2009/12/10 : CIA-RDP67B00657R000100170001-1

HAMILTON STANDARD

SPEC_NO. HS __1502B

DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

CODE IDENT NO. 73030

PAGE _____59_OF

TITLE:

Shimming of Area Linkage System.

OBJECTIVE:

To shim the idler lever bracket and the area cam follower so as to place the proportional gain rollers in a parallel position.

REFERENCE:

Figure Titled: Area Linkage Speed Transfer Fixture 569455T-26.

- Determine the available travel of the integrating piston, with spacers but no shims under screws, from the mounting surface of the CBA housing to the bottom of the bore. (See figure 2, sheet 41). Call this dimension "C". Measure the projection of the CBA cover into the integrating piston bore and call this dimension "D". Determine C-D=E. "E", travel of the piston, to equal .727 ± .005 in. after shimming under the screw on CBA cover side of the integrating piston. The difference between "E" and .727 in. is the amount of thickness to be added under the outboard screw. No shims are to be used under the screw toward the bottom of the bore."
- 2. Install a .342 ± .001 in. spacer block, 569455T, on the CBA side of the integrating piston so that the piston is .342 ± .001 in. from the projection of the CBA cover into the integrating piston bore. Maintain the piston against the spacer block until the area system shimming has been completed."
- 3. Remove the plate which holds the dial indicator and "jack" screw from Area Linkage Speed Transfer Fixture 569455T-26; and place the fixture on the linkage housing (2 screws - normal torque) so that the adjustable arm will reach into the housing and pick up the area pin on the Ng servo pull rod. Install Cam Motion Fixture 569455T-55 on to linkage housing (2 screws - normal torque); index fixture with Tt2 reset follower in Ng cam detent; and then insert appropriate gage block (corresponding to the L.P. for the low R.P.M. reference marked on C.I.P. dummy cam 569455T-27 to be used per technique discussed in trimmer shimming (page 22 part 20b. Adjust the arm on the transfer fixture so that the area pin fits into the slot of the fixture arm. The fixture is now set so that when transferred to the servo housing, the moveable pin on the arm will simulate the low R.P.M. reference position of the Ng servo. Install the plate holding the dial indicator and "jack" screw configuration back on the Transfer Fixture.
- Insert a nominal .028" shim under the idler bracket and a .090" shim under the area cam follower and complete the assembly of the area linkage into the servo housing. Install the Area Linkage Speed Transfer Fixture and retain in place on the servo housing (2 screws normal torque).

 Install the appropriate cam retainers on CIP Dummy Cam 569455T-27 and insert the drummy cam assembly into the CIP bore of the servo housing using Cam Motion Fixture 569455T -55. Exercise care to prevent cam follower damage of nitrided cam retainers during installation and movement of dummy cam.

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Hamilton Standard DIVISION OF UNITED AIRCRAFT WINDSOR LOCKS, CONNECTICUT . U.S.A.

SPEC. NO. HS

CODE IDENTINO. 73030

PAGE 60 OF

The latest version CIP dummy cam is sectioned into three parts: Notes

- The "front" third contains the two notches which are used to index the area follower with the proportional gain follower.
- b. The middle section is recessed to simulate the high R.P.M. reference cam radius and, also, contains a special "saddle" which is to be used to stroke the proportional gain rollers during the initial paralleling operation.
- The back section of the cam simulates the low RPM reference cam radius. Always install the dummy cam into the CIP bore with the cam retainers in place and the end containing the two notches in first.
- 5. Install the special sleeve and AN-plug of Fixture 569455T-71 into the transducer housing and mount the transducer on to servo housing. (4 screws normal torque). Set up the dial indicators of fixture 569455T-71 to pick up the travel of the integrating pilot valve and the transducer piston. Use "jack" screw on Transfer Fixture 459455T-26 to move the paddle of the idler lever.
 - Check transducer and integrating pilot valve dial indicator pickup for repeatability by cycling idler lever between 3220 and h210 RPM positions.
- Place dummy cam at the high R.P.M. reference position (with proportional gain follower in line with saddle of cam); and using the cam motion fixture, stroke the dummy cam so that the proportional gain follower ONLY rides up and down the saddle in the middle section of the cam. While stroking cam (to move gain rollers), use "jack" screw to move idler lever to a position where no travel change is noted on transducer piston dial indicator. This places the proportional gain system in a parallel position. Set the dial indicator on the integrating pilot valve to read zero. This zero setting must not be disturbed during the remainder of the calibration procedure; and stroking of the proportional gain rollers should be discontinued. If integrating PV dial indicator is disturbed repeat "zeroing" procedure described above.
- 7. Move the dummy cam from the high R.P.M. reference position to where the two cam notches will engage with the cam followers. Use the following procedure to be sure that both followers are seated squarely in the notches: Loosen the locking screw on the idler adjusting plates and rotate the dummy cam clockwise until both followers are out of their notches. This will move the area follower to the end of the adjusting slots. Now rotate the cam counterclockwise until the proportional gain follower seats squarely in its notch. To seat the area follower, place the blade of a small screwdriver between the idler bracket and one of the idler lever adjusting plates and lift the follower until the spring load on the follower causes it to seat squarely in its notch. Tighten the locking screw on the adjustment (use normal torque).

Use a .0001 inch indicator to measure indicator to measure integrator pilot

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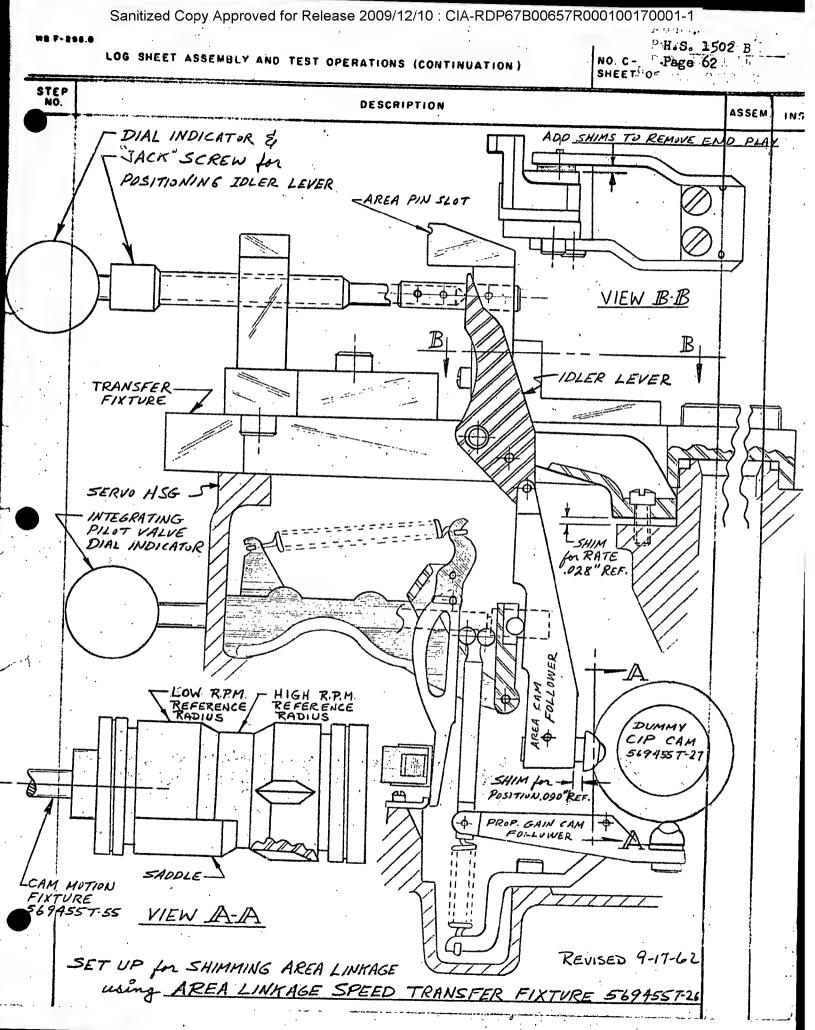
Hamilton Standard DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT . U.S.A.

CODE IDENT NO. 73030

PAGE 61

valve motion. After making and securing the position adjustment, note IPV indicator reading. Rotate the dummy cam slightly from this "notch" position in both directions and note IPV motion. If position adjustmentis correct, rotation of the cam in both directions will cause IPV motion in only one direction from the notch reading.

- 8. Move the dummy cam to the low RPM ref. position and adjust idler lever paddle with "jack" screw until integrating pilot valve dial indicator reads zero as indexed in Part 6. Take reading on dial indicator or which measures the stroke of the idler lever and record. Move the dummy cam to the high R.P.M. reference position and repeat "zeroing" the two measurements. If the stroke difference (at the two Ng speed conditions) does not equal the reference dimension marked on CIP dummy cam 569455T-27. go to figure #1 page 46 to determine whimming correction required. Everytime the idler bracket is raised or lowered by re-shimming, the area cam follower must be re-adjusted using the procedure in Part 7. so that the follower is squarely seated in its notch.
- 9. After the idler lever stroke has been set to the reference dimension, check the idler lever paddle to Ng pin relationship at either dummy cam speed condition (low or high RPM reference) by using the dial indicator to determine the gap (if any) between lever and pin while the integrating pilot valve dial indicator is reading zero. If there is a gap between lever and pin, it can be eliminated by shimming under the area cam follower. For every .005" at lever-to-pin gap, a .001" shim under the area cam follower will be required (5 to 1 ratio). Movement of the idler lever towards a low speed direction in order to eliminate the gap between the paddle and pin requires the addition of shims to the area cam follower. Movement of the idler lever towards a high speed direction to eliminate the gap requires the removal of shims from the camfollower. If any change is made to the shimming under the area cam follower, the reference stroke of the idler lever must be checked and corrected if necessary.
- 10. After the area linkage system has been shimmed correctly, set the idler lever and dummy cam at their respective low RPM reference positions and then move the idler lever to speed positions of less than and greater than the low RPM reference. Minimum travel (as registered on the transducer piston dial indicator) should be ±.045" from the set speed. Repeat the above with the idler lever and dummy cam at their high RP reference positions. Min travel of the transducer piston should be ±.045" from the set speed. If less than the required $\pm .015^n$ travel is observed, a dimensional check of the servo casting and related parts will be required.
- 11. Add shims to idler lever bracket to prevent intereference of idler lever with Ng pull rod sleeve and to remove "end play". See view B-B, page. Record shim stack
- 12. Place Tt2 servo into the fully saturated hot position (.450 L.P.) set speed idler lever to an L.P. position of .1011 (6630 Ne rpm) and stroke CIP servo cam from .340 L.P. to 1.360 L.P. Measure transducer motion for this range of CIP servo travel and it should not exceed .003 maximum. If transducer travel exceeds this limit, recheck shimming.



EUGENE DIETZGEN CO.

EXAMIDIES

I. ASSUME IDLER LEYER

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SHIMS SHOULD BE ADDED

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DER LEVER STROKE (A) - Works

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

SPEC. NO. HS <u>1502 B</u>

CODE IDENT NO. 73030

| | | PAGEOF |
|---|--|---|
| | TITLE | Shimming of CIP System |
| , | OBJECTIVE: | To shim the bellows housing cover, multiplying lever, and CIP pilot valve for a null position. |
| | REFERENCE 8 | Figures Titled: CIP Weight Fixture 569455T-19 and Dummy CIP Sensor Lever 569455T-91. |
| | torque) valve am inclusion defined housing | ervo housing on Adapter Arm 569445T-14 (4 screws - normal torque). The Dummy CIP Sensor Lever into the servo housing (2 screws - use normal along with the multiplying lever mounting bracket, and the CIP pilot and housing assembly (which has been preset at the null position by the on of shims between the pilot valve gear-head and housing in the amount by the "S" dimension on the match data sheets). Note: CIP pilot valve requires positioning in the servo housing with Locating Fixture 35 which has been preset to the bore (in the linkage housing lug) that the CIP shaft. |
| | 2. Use smal Dummy CI of the s | l "C" clamp to fasten the CIP multiplying lever assembly on to the P Sensor Lever so as to be aligned with the dowels in the mounting lugs ervo housing and the bracket. |
| • | 3. With the determin | multiplying lever locked in this position, use a feeler gage to e the amount of shims that will be required: |
| | a. Between recor | een multiplying lever pivot bracket and servo housing lug |
| | b. Betwe | een multiplying lever pivot bracket and mounting bracket lug record. Pivots must be in contact. |
| | c. Betwe | een multiplying lever and CIP pilot valve tip record. |
| 4 | . Insert th | ne required amount of shims between the multiplying lever pivot and the lugs of the servo housing and mounting bracket, and retain plying lever bracket in place using 2 screws - normal torque. |
| _ | _ | |

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HAMILTON STANDARD

SPEC. NO. HS __ 1502 B

DIVISION OF UNITED AIRCRAFT CORPORATION

CODE IDENT NO. 73030

WINDSOR LOCKS, CONNECTICUT, U. S. A.

| PAGE | 65_ 0F |
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|------|---------------|

With the bellows assembly pivoting on the Dummy C.I.P. Sensor Lever slot, position C.I.P. Bellows Shimming Fixture (569455-T-19) over the C.I.P. bellows housing and bolt on using the two (2) lower bolt holes of the housing and bolts provided with the fixture. Next screw rod into the plate which is attached to bellows header. Engage hook from Hunter force gage (part of C.I.P. Bellows Shimming Fixture) over pin in rod. Install Indicator Stand (569455-T-28) onto the fixture by means of the tapped hole provided. Position Testmaster Indicator (part of 569455-T-28) so that the indicator point rests on the C.I.P. Bellows Shimming Fixture Plate. Zero out the indicator. Using the adjusting screw of the Fixture, load the Hunter force gage with 13.67 lbs. (read as close as possible). Next, read the Testmaster Indicator. Record this amount . This reading Testmaster Indicator. Record this amount is the additional amount of shims to install between the C.I.P. bellows housing and the bellows header.

7. Remove the C.I.P. Bellows Shimming Fixture and the C.I.P. bellows housing from the servo housing. Remove plate from bellows and build up C.I.P. bellows housing using shims recorded in part 6. above.

H.S. 1502 B LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION) Page 66 SHEET OF DESCRIPTION ASSEM MOUNTING BRACKET CIP MULTIPLYING LEVER ASSYT C"CLAMP -CIP BELLOWS HSG. - SERVO HSG. SHIM PER PART 32. DUMMY CIP SENSOR LEVER 569455791 CIP BELLOWS Assy. SHIM PER PART 36 SHIM PUR PART 3C. S"DIM. WEIGHT FIXTURE PLATE LSHIM per PART 6. CIP PILOT VALVE & HSG. ASSY. WIEW ALA CIP WEIGHT FIXTURE 569455

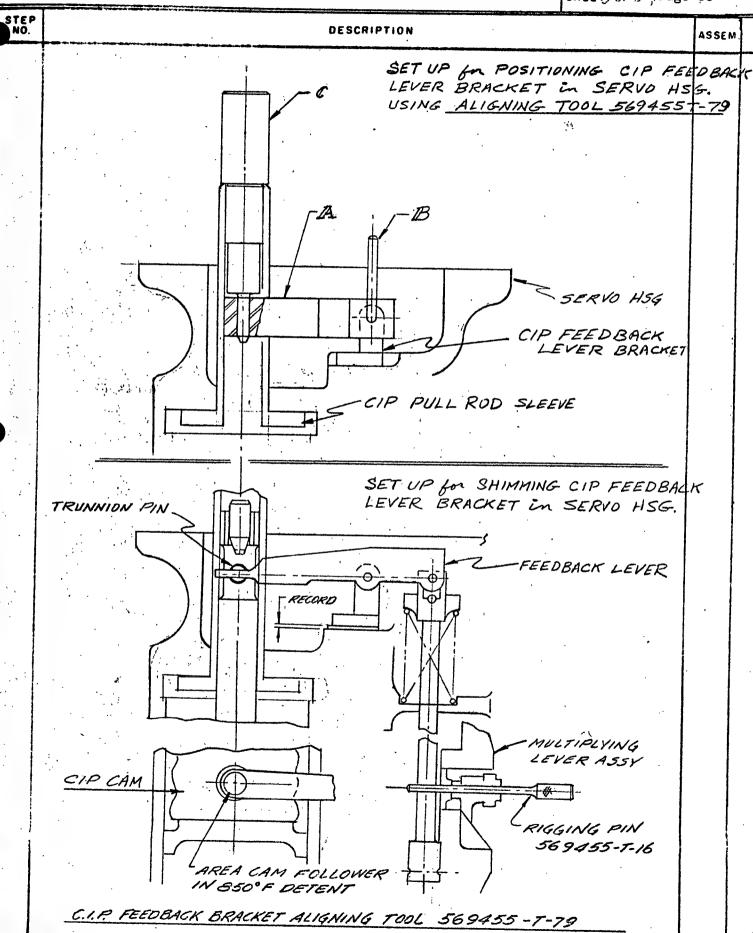
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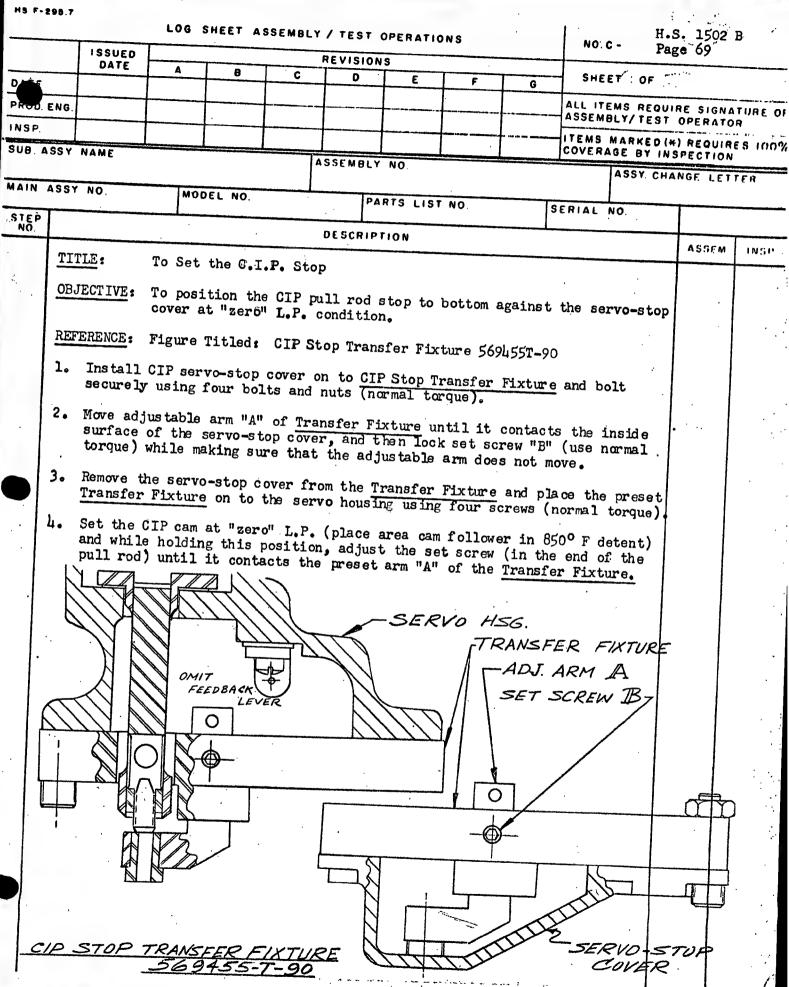
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MS F-295.0

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

NO. C- H.S. 1502 B





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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

| | SPEC. NO. HS1502 B |
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| l | CODE IDENT NO. 73030 |
| | PAGE71 OF |

TITLE: Shimming of Transducer Pilot Valve. (Group I controls only. No shimming required on Group II controls)

OBJECTIVE: To shim the transducer valve to provide a proper signal to the ENC

REFERENCE: Figures #1 and #2

- 1. Proper shimming of the transducer valve can be accomplished, only, during rig calibration.
- 2. It is imperative, however, that the person doing the shimming be aware that the pin in the transducer sleeve should always be aligned with the proper slot (in the transducer housing) which is "in-line" with the external pressure balance tube. There are two machined wall "break throughs" approximately 90° each way from the correct pin slot that may be mistaken for slots.
- 3.0 The transducer assembly is to be built with a nominal stack-up of .100.

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HAMILTON STANDARD

SPEC. NO. HS 1502 B

DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

CODE IDENT NO. 73030

PAGE ___73__OF _

TITLE: Indexing of Tt2 Servo Piston Adjusting Screw

OBJECTIVE: Adjusting Screw indexes two halves of Tt2 servo into proper relationship with each other.

REFERENCE: Figure Titled: Tt2 Transfer Tool 569455T-77

- 1. Position linkage housing half of Tt2 piston so that the Tt2 reset lever ball follower falls into 850°F end of the slot detent of the Ng Cam (push in on piston toward linkage housing parting surface).
- 2. Insert appropriate end of Tt2 Transfer Tool onto the Tt2 bore of the linkage housing; secure in place (4 bolts B, & nuts C) using normal torque; and loosen screw A to allow plug of Transfer Tool to make contact with insert in Tt2 piston of linkage housing. Lock screw A in this position (use normal torque).
- 3. Set up the area jack screw fixture T26, previously adjusted in "shimming of area linkage" sheet 42, on the parting face of the servo housing. Index the area paddle lever dial indicator at zero with the area pin simulator in the low speed position. Turn jack screw to move the area paddle .556" in the increasing speed direction. With the area paddle in this position place the area follower in the CIP cam detent. With these conditions set up measure and record the dimension from the indicator mounting surface on the servo housing to the end of the Tt2 piston.
- 4. Remove the area fixture and place the preset temperature transfer fixture over the Tt2 bore of the servo housing.
- 5. Adjust the screw until the dimension obtained in part 3 is established. Make sure the Tt2 piston is loaded against the fixture and shims.

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HAMILTON STANDARD

SPEC. NO. HS __ 1502 B

DIVISION OF UNITED AIRCRAFT CORPORATION

WINDSOR LOCKS, CONNECTICUT, U. S. A.

PAGE __74 OF

CODE : ENT NO. 73030

TITLE: Shimming of P.R.V. Fail-Safe Relief Valve

OBJECTIVE: To Set the Opening Point of the P.R.V. Relief Valve

REFERENCE: Figure Titled: Ball Check Valve Fixture 569455T-97

- Lightly lap the valve to the hydraulic housing seat.
 (a) A continuous circumferential land on the valve seat should be noted (about .001" wide).
- 2. Thoroughly clean all parts.
- 3. Install the P.R.V. check valve, retainer and spring into the hydraulic housing and secure with Tru-arc ring.
- 4. Place "O" seal on Ball Check Valve Fixture and install Fixture into P.R.V. bore. Secure Fixture to hydraulic housing by inserting 3 screws through the fixture flange plate (normal torque). Adjust Fixture Plug until bottomed in P.R.V. bore and tighten lock nut (use hand torque).
- 5. Connect the Ball Check Valve Fixture to a source of air pressure (0 to 50 psig) with an attached air pressure gage.
- 6. While listening to the relief valve bore and observing the air gage, slowly increase the air pressure to the <u>Fixture</u>. The relief valve cracking pressure is characterized by a sudden expulsion of air from the relief valve bore and an approximate 1 psig pressure drop-off on the air gage.
- 7. Add (or remove) shims from the check valve spring until it takes a pressure of 21 to 24 psig to unseat the check valve.

 Record shim stack

8. With check valve properly shimmed, check the total valve travel (from seat to stop), which should be .020* min.

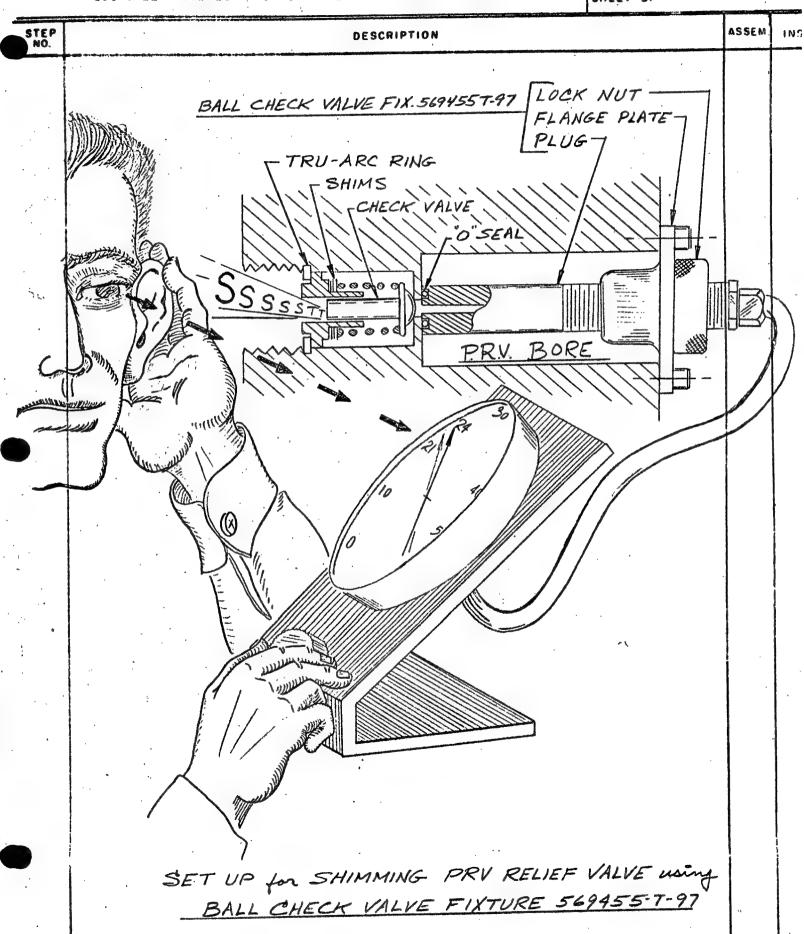
Record travel

9. Repeat Step 5. using fluid (calibrating fluid or fuel) instead of air. Vary the fuel pressure from 0 to the cracking pressure (21-24 psi). Leakage from the valve shall be no more than 10 cc/min up to the cracking pressure. If leakage exceeds this value, repeat Step 1 until the leakage is sufficiently reduced.

HS F-295.0

LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

H.S. 1502 B NO. C-SHEET OF Page 75

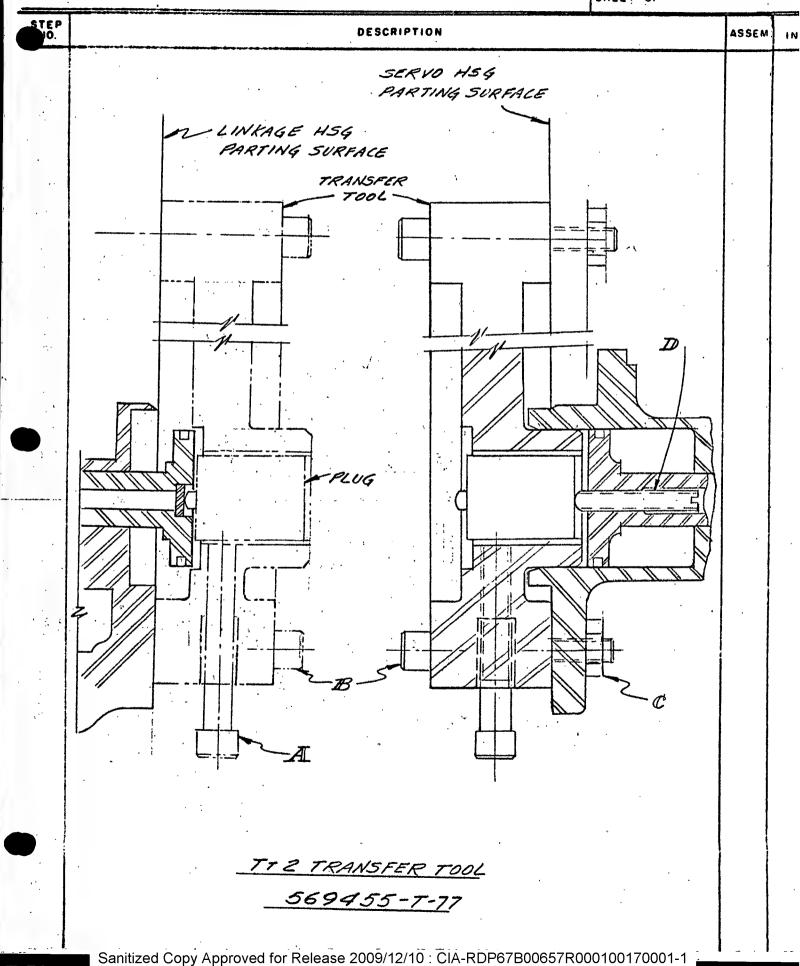


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LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION)

NO. C-SHEET OF H.S. 1502 B Page 76



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HS1502 K

Sanitized Copy Approved for Release 2009/12/10: CIA-RDP67B00657R000100170001-1 NO. C-SHEET OF 78 LOG SHEET ASSEMBLY AND TEST OPERATIONS (CONTINUATION) ASSEM DIAL INDICATOR MOUNTING SURFACE RECORD RECORD CIP CAM RETAINER No PULL ROD SLEEVE CIP CAM FIGURE 2. CIP SERVO POSITION FIGURE 1. No SERVO POSITION

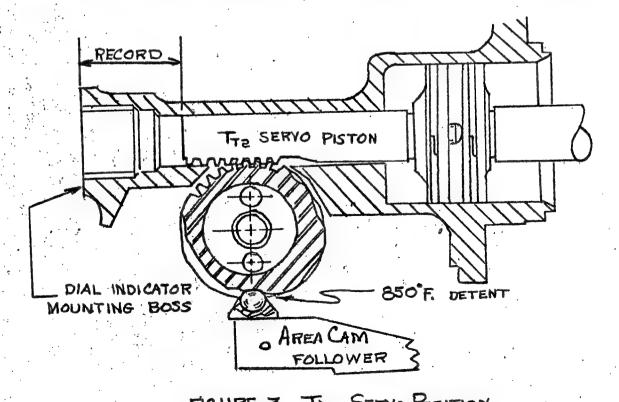


FIGURE 3 TO SERVO POSITION

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HS F-758. 1B 6/62

Hamilton Standard PIVISION OF UNITED WINDSOR LOCKS, CONNECTICUT . U.S.A.

D AIRCRAPT CORPORATION

SPEC. NO. H\$ __1502B_
CODE IDENT.NO. 73030

PAGE __79_ OF

INDEX Page No. Use of Solid Shims Run-in of PRV Sensor Assembly Shimming of Speed Set Cam Follower Lever (Group I Controls) 6 Shimming of Speed Set Cam Follower Lever (Group II Controls) 4. 8 Shimming of CBA Cam Follower Lever. 10 6. Run in of Speed Servo 12 Shimming of Temperature Reset Follower and Bracket 7. 14 8. Shimming of Acceleration Lever Cam Follower 16 Adjustment and Shimming of CDP System 9. 18 10. Adjustment of Min Ratio 24 11. Shimming of Throttle Valve position adjustment (Group II Controls Only) Shimming of Ng and CIP Cam and Pull Rod Assemblies 12. Shimming of Droop Lever Assembly (Group II Controls Only) 13. 14. Shimming of Ng Speed Governor and Pilot Valve 30 15. Run in of Spider Housing Pilot Valves 32 Shimming of CIP Worm Wheel and Shaft. 16. 33 Alignment and Shimming of Trimmer Housing (Group I Controls) 17. 35 Alignment and Shimming of Trimmer Housing (Group II Controls) 18. 39 19. Shimming of Temperature Cover 屲 Shimming of Tt2 Pilot Valve 20. 48 Shimming of Tt2 Pivot Base 21. 50 Shimming of Tt2 Piston, Servo with Adjustable Rack 22. (Group II Controls only) 23. Tt2 Rollers Adjustment Indexing of Proportional Gain Lever Rollers 24. 54 57 59 64 25. Shimming of Integrating Piston 26. Shimming of Area Linkage System 27. Shimming of CIP System To Position & Shim the CIP Feedback Lever Bracket 28. 67 29. To set CIP Stop 69 Power Lever Shims 20. 70 Shimming of Transducer Pilot Valve (Group I controls only) 31. 71 Indexing of Tt2 Servo Piston Adjusting Screw. 32. 73 Shimming of PRV Failsafe Relief Valve 33. 74 Documentation of Ng, Tt2 and CIP Servo Postions. 34.

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT

H.S. 1502B Amond. 1 Page 1 of 3. E.C. AZ 72199 Date: 10-10-62

H.S. 1502B

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Change Page 5 to read as follows:

TITLE: Run in of PRV Sensor Assembly

OBJECTIVE: To break-in the PRV Pilot Valve, Housing, Damper and Drive Gear.

REFERENCE:

- 1. Assemble PRV Sensor into fixture. The bi-metallic disc retainer and the spring and seat do not have to be included in this test.
- 2. Adjust the Pilot Valve for Position I defined as ±.050 ±.010 displacement from "null" toward the damper end. With the PRV Sensor immersed in the test fluid (room temp. to 200°F, 10 micro filtration) drive the sensor at a speed of 300 to 4500 RPM for four (4) hours.
- 3. Upon completion of above four-hour run, adjust Pilot Valve for Position II, defined as approximately null position. At this position, repeat the hours and speed called out for Position I.
- 4. Upon completion of Position II running, adjust pilot valve for Position III, defined as .050 ±.010 displacement from null away from damper end. Repeat the hours and speed called out for Position I.
- *5. Disassemble the PRV Sensor and examine parts for distressed area following the 12 hours running. Abnormal wear or scuffing of bearing surfaces shall be cause for rejection, replacement of parts, and rerun of the 12-hour break-in.
- 6. Ref. Note: This procedure is the only operation necessary to match the gear to the housing.

Abnormal Wear of Scuffing is defined as follows:

- 1. In those areas which show contact, no deterioration in surface finish shall result.
- 2. No locally distressed areas are permitted (circumferential or exial scratches or chips).
- 3. Circumferential wear patterns if evident shall occupy no less than 90% of the total circumference for purely rotating applications (pilot valves) and no less than 60% for rotating and translating applications (servo pistons)

Quality Engineering shall ascertain if a part is acceptable using the above definitions as a guide.

Note: The Fluid Media for this test shall consist of a mixture of 10% by volume of Texaco Capella AA oil and 90% by volume of P&WA 523B fuel.

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT

H.S. 1502B Amend. Page 2 Of E.C. AZ 74299 Date: 10-10-6

H.S. 1502B

Amendment

Change Page 12 to read as follows:

TITLE: Run in of Speed Servo.

OBJECTIVE: To break-in the speed servo in its linkage housing.

REFERENCE: Figure titled speed servo cycling schematic.

1. Assemble and shim detail parts into linkage housing per following sections of this specification (HS 1502B)

Sheet 14 Temperature Reset Follower and Bracket

Sheet 6 &8 Speed Set Cam Follower Lever

Sheet 10 CBA Cam Follower Lever

Sheet 27 Ng Pullrod

Assemble CBA pilot valve, spring and associated hardware to side load 3-D cam. install Ng pullrod slegve, seal, and screws and lockwire screws. Install Ng pullrod and shoes with piston rings.

Install Tt2 servo piston into linkage housing with a piston ring. Set fixture 569455-T-26 as described in Para 3. on sheet 59 so that area linkage in servo housing can be calibrated while linkage housing is being run-in. Install fixtures containing other half of Tt2 servo for supplying pressures to

- 2. Cycle the speed and temperature servos 5000 cycles, full translation and return, with test fluid (10 micron filtration), Fluid temperature to be held between room temperature and 200°F. Maximum hydraulic pressure to servos to be 500 psig.
- *3 . Following completion of 5000 cycles disassemble and examine the Ng servo cam shoes, pistons, rings, pullrod and sleeve and Tt2 servo bore for distressed areas. Do not remove cam shoes from 3-D cam. Do not remove piston rings from cam shoes. Abnormal wear or scuffing of bearing surfaces shall be cause for rejection, replacement of parts and rerun of the 5000 cycle break-in. Replacement of any of the parts listed in this paragraph shall be cause for repeating the break-in procedure.

Abnormal Wear or Scuffing is defined as follows:

- 1. In those areas which show contact, no deterioration in surface finish shall result.
- 2. No locally distressed areas are permitted (Circumferential or axial scratches or chips). 3. Circumferential wear patterns if evident shall occupy no less than 90% of the total circumference for purely rotating applications (Pilot Valves) and no less than 60% for rotating and translating applications (Servo Pistons).

Quality Engineering shall ascertain if a part is acceptable using the above definitions

The Fluid Media for this test shall consist of a mixture of 10% by volume of Texago Capella AA oil and 90% by volume of P&WA 523B fuel.

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT

H.S. 7502B Amend. Page 3 of 3; E.C. AZ 7/1299 Date: 10-10-62

H.S. 1502B

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Change Page 32 to read as follows:

TITLE: Run in of Spider Housing Pilot Valves

OBJECTIVE: To break-in the Ng, It2 and T.V. Pilot Valves over their full operating ranges.

REFERENCE:

- 1. Install the spider housing assembly containing the Ng, Tt2 and T.V. pilot valves, the bearing housing, and the splined worm shaft into the test fixture. It is not necessary to run the bronze worm wheel and its shaft, but since this is matched to the splined worm shaft, care must be taken to prevent mixing up details.
- 2. Adjust $^{\mathrm{N}}$ g, Tt2 and T.V. pilot valves for Position I such that each valve is approximately at its "null" position (within ±.005). With the spider housing immersed in the test fluid, drive this input shaft at 3500 to 4500 rpm for four (4) hours. Fluid temperature is to be from room temperature to 200°F and oil is to be filtered to 10 microns.
- 3. After completion of the above four-hour run, adjust the Ng, Tt2 and T.V. pilot valves for Position I which is defined as .050 ±.010 displacement from null position in either direction. At this position run the spider housing for four (4) hours at 300-4500 rpm as was done at Position I.
- After completion of four-hour run at Position I, adjust the N g, Tt2 and T.V. pilot valves to Position III which is defined as .050 ±.010 from null in the opposite direction from Position II. Repeat the speed and hours called our for Position I.
- *5. Following the 12 hours total minning, disassemble pilot valves and gear train and examine parts for distressed areas. Abnormal wear or scuffing of bearing surfaces shall be cause for rejection, replacement of parts, and rerun of the 12-hour break-in. Gaution - Be sure that correct serial numbered bronze worm wheel is included with pilot valves when parts are packaged after completion of run-in.

Abnormal Wear or Scuffing is defined as follows:

- 1. In those areas which show contact, no deterioration in surface finish shall result.
- 2. No locally distressed areas are permitted (circumferential or axial scratches or chips) Circumferential wear patterns if evident shall occupy no less than 90% of the total circumference for purely rotating applications (Pilot Valves) and no less than 60% for rotating and translating applications (servo pistons)

Quality Engineering shall ascertain if a part is acceptable using the above definitions as a guide.

Note: The Fluid Media for this test shall consist of a mixture of 10% volume of Texaco Capella AA oil and 90% by volume of P&WA 523B fuel.

HAMILTON STANDARD DIVISION OF CHIEF ATROPP CORPORATION WINDSON LOCKS, CONNECTION

R.S. 1502B

Amend. 2

Page 1 of 1

B.C. AZ7370L

Date: //-/- Z

1502B "ASSEMBLY AND SHIMMING PROCEDURE JFCL7 MAIN FUEL CONTROL"

Amendment 2

Change H\$1502 as follows:

- 1. Belete paragraph 2.0 in its entirety.
- 2. Add the following to paragraph 1.2.2.

1.2.2 "569455-T-147 Speed Servo Run In Fixture M10673 PRV and Spider Housing Run In Fixture"

- 3. On makes 1 and 3 of Amendment 1:
 - * After the word reference add "Fixture M-10673."

b) Delete the asterisk preceding paragraph 5.

- c) Change the next to last sentence to read "Quality Engineering or Inspection Supervision shall ascertain ..."
- d) Change the note at the bottom of the page to read "The Fluid Media for this test shall be Dominion "A" Spindle oil that has been passed through a 10 micron-nominal filter."

On page 2 of amendment 1:

- a) After the word Fischematic" in the reference add "Fixture 569455-T-147."
 b) Believe the asterisk preceeding paragraph 3.
- c) Change the next to last sentence to read "Quality Engineering or Inspection Supervision shall ascertain ..."
- d) Change the note at the bottom of the page to read "The Fluid Media for this test shall be Dominion "A" spindle oil that has been passed through a 10 micron-nominal filter."

HAMILTON STANDARD DIVISION OF UNITED AIRCRAPT CORPORATION WINDSON LOCKS, CONNECTICUT

H.S. 1502B Amond. 3 Page 1 of 3.5 E.O. 73645. Date: //- £ - £ 2

HS 1502B MAGSEMBLY AND SHIMMING PROCEDURE JFC47 MAIN FUEL CONTROL

Amendment 3

- 1. Add atte and procedure titled "Torque Test of Tripmer Click Locks"
- 2. Add paragraph fand an ompanying metin to page 28.

Airer the Ng and CIP fams the completely assembled, inspect completed am and cam show assemblies to insure that cams shows A and B are with with ear other within \$0005 F.I.R. and that the surface assine to by O.P. "A" is parable, to the surface defined by OD. "B" within \$1005 for any rotational position."

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFY CORPORATION WINDSOM LOCKS, CONNECTION

Amend. Page 2 01 B.C. 73645 Date: //-2

H.8. ADSEMBLE AND SHIMMING PROCEDURE JFC47 MAIN FUEL CONTROL"

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OBJECT TH

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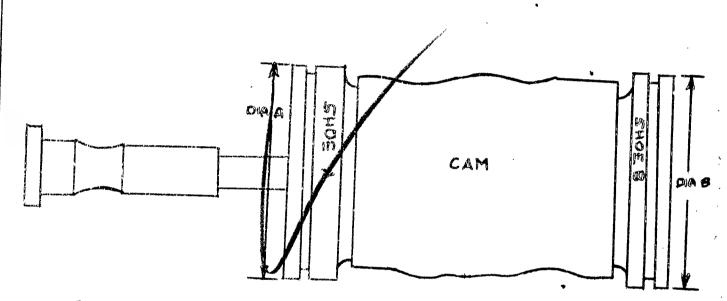
559455-1-455 Auspte Rights Screw Torque Wrench

Assemble (1, 1872) & rews springs, and crick locks into retainer and oneck link lock torms wrench. This torque must be 1.5 to 3.0 in #. If the torque not meet this requirement adjust the click a mad in or one action the above torque requirement is met.

HSF-755. 1A 5/61

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

H.S. 1502B Amend. 3 Page 3 of 3 E.C. 73645 Date: //-2-62



COMPLETE CAM AND SHOE ASSEMBLY

HAMIETON STANDARD DIVISION OF UNITED AIRCRAFT COMPORATION WINDSON LOCKS, COMMECTIGUT

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H. S. 1532 "ASSEMBLY AND SHIMMING PROCEDURE, JFC47 MAIN FUEL CONTROL"

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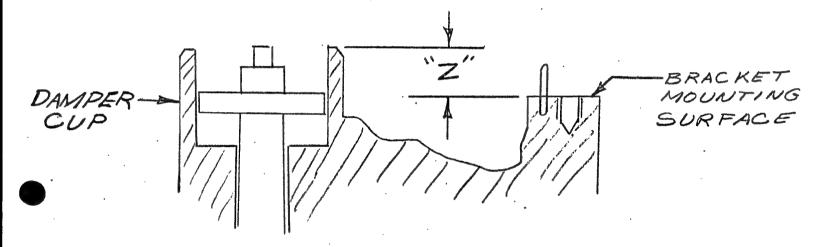
1. Add the Ng Governor anit rotation bracket shimming procedure per attached sheets.

H.S. 1502 Amend. 4/ Page 2 of 3 E.C. AZ71,622 Date: //-2/6-6-2-

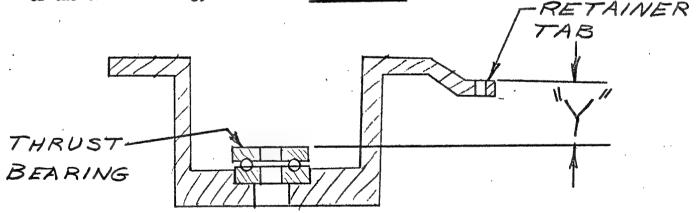
TITLE: Ng Governor Anti-Rotation Bracket Shimming

OBJECT: To shim the anti-rotation bracket so that anti-rotation link is perpendicular to the axis of the Ng pilot valve at its null condition.

1. Measure the distance from the bracket mounting surface to the top of the Ng damper cup "Z". Record_____.

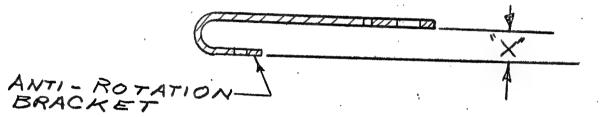


2. Measure the distance from the top of the Ng spring retainer tab to the top of the thrust bearing, "Y". Record_____.



H.S. 1502 Amend. 4 "Page 3 of 3 E.C. AZ74622 Date: //-16-62

3. Measure the anti-rotation bracket from the mounting surface to link attachment surface "X". Record



4. Determine the Ng pilot valve "S" dimension from the component pressure sensitivity data. Record

5. The thickness of shims to add between the bracket and the bracket mounting surface is as follows:

Shims = Y+Z+S-X



HS F-755.1 8/54

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

SPEC NO. HS _1521

CODE 730 30

Page 2 of ____

1.0 Scope

The purpose of this specification is to provide a pressure test procedure for Stainless steel Brazed housings used on the JFC-51 in order to insure detection of minute leaks.

- 2.0 Equipment Required
- 2.1 A pressure test rig capable of maintaining 1750 ± 20 psi for 15 minutes with seepage from the Housing being tested.
- 2.2 Pressure gages

High pressure 0-2000 psig 2000-psig, 1% Accuracy Low pressure 0-500 psig 1% Accuracy

- 2.3 Suitable fixtures to apply pressure to the high pressure area, as specified on the applicable blue print, and to bleed air from the area being tested.
- 2.4 Suitable fixtures to apply pressure to the low pressure area, as specified on the applicable blue print, and to pleed air from the area being tested.
- 3.0 TEST FLUID

Test fluid shall be MIL F-7024A type II or corrosion inhibited demineralized DEFINITION OF EXTERNAL TRACES.

14.0 DEFINITION OF EXTERNAL LEAKAGE

With the required pressure applied to the designated portion of the housing, remove all traces of fluid from the exterior surfaces. The term "no leakage" shall be defined as no appearance of fluid on the external surface of a housing, including no seepage or wetting on the surface, regardless of the fact that fluid does not run off the surface of the housing or forms droplets.

- 5.0 METHOD OF TEST
- 5.1.0 Leak test of high pressure area.
- 5.1.1 Install suitable fixtures to pressurize the high pressure area of the housing as designated by the applicable blue print. Apply pressure and bleed air from this area. Increase pressure to 1750 ± 20 psig and hold for 5 minutes.
- 5.1.2 Cycle pressure from 100psig to 1000 psig should be .5 12 seconds for each cycle.

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HS F-755.1 8/54

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT, U. S. A.

SPEC NO. HS _1521

GODE 73030

- 5.1.3 Apply 1750 ± 20 psig in high pressure area and hold for 15 minutes. There shall be no external leakage during this time.
- 5.2.0 Leak test of low pressure area.
- Install suitable fixtures to pressure the low pressure area of the housing as designated by the applicable time print. Arriy pressure and for 5 minutes.
- 5.2.2 Cycle pressure from 50psig to 115 psig 50 times. Time required to increase pressure from 50psig to 115 psig spould .5 12 seconds for each cycle.
- 5.2.3 Apply 175 ± 20 psig in low pressure area and hold for 15 minutes. There shall be no external leakage outing this time.



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SPEC. NO. HS _ 1572 B

DIVIDIDIT OF UNITED ARCHAFT CORPORATION - CODE IDENT NO. 70000 WINDOOR LOCKS, COUNTECTION, U.S. A.

PAGE _3___OF __

- 4.2.2 Assemble spring retainer 573173 to the connecting rod using two (2) sleeve spacers 571565 and hollow pin 69538A9-10.
- Place arring 573172 into its respective bore in the housing. Assemble 4.2.5 .050' of thim 57157? and washer 69387-21 on top of the spring. Note that, wasner must be in contact with spring. Install the complete piston assembly (Ref. paragraph 4.2.2) into the housing bord, carefully aligning the piston with the bore. Note that there must be no evidence of binding between piston and bera.
- 4.2.4 Install serve shim fixture 571575-T-2 to the housing using seven (7) screws 694,08525.8 and gasket 69397A-22. Apply a 11.6 pound lead to the piston pin. Take the complete assembly to the test rig for pressure calibration. Ref. HS spec. 1508B.
- Advemble shims 571577, determined at pressure calibration, piston stop 4.2.5 575982, and retaining ring 69983S-112 into the piston cover 575983.
- 4.2.5 Install platin cover 575983 to the housing using gasket coupling 69397A-22 and seven (7) bolts 69400825.8. Escure attaching bolts with lockwire MS 20995...321
- Accombly of Control Shaft Cover 575354 4.3
- Install bearing 88948 and sheare spacer 557116 into the bere of housing. 4.3.1
- Assumption of ring rotation 505758 to spring 576855 and then to commoding link 557115 using headed pin 69725-4618 and cotter pin MS24665-149. 4.3.2
- Silve this assembly, spring retainer and connecting link first, into the 4.3.3 bors of housing with the connecting link is in the bors ready to receive the centrel shaft. .
- Carefully winds the control shart 5572 a through the connecting link, 4.3.4 opicer, and bearing. Install gasket 69590-20 to housing surface.
- Asymmble retury as all a seembly 574.713, ring of Bookh, two (2) motallic 4000 hove on gaskets 69508-21, and packing rotainer 0950/B24 to the control shed't.
- Install flat apteor 506 (4, bearing 89016, retaining ring 69965887, and 4.1.6 washer Sadow into control shart cover 573334.
- Attach received shart form 573334 to the housing 571700 using davan acrows Lasar Galler So, Grand scrous with lookwing Milolyship. Install recaining rich dillibration to control short,
- 4.4 Asia ber at Police a transfer Empiredek Inver Praisent
- Assemble metallic ga ket 694 COAPPE, tab washer 578846, and P.R.V. nozzle 15770; but the housing and tabe 571400. Secure the P.R.V. nozzle assembly 4.4.1

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HAMILTON STANDARD
DIVISION OF UNITED AIRCRAFT CORPORATION
WINDSOR LOCKS, CONNECTICUT, U. S. A.

SPEC. NO. HS 3 18 18

PAGE ___ OF __

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to the bound gots benong the end of tar washer at against the beson the nozzale out the fat sau down and the horizon.

- 4.4.2 Reference note 9. As:

 Dwg. 5763 b prior to assembly of bracket to determine whether or not spaler 5.015 is to be installed. Assemble feedback letter bracket 577akh into the housing using four screws 69691A45 and lockwise using MS20995N20. The lockwise most not cover the spring seat.
- 4.5 Assembly of Adjusting Lever
- Assemble adjusting screw 57681d to feedback lever 578883. Assemble piston adjusting lever 57318) to feedback lever 578883 using two (2) straight headed pins 69725-4620, two (2) contentions MS24665 149 and a washer AN96006.
- 4.5.2 Place one shim 5/72.5 into the countert re in crarket 57714. Slide lever assembly into the how day. The expring 577135 into the counterbore in bracket 577144. Engage to lever accembly with bracket 577144 using hollow pin 69538A8-20.
- 4.5.3 Attach spring 57n the lever allege ye
- Measure the introduction the non-thought one time to the top of pin-4.5.4 69538A8-20. It has almension add the half of the pin diameter which is .0425 plus .0668. Then let this sam be known as "X". Measure and record the distance from the housing parting line to the top of spring retainer lip, and the distance between the spring retainer lip and spring seating surface of fee mack lever 578883. first making certain the lever is buttomed. Let the sum of these two dimensions represent "Y". Turn adjusting screw 576818 until dimension "Y" equals 'X" dimension. With the feedback lever set at the required dimension as determined by the above, position the control shaft 557114 so that its missing tooth is in line with the index hole on the control shaft lover. Install a shall indicator against the adjusting screw 576818. With the dial indicator at zero, ap, ly a pulling force at the supporting pin for the spring 5% of 12 and record the force required to move the feedback lever .005 inch at the adjusting screw. Take this force, enter the force balance adjustment har and determine the proper amount of snims to be added. Add and record tre number of shims 577145 until the correct load is obtained. Do not use more than seven full shines.
- Upon the completion of the shimming, install adjusting sorew 579283 1 to adjusting lever 573.83, and locking screw 579283-2 to lever 578883. Check adjusting lever 573183 for freedom of malement. Disassemble feedback lever assembly if correline action is necessary.
- 4.6 Assembly of Filet Valve
- Before assembly, wheak the pilot live for "free fail" within the housing bore Assemble adjusting strew 6945 Alon, flar washer 576868, and shim 576340 to the pilot value. Assemble feedwark connector 576826 to the pilot value with holiow pin 69538B9 7 and 100 kwire with MS20995N20. Install piston 69538B9-7 and lookwire with MS20995N20. Install piston ring 69353-24 to the pilot value ring groove. Install the twiside portion of the piston ring in such a manner that the breaks in the rings are 180° apart.

Use piston ring took 571575-T-23 to compress piston ring and carefully slide the pilot valve assembly with feedback connector first into the bore of

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HSF-755.1A 5/61

HAMILTON STANDARD

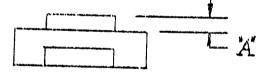
SPEC. NO. HS _ 1572 B'

DIVISION OF UNITED AIRCRAFT CORPORATION

CODE IDENT NO. 73030

WINDOOR LOCKS, CONNECTICUT, U. S. A. PAGE _5 OF __

- 4.6.2 (continued)
 69559ALL and plug ANSIL-10CL to housing 571700. Move the pilot valve from end to end and record the total travel. The total travel of the valve from the internal step to the position where the valve bottoms must equal the original "S", which is taken from the valve flow data, plus .060 maximum. The minimum value is .040. If valve travel does not meet the determined value, add or subtract shims 576540 until the required value is obtained.
- 4.6.3. Assemble spring 573168, spring retainer 576859, and self locking nut 69512-4. Adjust the nut until the spring retainer just contacts the spring. Then turn the nut down four complete turns.
- 4.7 Dampen Assembly
- 4.7.1 Assemble adjustable alignment shaft 576782 and friction collar 576785 into adjusting plate 576786. Slide fixture 571575-T-6 over the alignment shaft and bottom it. Using a depth micrometer measure from the top of adjusting plate to the top of the friction collar. Record this dimension.
- 4.7.2 Measure dimension "A" on spring retainer 576783.



Subtract this measurement from that obtained in 4.7.1 and then subtract and 5 Inch from this difference. This amount is the required thickness of snim 576045.

- 4.7.J Remove fixture 571575-T-6. Assemble shim 576845, spring 576784, spring 576783, and internal retaining ring 699835-50 into adjusting plate 576786. Compress spring using compression tool in order to assemble snup ring.
- 4.7.4 Assemble the hardware in 4.7.3 into damper adjusting housing 576786 and retain with self locking nut 69512-4.
- 4.7.5 Strew the stationary feedback plate 576788 into the lever 573554 until it bottoms.
- 4.8 Assumble Spring 576812 to Pin in the half area servo.
- 1.8.1 Place appling tendion washer 576819 and damper assembly into bore in the half area housing.
- 4.8.2 Place metallic face seals 568405-31, and -162 and metallic face seal showed 568406-31 and -81 into their proper Locations on the housing.
- 4.8.3 Install the complete half area housing wasembly to the housing and valve assembly. Engage spring 576812 with pin 69725-4620, which is in the lever assembly. Make certain spring 576812 is securely attached at two ends. Install seven screws 69408B25-11 and lockwire using MS20995N32.

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DIVISION OF UNITED AIRCRAFT CORPORATION

CODE IDENT NO. 73030

MANDEOR LOCKS, CONNECTIOUT, U. S. A.

PAGE _ 6 _ OF _____

- 4.8.4 Adjust damper screw 576782 clockwise until it bottoms. Back off screw (counterclockwise) one third of a turn. Hold the screw in this position and tighten belf locking nut. 69512-4. Assemble plug AN814-8CL and metallic pasket 69559A12.
- metallic pasket 69559A12.
 Cack adjusting screws 579283-1 and -2 for safety. Center the Hollow Pin 6955cAb-20 on the feedback lever bracket 57714.

 4.9 Assemble metallic face seal sleeve 568406-200, metallic face seal 568405-206. linkage cover 571561 and secure with nine screws 69408B25-7.
- 4.10 Assemble name plate 69444B4 to housing using two drive screws 69415-0-2.
- 4.10.1 Assemble plug AN814-2CL and metallic gasket 69559A5 to the housing.
- 4.10.2 Upon completion of test per HS spec. 1508Blockwire all external scrows, covers and fittings per assembly dwg. 576276.
- 5.0 PRESERVATION FOR STORAGE

After completion of testing, the exhaust nozzle control assembly shall be drained of fuel and prepared for storage in accordance with H.S. Specification No. 380.

6.0 PREPAIGNTION FOR SHIPPING

Each unit shall be completely free of internal and external foreign material at the time of packaging and during shipment. All ports shall be capped with suitable plastic caps or their equivalent.

HS 1572 B

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HAMILTON STANDARD DIVISION OF VNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT

H.S. 1572 B Amend. / Page 1 of 3 E.C. 72493 Date: R-R-G2-

H.S. 1572 B "Exhaust Nozzle Control, Assembly of"

| Amendment | 1 |
|-----------|---|
|-----------|---|

- 1. Change Paragraph 4.5.1 from: '
 - "h.5.1 Assemble adjusting screw 576818 to feedback lever 578883. Assemble pistor at lasting lever 573183 to feedback lever 578883 using two (2) straight needed pins 69725-4620, two (2) cotter pins MSN465-149, and a washer AN960C6."

To read:

- *4.5.1 Assemble two (2) adjusting screws 576818 and 579283-2 to feedback lever 580725. Record the inst. torque of the 579283-2 screw. Assemble piston adjusting lever 580727 to feedback lever 550725 using two (2) straight headed pins 69725-4620, two (2) couter pins MS24665-149, and a washer AN96006.**
- . Change Paragraph 4.5.5 Prone
 - Wh.5.5 Upon the complete of of the shimming, install adjusting screw 579283-1 to his histing lover 573183, and locking screw 579283-2 to less a cosse. Check adjusting lever 573183 for freedom of the second of the

To reads

- "4.5.5 Upon the complete of the seaming, install adjusting serew 579263-2 of a stag lever 580727. The installation torque of the 19 minutes through the helicoil mid-grip in the lever more to the lever more than the loss. Check adjusting lever 573183 in the series in movement. Disassemble feedback lever assembly if the the lever is a contact of the essary.
- y. Change Paregraph acts for re-
 - "L.b.3 Assemble opring of the operate of 16059, and self looking but 69512-L. August one out until the spring retainer dust contacts the operage. Them turn the but down four complete turns."

To reads

"4.6.3 Assemble spring 573168, spring retainer 576859, and self locking nut 69532- The installation torque of the nut along the screw threads must be 2.0 3.45 in. lbs. Adjust the nut until the loring retains

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSON LOCKS, CONNECTICUT

H.S. 1572 B

Amend. /
Page 2 of 3

E.C. 72493

Date: 8-8-62

H.S 1572 B "Exhaust Nozzle Control, Assembly of"

| Amendment / |
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4. Change Paragraph 4.5.4 from:

"4.5.4 Measure the distance from the housing parting line to the top of pin 69538A8-20. To this dimension add one half of the pin diameter which is .0425 plus .0062. Then let this sum be known as "X". Measure and record the distance from the housing parting line to the top of spring retainer lip, and the distance between the spring retainer lip and spring seating surface of feedback lever 578883. first making certain the lever is bottomed. Let the sum of these two dimensions represent "Y". Turn adjusting screw 576818 until dimension "Y" equals "X" dimension. With the feedback lever set , at the required dimension as determined by the above, position the control shaft 557114 so that its missing tooth is in line with the index hole on the control shaft cover. Install a dial indicator against the adjusting screw 576818. With the dial indicator at zero, apply a pulling force at the supporting pin for the spring 576812 and record the force required to move the feedback lever.005 inch at the adjusting screw. Take this force, enter the force balance adjustment chart and determine the proper amount of shims to be added. Add and record the number of shims 577145 until the correct load is obtained. Do not use more than seven full shims.

To read:

"4.5.4 Measure the distance from the housing parting line to the top of pin 69538A8-20. To this dimension add one half of the pin diameter which is .0425 plus.0062. Then let this sum be known as "X". Measure and record the distance from the housing parting line to the top of spring retainer lip, and the distance between the spring retainer lip and spring seating surface of feedback lever 580725 first make certain that the lever is bottomed. Let the sum of these two dimensions represent "Y". Turn adjusting screw 576818 until dimension "Y" equals "X" dimension. With the feedback lever set at the required dimension as determined by the above, position the control shaft 557114 so that its missing tooth is in line with the index hole on the control shaft cover. Install a dial indicator against the adjusting screw 576818. With the dial indicator at zero, apply a pulling force at the supporting pin for the spring 576812 and record the force required to move the feedback lever .005 inch at the adjusting screw. Take this force, enter the force balance adjustment chart and determine the proper amount of shims to be added. If the force is less than 12.6#, no shims are required. Add and record the number of shims 577145 until the correct load is obtained. Do not use more than seven full shims.

HAMILION STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSOR LOCKS, CONNECTICUT

H.S. 1572 B

Amend. /

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E.C. 72493

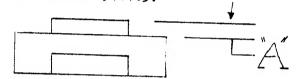
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H.S. 1572 B "Exhaust Nozzle Control, Assembly of"

| Amendment / | / |
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5. Change Paragraph 4.7.2 from:

"4.7.2 Measure dimension "A" on spring retainer 576783.



Subtract this measurement from that obtained in 4.7.1 and then subtract .468 inch from this difference. This amount is the required thickness of shim 576845."

To read:

- "4.7.2 From the dimension obtained in 4.7.1 subtract 0.585 inch (Nom. Dim.). This amount is the required thickness of shim 576845."
- 6. Change Paragraph 4.7.4 from:
 - "4.7.4 Assemble the hardware in 4.7.3 into damper adjusting housing 576786 and retain with self locking nut 69512-4."

To read:

Assemble the hardware in 4.7.3 into damper adjusting housing 576819 and retain with self locking nut 69512-4. The nut and the adjusting screw must have an installation torque (torque required to move the nut along the screw and the screw through the helicoil in the housing) of 2.0-13.0 in. lbs."

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSON LOCKS, CONNECTICUT

Amend. 2 Page 1 of 2 E.C. 72505 Date: 10-3-62

H.S. 1572B "EXHAUST NOZZLE CONTROL, ASSEMBLY OF"

| Amendment | 2 |
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| | |

1. Change paragraph 4.2.3 from:

"Place spring 573172 into its respective bore in the housing. Assemble .050" of shim 571577 and washer 69387-21 on top of the spring. Note that washer must be in contact with spring. Install"

to read:

"Place spring 573172 into its respective bore in the housing. Assemble .050" of shim 571577 and washer 69387-21 on top of the spring. Peel laminated shims in accordance with Fig. 1. Note. that washer must be in contact with spring. Install"

2. Change paragraph 4.5.2 from:

"Place or shim 577145 in to the counterbore

to read:

"Place eight shims 577145 into the counterbore"

3. Change paragraph 4.5.4 from:

supporting pin for the spring 576812 and record the force required to move the feedback lever .005 inch at the adjusting screw. Take this force, enter the force balance adjustment chart and determine the proper amount of shims to be added. Do not use more than seven full shims.

to read:

pin for the spring 576812 and record the force required to move the feedback lever .005" inch at the adjusting screw. This force must be 11.6 ± .2 lb. However, with 8 shims under the lever the force should be too high. To decrease the load rotate the spring 577135 clockwise or counter-clockwise 180°. Rotation lever load cam thus be provided. For best ENC performance as many shims as ditions should sufficient shims be added to cause the spring to go to solid height when the flapper hits the nozzle.

4. Add the attached figure.

Sanitized Copy Approved for Release 2009/12/10: CIA-RDP67B00657R000100170001-1 H.S. 1572B REPORT NO. Pg. HAMILTON STANDARD Date: E 2-VSE THIS TECHNIQUE ON ALL LAWINMITTED STIME ROUDDED ELSE SHIMS MUST SHOND EDSE COUNDED SEEL BURES. FROM STROLL A TIME Coepas Y ST 1600 H. S. F-62

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HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION WINDSON LOCKS, CONNECTION

H.S. 1572B

Amend. 3

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Date: 13 - 16 - 6 2

H.S. 1572B "EXHAUST NOZZLE CONTROL, ASSEMBLY OF"

Amendment 3

1. Delete page 7 (Graph F-5263).